

AD-A093 215

HUGHES AIRCRAFT CO CULVER CITY CA

F/6 14/2

EVALUATION OF ELECTRICAL TEST CONDITIONS IN MIL-M-38510 SLASH 5--ETC(U)

AUG 80 K SANDGREN

F30602-78-C-0193

UNCLASSIFIED

HAC-FR-80-76-706

RADC-TR-80-263

NL

1 of 2

20-A  
50301



AD A093215

LEVEL <sup>IV</sup> 12

RADC-TR-88-263  
Final Technical Report  
August 1980



# EVALUATION OF ELECTRICAL TEST CONDITIONS IN MIL-M-38510 SLASH SHEETS

Hughes Aircraft Company

K. Sandgren

DTIC  
ELECTE  
DEC 23 1980  
S C D

APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED

ROME AIR DEVELOPMENT CENTER  
Air Force Systems Command  
Griffiss Air Force Base, New York 13441

DTIC FILE COPY

80 12 22 174

This report has been reviewed by the RADC Public Affairs Office (PAO) and is releasable to the National Technical Information Service (NTIS). The NTIS will be interested in the general public, including foreign nations.

RADC TR-80-263 has been reviewed and is approved for publication.

APPROVED:

*W. Keith Conroy, Jr.*  
W. KEITH CONROY, JR.  
Project Engineer

APPROVED:

*David C. Luke*  
DAVID C. LUKE, Lt Colonel, USAF  
Chief, Reliability & Compatibility Division

FOR THE COMMANDER:

*John P. Huss*  
JOHN P. HUSS  
Acting Chief, Plans Office

#### SUBJECT TO EXPORT CONTROL LAWS

This document contains information for manufacturing or using munitions of war. Export of the information contained herein, or release to foreign nationals within the United States, without first obtaining an export license, is a violation of the International Traffic in Arms Regulations. Such violation is subject to a penalty of up to 2 years imprisonment and a fine of \$250,000 under 22 U.S.C 1778.

Include this notice with any reproduced portion of this document.

If your address has changed or if you wish to be removed from the RADC mailing list, or if the addressee is no longer employed by your organization, please notify RADC (NSRP) Office, APO NY 13441. This will assist us in maintaining a current mailing list.

Do not return this copy. Recycle or destroy.

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

19 REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER RADC-TR-80-263	2. GOVT ACCESSION NO. AD-A093	3. RECIPIENT'S CATALOG NUMBER 215
4. TITLE (and Subtitle) EVALUATION OF ELECTRICAL TEST CONDITIONS IN MIL-M-38510 SLASH SHEETS.		5. TYPE OF REPORT & PERIOD COVERED Final Technical Report August 78 - September 79
7. AUTHOR(s) K. Sandgren		6. PERFORMING ORG. REPORT NUMBER FR-80-76-706
9. PERFORMING ORGANIZATION NAME AND ADDRESS Hughes Aircraft Company Culver City CA 90203		8. CONTRACT OR GRANT NUMBER(s) F30602-78-0193
11. CONTROLLING OFFICE NAME AND ADDRESS Rome Air Development Center (RBRP) Griffiss AFB NY 13441		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS 62702F 23380157
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office) Same		12. REPORT DATE August 1980
15. SECURITY CLASS. (of this report) UNCLASSIFIED		13. NUMBER OF PAGES 110
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited		15a. DECLASSIFICATION DOWNGRADING SCHEDULE N/A
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report) Same		
18. SUPPLEMENTARY NOTES RADC Project Engineer: W. Keith Conroy, Jr. (RBRP)		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Microcircuits Specification Automatic Test Equipment Electrical Testing		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) Adequacy of MIL-M-38510 slash sheet requirements for electrical test conditions in an automated test environment were evaluated. Military temperature range commercial devices of 13 types from 6 manufacturers were purchased. Software for testing these devices and for varying the test conditions was written for the Tektronix S-3260 test system. The devices were tested to evaluate the effects of pin-condition settling time, measurement sequence of the same and different D-C parameters, -cont		

DD FORM 1 JAN 73 1473 EDITION OF 1 NOV 65 IS OBSOLETE

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

172300

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

*cont* → temperature sequence, differently defined temperature ambients, variable measurement conditions, sequence of time measurements, pin-application sequence, and undesignated pin condition ambiguity.

→ An alternative to current tri-state enable and disable time measurements is proposed; S-3260 "open" and "ground" conditions are characterized; and suggestions for changes in MIL-M-38510 slash sheet specifications and MIL-STD-883 test methods are proposed, both to correct errors and ambiguities and to facilitate the gathering of repeatable data on automated test equipment.

*V<sub>sub</sub> ICP* → Data obtained showed no sensitivity to measurement or temperature sequence nor to temperature ambient, provided that test times were not excessive. *V<sub>ICP</sub>* tests and some low current measurements required allowance for a pin condition settling time because of the test system speed. Some pin condition application sequences yielded incorrect measurements. Undefined terminal conditions of output pins were found to affect *I<sub>OS</sub>* and propagation delay time measurements. Truth table test results varied with test frequency and *V<sub>IL</sub>* for low-power Schottky devices. *I<sub>sub</sub> OS* *V<sub>sub</sub> IL* ↑

Accession for	<input checked="checked" type="checkbox"/>
RTIS GRA&I	<input type="checkbox"/>
DMC TAB	<input type="checkbox"/>
Unannounced	
Justification	
By	
Distribution/	
Availability Codes	
Dist	
Avail and/or	
Special	

*PI*

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

## TABLE OF CONTENTS

1.	INTRODUCTION.....	1
2.	DESCRIPTION OF STUDY .....	3
	Test Plan .....	3
	Plan for Investigating Sequence Sensitivities .....	6
3.	INITIAL PROGRAMMING AND TEST .....	8
4.	EFFECTS OF PIN CONDITION SETTLING TIME .....	12
	Test Description .....	12
	Results and Conclusions .....	14
5.	EFFECTS OF MEASUREMENT SEQUENCE OF DIFFERENT DC PARAMETERS .....	20
	Test Procedure .....	20
	Results and Conclusions .....	20
6.	EFFECTS OF TEMPERATURE SEQUENCE .....	23
	Test Procedure .....	23
	Results and Conclusions .....	23
	Additional Testing .....	27
7.	EFFECTS OF DIFFERENTLY DEFINED AMBIENTS .....	29
	Test Procedure .....	29
	VFF Measurement .....	30
	Results and Conclusions .....	30
8.	EFFECTS OF VARIABLE MEASUREMENT CONDITIONS ....	34
	Test Description and Procedure .....	34
	Results and Conclusions .....	35
9.	EFFECTS OF SEQUENCE IN TIME MEASUREMENTS .....	38
	Test Procedure .....	38
	Results and Conclusions .....	38
10.	EFFECTS OF MEASUREMENT SEQUENCE OF THE SAME (DC) PARAMETER .....	42
	Test Description and Procedure .....	42
	Results and Conclusions .....	42

## CONTENTS (Continued)

11.	EFFECTS OF PIN APPLICATION SEQUENCE . . . . .	45
	Test Description and Procedure . . . . .	45
	Results and Conclusions . . . . .	45
12.	EFFECTS OF UNDESIGNATED PIN AMBIGUITY . . . . .	47
	Test Description and Procedure . . . . .	47
	Results and Conclusions . . . . .	47
13.	INVESTIGATION OF MEASUREMENT EFFECTIVENESS . . . . .	49
	Tri-State Time Measurement Investigation . . . . .	49
	Maximum Frequency Test Investigation . . . . .	57
	Measurement of Input Capacitance . . . . .	59
14.	DEFINE "OPEN" AND "GROUND" . . . . .	60
	Test Description and Procedure . . . . .	60
	Results and Conclusions . . . . .	60
15.	SLASH SHEET REVIEW . . . . .	65
	Additional Conditions for $I_{OS}$ Measurements . . . . .	69
	Minimum or Maximum Pulse Width in Recommended	
	Operating Conditions in MIL-M-38510/9C . . . . .	73
	Output Loading in Figures for Measurements of	
	Time . . . . .	73
	Comparator Levels for Truth Table Tests . . . . .	73
	Fifty Percent Duty Cycles . . . . .	78
	$F_{MAX}$ Limits and Test Conditions . . . . .	78
	Flip-Flop Propagation Delay . . . . .	84
	Redundant or Superfluous Measurements and	
	Omitted Measurements . . . . .	85
	Typographical Errors and Omissions . . . . .	89
16.	SUGGESTIONS FOR MIL-STD-883 . . . . .	92
17.	CONCLUSIONS . . . . .	95

## LIST OF ILLUSTRATIONS

Figure		Page
1	S-3260 Microelectronic Test System . . . . .	2
2	Example of the Insertion of a Variable Delay from the Program Used for the 5400 Devices . . . . .	12
3	Large Variations in $I_{CEX}$ Readings at 54S74 Devices at 25°C . . . . .	13
4	Current Noise in S-3260 Test System, 1 $\mu$ A Range on System 1 . . . . .	14
5	Histogram of Pin Condition Settling Time Percent Change Data for $V_{OL}$ of 54S74 Devices at 25°C . . . . .	16
6	Histogram of Pin Condition Settling Time Percent Change Data for $I_{OS}$ of 54S00 Devices at 25°C . . . . .	17
7	Histogram of Pin Condition Settling Time Percent Change Data of $I_{IH5}$ of 54S74 Devices at 25°C . . . . .	18
8	Histogram of Pin Condition Settling Time Percent Change Data for $I_{IH5}$ of 54S74 Devices at 25°C . . . . .	19
9	Flow of Parameter Sequence Programs . . . . .	21
10	Temptronic TP450A ThermoStream® Unit . . . . .	24
11	$I_{IH1}$ of 54LS00 Devices at 125°C . . . . .	25
12	$I_{IH1}$ of 54LS00 Devices at 25°C after 30 second Stabilization Time from Previous Temperature . . . . .	26
13	Temperature of the DUT Versus Stabilization Time . . . . .	27
14	Chamber Surrounding DUT to Maintain Relative Humidity below 40 Percent . . . . .	29
15	A $t_{dif}$ Error is Included in Measurements of Time . . . . .	34
16	Histograms of $T_{PLH1}$ Data for 54S74 Devices at 25°C Using Standard (3.5 to 4 nanosecond) and 10 nanosecond Rise and Fall Times . . . . .	36
17	Chart of $V_{IL}$ Versus Truth Table Test Frequency at 125°C for 54LS295 Devices . . . . .	37
18	MIL-M-38510/9C, Page 71 . . . . .	41



# LIST OF ILLUSTRATIONS (Continued)

Figure		Page
19	Result of Dynamic Volt Measurement Dependent on Timing Characteristics of Comparator Window . . . . .	50
20	Theoretical Alternative to Tri-State Measurements . . . . .	51
21	ATE Model of Figure 20 . . . . .	52
22	Initial Waveforms for Suggested $t_{HZ}$ Measurement . . . . .	54
23	Initial Waveforms for Suggested $t_{ZH}$ Measurement . . . . .	55
24	S-3260 Socket Card with a 14-Pin Zero-Insertion-Force Dual-In-Line Socket Attached (Approximately 1/2 Scale) . . . . .	62
25	Page 1 of MIL-M-38510/2E Showing That the Absolute Maximum Ratings of the Device Types Require Withstanding an $I_{OS}$ Condition at Only One Pin for 5 Seconds . . . . .	70
26	MIL-M-38510/2E, Page 3, Table I, Note 4 Requires That No More Than One Output Be Shorted at One Time . . . . .	71
27	MIL-M-38510/2E, no Maximum Duration Specified for $I_{OS}$ and No Restriction Placed on Untested Outputs (No Notes Apply) . . . . .	72
28	MIL-M-38510/9C, Page 3 . . . . .	74
29	MIL-M-38510/306B, Page 2 . . . . .	75
30	From MIL-M-38510/9C, Page 55 . . . . .	76
31	From MIL-M-38510/9C, Pages 69 and 72 . . . . .	77
32	MIL-M-38510/22B, Page 28, Clock Pulse Specified as Having "50 Percent DUTY CYCLE, MIN" . . . . .	79
33	MIL-M-38510/22B, Page 38, $f_{MAX}$ Test Conditions Conflict with Other Switching Test Conditions . . . . .	80
34	From MIL-M-38510/71A, Pages 59 and 61, $f_{MAX}$ Conditions and Limits . . . . .	82
35	MIL-M-38510/51B, Page 40 . . . . .	83

# LIST OF ILLUSTRATIONS (Continued)

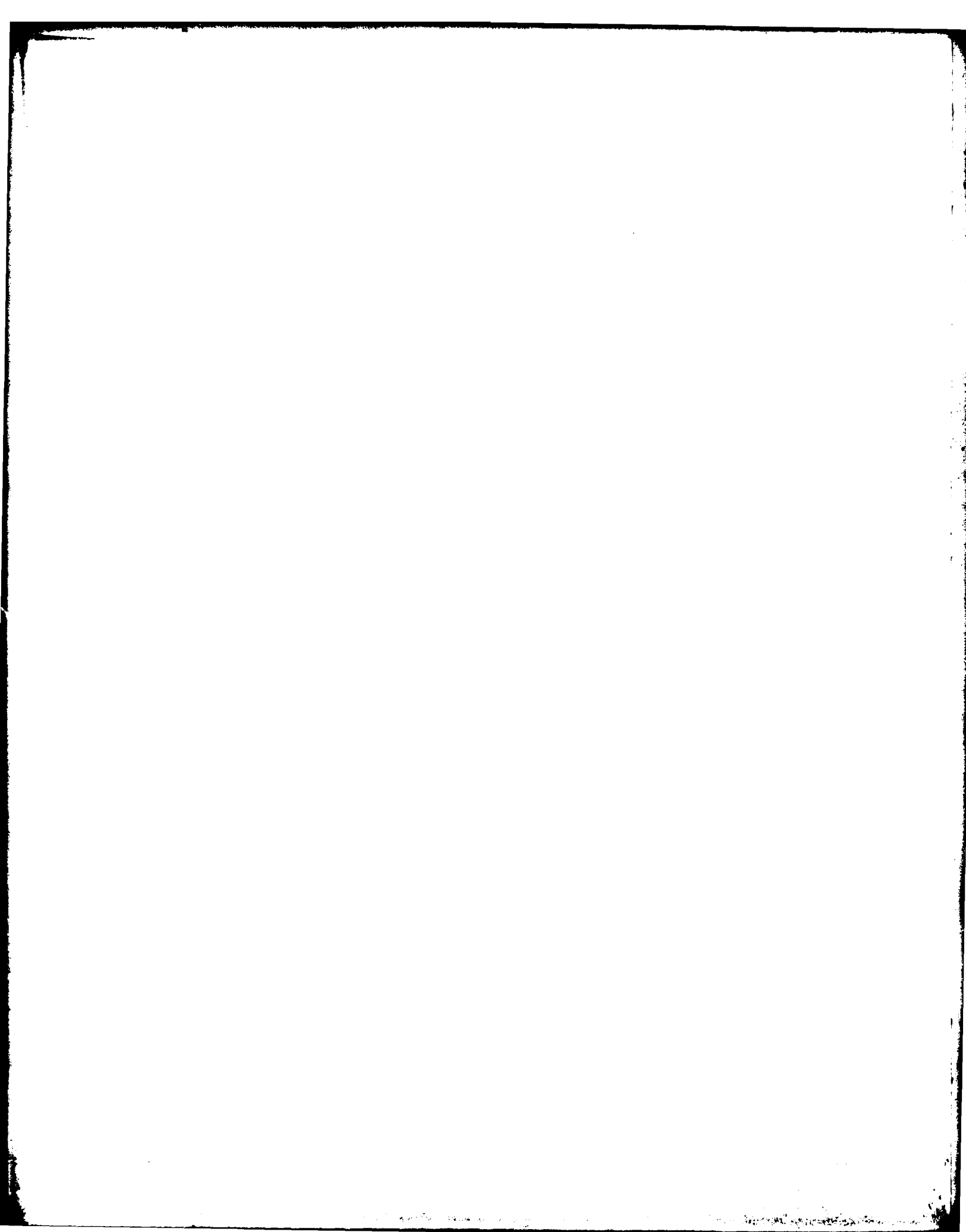
Figure		Page
36	Logic Diagram of a D-Type Flip-Flop (from MIL-M-38510/22B, p. 17) . . . . .	84
37	ISS Requirement for FET (from MIL-M-38510-50C, page 12) . . . . .	86
38	Schematic of CD4011A (MIL-M-38510/50C, p. 8) . . . . .	87
39	Logic Diagram of 54LS295 (MIL-M-38510/306B, Page 24) . . . . .	88

## LIST OF TABLES

Table		Page
1	Task or Study Description Table . . . . .	4
2	Device Types and Suppliers . . . . .	5
3	Test Matrix for Sequence of Ten Measurements . . . . .	6
4	Exceptions to MIL-M-38510 Test Conditions . . . . .	9
5	Data Comparison between Special Program and Initial Program . . . . .	10
6	Summary of Initial Test Results . . . . .	11
7	Typical Pin Condition Settling Time Data, 5400 Devices at 25°C . . . . .	15
8	Sequence Variation of Temperature-Sensitive Parameters . . . . .	22
9	Temperature Rise in Differently Defined Ambients . . . . .	31
10	Schottky Flip-Flop Parameters as Function of Environment after Stabilization (10 Minute Operation) . . . . .	33
11	Representative 54S74 Sequence in Time Data . . . . .	39
12	Representative 54LS295 Sequence in Time Data . . . . .	40
13	54164 Sequence of Same Parameter Data for $V_{OL}$ . . . . .	43
14	54LS00 Sequence of Same Parameter Data for $I_{IH1}$ . . . . .	43
15	54H74 Sequence of Same Parameter Data for $V_{IC}$ . . . . .	44
16	$V_{OH}$ Test Pin Conditions (D-Type Flip-Flop) . . . . .	45
17	Effect on $I_{OS}$ of Grounding More Than One Output . . . . .	48
18	Comparison of Suggested Tri-State Method to Specification Method . . . . .	56
19	Comparison of S-3260 $F_{MAX}$ Test to Specification Test . . . . .	59

# LIST OF TABLES (Continued)

Table		Page
20	Inductance of Ground Paths and Capacitance of the S-3260 . . . . .	61
21	Propagation Delay Times for Two Ground Path Inductances . . . . .	64
22	Procurement Specifications Used During This Study . . . . .	65
23	Identification and Location of Specification Errors . . . . .	66



## EVALUATION

The automatic electrical testing of microcircuits has forced a reevaluation of the JAN electrical specification to which devices are procured. Experience has shown that data obtained with automatic test equipment (ATE) very often differs significantly from that obtained on the bench, or even from other ATE. It was the purpose of this study to identify some of those differences and derive new testing procedures that will result in more accurate and repeatable parametric measurements on digital devices.

To this end, hypotheses were tested regarding the effects of test sequence, test condition settling times, and variations in unspecified test conditions. This involved writing computer programs for the test equipment so that several series of measurements could be repeated while carefully changing only one of the questionable parameters at a time, and observing the effects which were manifested in the volumes of data generated. Special analysis programs were therefore written for reducing this data. Also, the existing tri-state time measurement technique in one of the specs was found inadequate and a new, more accurate and more meaningful method was proposed.

The problem of how to define the ambient temperature of a device was addressed by performing experiments with different heating and cooling techniques. This data, combined with the application of thermodynamic principles involved in determining  $T_J$  and  $T_A$ , led to some proposed changes to MIL-STD-883, "Test Methods and Procedures for Microelectronics." This work required a great deal of insight into the operation of the ATE itself, such as knowing how the

equipment makes a measurement when the appropriate statement is encountered in the software, and understanding the parasitic elements of the test equipment's electrical measurement paths and nodes.

As a result of this work, improvements can be made not only to MIL-STD-883, but also MIL-M-38510, "Military Specification Microcircuits General Specification For," and the specific slash sheets reviewed in this effort. This impacts both the quality of the part procured to the spec, as well as the repeatability of data taken on different ATE systems when programmed to test to the same electrical spec.

This effort, however, covers only small and medium scale digital microelectronics. In order to fill the need for a detailed inspection and verification of JAN electrical specs, this work must be extended to cover linear and large scale complex microcircuits as well. This will contribute significantly to assuring the reliability of parts procured to the JAN spec system, which is one of the most important aspects of the R5B TPO thrust in Solid State Device Reliability.

*W. Keith Conroy, Jr.*  
W. KEITH CONROY, JR.  
Project Engineer

## 1. INTRODUCTION

This final report summarizes the results of a study completed by the Technology Support Division of Hughes Aircraft Company under Rome Air Development Center, Contract No. F30602-78-C-0193. The title of this study is "Evaluation of Electrical Test Conditions in MIL-M-38510 Slash Sheets." The test and evaluation period was from August 1978 to September 1979.

Integrated circuit test facilities typically use computer-controlled automatic test systems which perform hundreds of voltage, current, and time measurements in a few seconds. However, the systems in use have varying capabilities. Also, test engineers in different facilities do not always use the same test techniques. These differences have occasionally resulted in correlation problems between facilities. Factors that have caused problems include the following:

1. Undefined pin conditions, such as those resulting from failure to define the input or load for pins not included in the immediate test. (These pins are assumed to have no effect on test results.)
2. The sequence for the application and removal of bias supplies.
3. Such variables as test sequence, stimulus rise time, post-measurement settling time, chip heating as a function of test duration, and chip temperature control methods.

Many of these variables, which can potentially affect test results, are encountered only in the use of automatic test equipment (ATE). Thus, test specifications prepared before the widespread use of computer-controlled automatic test systems often fail to define some of these variables. Some existing MIL-M-38510 microcircuit specifications and MIL-STD-883 test methods fall into this category.

The principal objectives of this program have been to review existing MIL-M-38510 slash sheet electrical test conditions and evaluate the effects on parameter measurements of variations in test conditions where allowed in the specification. Such variables as undesignated pin options, measurement sequence, temperature environment, test set-up time, test recovery time, and temperature sequence were evaluated as well as the adequacy of



definitions of "open" and "ground" pin conditions. The specifications were also reviewed for errors and superfluous tests. These factors were evaluated for an automated testing environment where timing, test sequence, and other factors become important parts of the specification.

All of the tests were performed at the Hughes Aircraft Culver City facility on one of three Tektronix S-3260 automated IC test systems, illustrated in Figure 1. All test personnel contributing to the study are part of Hughes Technology Support Division, Components Laboratory.

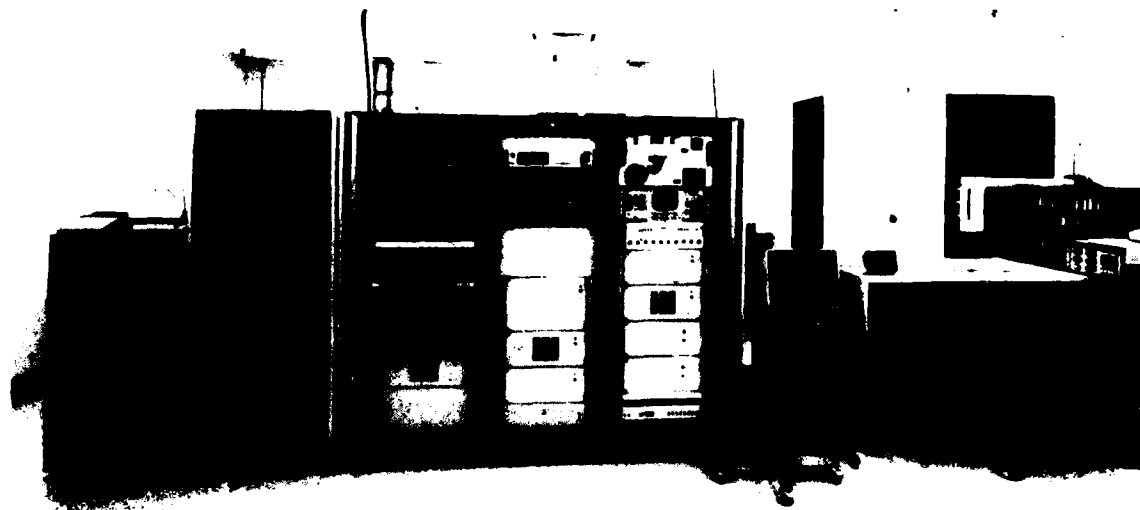


Figure 1. S-3260 Microelectronic test system

## 2. DESCRIPTION OF STUDY

The scope of this study includes the completion of 15 tasks as outlined in Table 1. Tasks 1, 2, and 3 are interrelated and required completion before any of the remaining 12 tasks or studies could be initiated.

### TEST PLAN

Task 1 required parts selection and procurement from vendors listed on the qualified parts list (QPL). Within the time frame of this study, 6 vendors were able to supply 15 samples of each device type required. In all, 13 device types, representing each of the major TTL device families and the CMOS devices currently used in military hardware, were tested during the course of this study. These devices include gates, flip-flops, and a shift register from each family. As Table 2 shows, 26 sets of 15 vendor parts were received for testing, for a total of 390 purchased parts.

All 390 purchased parts were submitted to initial electrical measurements at the three temperatures of  $-55^{\circ}\text{C}$ ,  $+25^{\circ}\text{C}$ , and  $+125^{\circ}\text{C}$ . Then 10 samples from each group of 15 parts were selected as test samples and used in the test method evaluation studies. Two samples from each group were retained as control samples and were not subjected to evaluation studies. The remaining three samples from each group were set aside for contingencies.

The 10 test samples from each of the 26 vendor sets formed the 260-sample test group for the remainder of the study. These parts were subjected to a series of tests designed to measure the effectiveness and adequacy of the applicable military specification in an automated test environment. The tests varied settling time, measurement sequence, temperature sequence, and different temperature ambients, and measured their effects.

TABLE 1. TASK OR STUDY DESCRIPTION TABLE

Task No.	Test Samples	Description of Task or Investigation	Temperature (°C)
1	15 per vendor	Parts Selection and Procurement	NA
2	NA	Initial Programming Effort	NA
3	390	Initial Electrical Tests Required by Slash Sheets	25, -55, +125
4	260	Effects of Pin Condition Settling Time	25
5	260	Effects of Measurement Sequence of Different Parameters	25
6	260	Effects of Temperature Sequence	25, -55, +125
7	260	Effects of Differently Defined Ambients	25, -55, +125
8	260	Effects of Variable Measurement Conditions	25, -55, +125
9	260	Effects of Sequence in Time Measurements	25
10	260	Effects of Measurement Sequence of the Same Parameter	25
11	260	Effects of Pin Application Sequence	25
12	260	Effects of Undesignated Pin Ambiguity	25, -55, +125
13	52	Effectiveness of Test Measurement	25
14	52	Defining Open and Ground	25
15	260	Effects of Measurement Times and Time between Measurements	25, -55, +125

TABLE 2. DEVICE TYPES AND SUPPLIERS

Part No.	RCA	SIG	NSC	FSC	TI	MOT
5400		X	X	X	X	X
54H00		X		X	X	X
54S00		X				
54LS00		X	X	X		
4011A	X		X			
5474				X		
54H74		X				
54S74		X				
54LS74					X	X
4013A	X		X			
54164		X				
54LS295					X	
4015A	X		X			

Note: RCA - RCA Corp, Solid State Div  
 SIG - Signetics Corp  
 NSC - National Semiconductor  
 FSC - Fairchild  
 TI - Texas Instruments  
 MOT - Motorola Semiconductor

## PLAN FOR INVESTIGATING SEQUENCE SENSITIVITIES

The investigation of sequence related sensitivities requires a large number of tests. Since the majority of tasks included in this study involve some sort of sequence investigation, the approach used is as follows:

1. Tests were run (or pins were tested) in groups of  $N$ , so that there would be no difference in program flow for different sequences.
2. Each test (or pin) appeared at least once in each position of a group (i.e., 1st, 2nd, 3rd, ... Nth).
3. Each test (or pin) was followed and preceded by each of the other tests (or pins) at least once.
4. A minimum number of tests were performed to minimize both data and test run times.
5. The sequences were generated automatically within the test program to facilitate data logging and reduction and to minimize programming errors.

An algorithm was developed which meets these sequencing requirements. It can be shown that for any even  $N$  tests performed in sequence, the minimum number of test runs is  $N$ , for a total of  $N^2$  tests. For odd numbers the number of test runs becomes  $N+1$ , for a total of  $(N+1)^2$  tests. Therefore, for a sequence of ten tests, the minimum matrix of tests would be as shown in Table 3, where each column represents a test run of ten items.

TABLE 3. TEST MATRIX FOR SEQUENCE OF TEN MEASUREMENTS

Sequence Number									
1	2	3	4	5	6	7	8	9	10
1	3	5	7	9	10	8	6	4	2
2	1	3	5	7	9	10	8	6	4
3	5	7	9	10	8	6	4	2	1
4	2	1	3	5	7	9	10	8	6
5	7	9	10	8	6	4	2	1	3
6	4	2	1	3	5	7	9	10	8
7	9	10	8	6	4	2	1	3	5
8	6	4	2	1	3	5	7	9	10
9	10	8	6	4	2	1	3	5	7
10	8	6	4	2	1	3	5	7	9

In order to use the same algorithm for both even and odd numbers, it becomes necessary to generate the matrix for the next higher even number and substitute the highest odd number for the highest even number. Thus, a matrix for the number 9 would be exactly the same as the Table 3 example except that all 10s would be replaced with 9s. This substitution generates some redundancies, but in order to conform to requirement 1, that all tests be run in groups of N, the step is unavoidable.

### 3. INITIAL PROGRAMMING AND TEST

In order to manipulate test sequences, as was required throughout this investigation, very flexible test software is necessary. Although programs were available in-house for 11 of the 13 device types used, the existing software for 8 of these was found not to be sufficiently flexible. Therefore, new programs were written for these 8 as well as for the 2 with no program, and extensive modifications were made to the remaining 3 programs. Because of test equipment limitations, some tests called out in the MIL-M-38510 slash sheets were modified or eliminated, as indicated in Table 4. The initial programs were written with minimized test set-up times. Minimum time between measurements was sought, but the constraints of the required measurement flexibility forced some compromises. A special minimum-time-between-measurements program for the 5400 (/00104) was written using "brute-force" techniques limiting the use of variables, loops, call statements, and conditional statements. Then the results were compared to the corresponding program used in this investigation. No significant measurement differences were encountered (see Table 5).

Fifteen devices per qualified vendor per slash sheet were purchased and tested to the conditions of the appropriate specification (as excepted in Table 4). Table 6 furnishes a summary of the initial test results. A zero denotes no failed devices for the parameter tested. As the data reveals, seven device types remained within limits throughout the testing. In addition, three CMOS device types remained within limits except that  $V_{ICP}$  tests required a delay for pin condition settling. Only three device types failed any of the parametric tests, and these failures were marginal. These parts were carefully included in the groups of ten parts per vendor per slash sheet so that changes in test method which might permit marginally failing parts to pass or marginally passing parts to fail might be more easily detected.

TABLE 4. EXCEPTIONS TO MIL-M-38510 TEST CONDITIONS

Test Type	Problem Preventing Compliance	Resolution	Slash Sheets Affected
1. Maximum Frequency Test	Machine clock rate capability is only 20.48 MHz. Faster requirements could not be met.	Where slash sheet requires higher clock rates, F max calculated from propagation delays, set up, hold, and minimum pulse width times.	/00903, rev C /00203, rev B /07101, rev A
2. Propagation Delay Times	Slash sheet rise, fall time requirements could not be met without modifying test equipment.	Existing system rise, fall times of 4 ns were used.	/07001 /07101, rev A /05001, rev B /05101, rev B
3. Propagation Delay Times	Slash sheet requires a pulse width of 5 ns on an input. System minimum is 8 ns.	8 ns used in place of 5 ns.	/07101, rev A
4. Input Capacitance Test	System does not have capability to measure input capacitance directly.	Test not done.	/05001, rev B /05101, rev B /05703, rev B
5. All Tests	54LS295B not available in time for study.	54LS295A used and input, output currents modified to limitations of this device type.	/30606, rev A



TABLE 5. DATA COMPARISON BETWEEN SPECIAL PROGRAM AND INITIAL PROGRAM

Parameter (5400-Type Device at 25°C)	Special Program (Minimum Time between Measurements)		Initial Program (Baseline)		Units
	Mean	Standard Deviation	Mean	Standard Deviation	
t <sub>PHL</sub>	9.888	0.787	9.885	0.784	ns
t <sub>PLH</sub>	9.291	0.748	9.296	0.753	ns
V <sub>OH</sub>	2.802	0.202	2.806	0.201	V
V <sub>OL</sub>	237.1	23.9	234.2	24.0	mV
I <sub>OS</sub>	-36.10	5.68	-36.08	5.65	mA
I <sub>IL</sub>	-1.067	0.074	-1.067	0.074	mA
I <sub>IH1</sub>	4.923	3.17	4.980	3.14	μA
I <sub>IH2</sub>	6.587	3.81	6.600	3.84	μA
I <sub>CCH</sub>	4.732	0.370	4.735	0.372	mA
I <sub>CCL</sub>	15.22	1.53	15.21	1.52	mA
V <sub>IC</sub>	-1.100	0.230	-1.095	0.231	V

TABLE 6. SUMMARY OF INITIAL TEST RESULTS

Device Type and Number of Type	Temperature	Test Parameters										
		Test Value	Ain Parameters			Output			Input			Notes
			$I_{L(1)}$	$I_{L(2)}$	Other	$V_{OH}$	$V_{OL}$	$I_{OS}$	$V_{IK}$	$I_{IH}$	$I_{IL}$	$I_{CC}/I_{CS}$
5400	+55	0	0	0	0	0	0	1.5	N/A	0	0	0
	-75	0	0	0	0	0	0	0	0	0	0	1
	125	0	0	0	0	0	0	0	N/A	0	0	0
5406	+55	0	0	0	0	0	0	0	N/A	0	0	0
	-75	0	0	0	0	0	0	0	0	0	0	0
5408	+55	0	0	0	0	0	0	0	N/A	0	0	0
	125	0	0	0	0	0	0	0	N/A	0	0	0
54806	+55	0	0	0	0	0	0	0	N/A	0	0	0
	-75	0	0	0	0	0	0	0	0	0	0	0
15	+55	0	0	0	0	0	0	0	N/A	0	0	0
	125	0	0	0	0	0	0	0	N/A	0	0	0
54LS10	+55	0	0	0	0	0	0	0	N/A	0	0	0
	-75	0	85	0	0	0	0	0	0	0	0	0
45	+55	0	0	0	0	0	0	0	N/A	0	0	0
	125	0	0	0	0	0	0	0	N/A	0	0	0
4011	+55	0	0	0	0	0	0	0	N/A	0	0	0
	-75	0	0	0	0	0	0	0	4	0	0	0
30	+55	0	0	0	0	0	0	0	N/A	0	0	0
	125	0	0	0	0	0	0	0	N/A	0	0	0
5474	+55	0	0	0	0	0	0	0	N/A	0	0	0
	-75	0	0	0	0	0	0	0	0	0	0	0
15	+55	0	0	0	0	0	0	0	N/A	0	0	0
	125	0	0	0	0	0	0	0	N/A	0	0	0
54H74	+55	0	0	0	0	0	0	0	N/A	0	0	0
	-75	0	0	0	0	0	0	0	0	0	0	0
15	+55	0	0	0	0	0	0	0	N/A	0	0	0
	125	0	0	0	0	0	0	0	N/A	0	0	0
54874	+55	0	0	0	0	0	0	0	0	0	0	0
	-75	0	0	0	0	0	0	0	0	0	0	0
15	+55	0	0	0	0	0	0	0	0	0	0	0
	125	0	0	0	0	0	0	0	0	0	0	0
54LS74	+55	0	0	0	0	0	0	0	N/A	0	0	0
	-75	0	0	0	0	0	0	0	0	0	0	0
30	+55	0	0	0	0	0	0	0	N/A	0	0	0
	125	0	0	0	0	0	0	0	N/A	0	0	0
4013	+55	0	0	0	0	0	0	0	N/A	0	0	0
	-75	0	0	0	0	0	0	0	4	0	0	0
30	+55	0	0	0	0	0	0	0	N/A	0	0	0
	125	0	0	0	0	0	0	0	N/A	0	0	0
54164	+55	0	0	0	0	0	0	0	N/A	0	0	0
	-75	0	0	0	0	0	0	0	0	0	0	0
15	+55	0	0	0	0	0	0	0	N/A	0	0	0
	125	0	0	0	0	0	0	0	N/A	0	0	0
41LS745	+55	0	0	0	0	0	0	0	N/A	0	0	0
	-75	0	0	0	0	0	0	0	0	0	0	0
15	+55	1.5%	0	0	0	0	0	0	N/A	0	0	0
	125	0	0	0	0	0	0	0	N/A	0	0	0
4015	+55	0	0	0	0	0	0	0	N/A	0	0	0
	-75	0	0	0	0	0	0	0	4	0	0	0
30	+55	0	0	0	0	0	0	0	N/A	0	0	0
	125	0	0	0	0	0	0	0	N/A	0	0	0

1. N/A = Not applicable

2. The measured base current, Motorola yielded a typical  $I_{BQ}$  value of 1.01 V (typ. specification limit 8.01 V (typ.)

3. The measured base current, all tested  $I_{BQ}$  at +55°C. Specification limits are 0.1 to 1.0 mA. Measured results are 0.1 to 1.0 mA.

4. 0.15 to 0.20 percent out of specification without insertion or a delay to allow for per cent out of specification.

5. Failures due to running test with  $V_{IL}$  = 0.7V and a FRR of 1 MHz

#### 4. EFFECTS OF PIN CONDITION SETTTLING TIME

##### TEST DESCRIPTION

Pin condition settling time is the time between establishment of measurement conditions and the time at which a measurement is made. In the case of  $I_{IH1}$  on 5400 devices, for example, all pin conditions except for those on the pin to be measured were set up.  $V_{CC}$  was forced with the appropriate voltage (5.5V) and the other inputs were grounded. The specified voltage (2.4V) was forced at the input pin at some time  $t_0$ . The measurement was then made at some later time  $t_1$ . The pin condition settling time is  $t_1 - t_0$ . The purpose of this study was to determine the effects of different pin condition settling times.

The baseline programs developed for the initial test were modified by inserting a variable delay between measurement condition setup and actual measurement (Figure 2). The S-3260 test systems used have a 1 ms minimum pin condition settling time for all measurements except the lowest range (100 ma) of current measurement. This system value is limited by

```

1.0100 * SN5400
1.0200 * QUADRUPLE 2 INPUT POSITIVE NAND GATE
:
4.9000 ARRA1 DLY(8)
4.9100 PRESEI DLY=0,1M,2M,4M,9M,99M,999M,9.999
:
8.4000 LOOP 190.13 N=1,8
:
44.0100 * I1H1
44.0200 *
44.0300 VS1=5.5V,200MA
44.0400 LOOP 44.16 X=1,8
44.0500 I1DRIVE=0.0V ON INS
44.0600 DISCONNECT INPUT FROM DRIVER ON INS(X)
44.0700 SETUP TO MEASURE CURRENT ON INS(X) FROM VS4=2.4V AT 100A
44.0710 WAIT DLY(N)*1S
44.0800 I1H1(X)=CURRENT
44.0900 UNSET TO MEASURE CURRENT ON INS(X) FROM VS4
44.1000 IF(I1H1(X) LT 100A)44.14
44.1100 SETUP TO MEASURE CURRENT ON INS(X) FROM VS4=2.4V AT 1000A
44.1110 WAIT DLY(N)*1S
44.1200 I1H1(X)=CURRENT
44.1300 UNSET TO MEASURE CURRENT ON INS(X) FROM VS4
44.1400 CONNECT INPUT TO DRIVER ON INS(X)
44.1600 CONTINUE
:
190.1200 CONTINUE
190.1300 CONTINUE
190.1400 SN=SN+1

```

Figure 2. Example of the insertion of a variable delay from the program used for the 5400 devices

the time that it takes for switching reeds and the measurement system to settle. This delay is part of the system hardware-software interface not subject to reduction by the test engineer. Delays of 0 ms, 1 ms, 2 ms, 4 ms, 9 ms, 99 ms, 999 ms, and 9.999 s were added to the inherent 1 ms delay of the test system to give measurement delays of 1 ms, 2 ms, 3 ms, 5 ms, 10 ms, 100 ms, 1 second, and 10 seconds. Ten parts per vendor per slash sheet were tested at each delay value, except 1 second and 10 seconds. Because of the long test times required at 1 and 10 seconds per measurement, only two parts per vendor per slash sheet were tested for these conditions. Also, input current measurements on CMOS devices were not tested. Because of delays inherent in the MC-1 low current measurement option on our S-3260 (delays which can reach 3 seconds per measurement), adding extra milliseconds was deemed superfluous and costly in terms of test time.

In evaluating the data for low current measurements (typically  $I_{IH1}$  for Schottky or low-power Schottky parts), wide measurement swings were observed (see Figure 3). The S-3260 tester appears to be responsible for

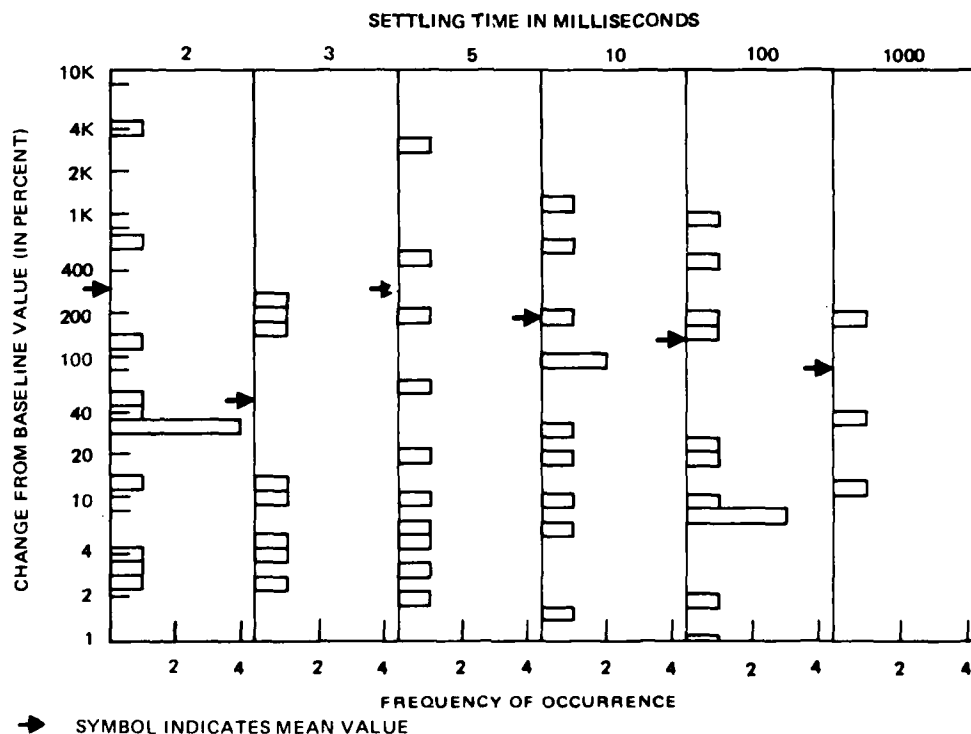


Figure 3. Large variations in  $I_{CEX}$  readings of 54S74 devices at 25°C

this inconsistent data. Repeatedly measuring current on a fixture with no device in the socket and plotting the current measured versus the delay in the measurement shows a noise with a frequency of between 120 and 150 Hz and an amplitude that depends on the current range and the particular S-3260 system used (see Figure 4). The amplitude was as high as 14.5 nA peak-to-peak on the 1  $\mu$ A range of one system, and as high as 9 nA peak-to-peak on the 100 nA range, whereas it was necessary to measure leakage currents of 2 nA. Fortunately, one system had much smaller variations (2 nA peak-to-peak). Using this system, with 5 nA as the significance guideline on the 100 nA range, the remainder of the data was taken without further difficulty.

## RESULTS AND CONCLUSIONS

The bulk of the data showed no dependence on pin condition settling time. Table 7 summarizes the data from the 5400-type devices where the mean values changed by less than 2 percent in all cases. It would be expected that measurements requiring significant power dissipation would be dependent on the length of time a forcing voltage or current was applied. While such

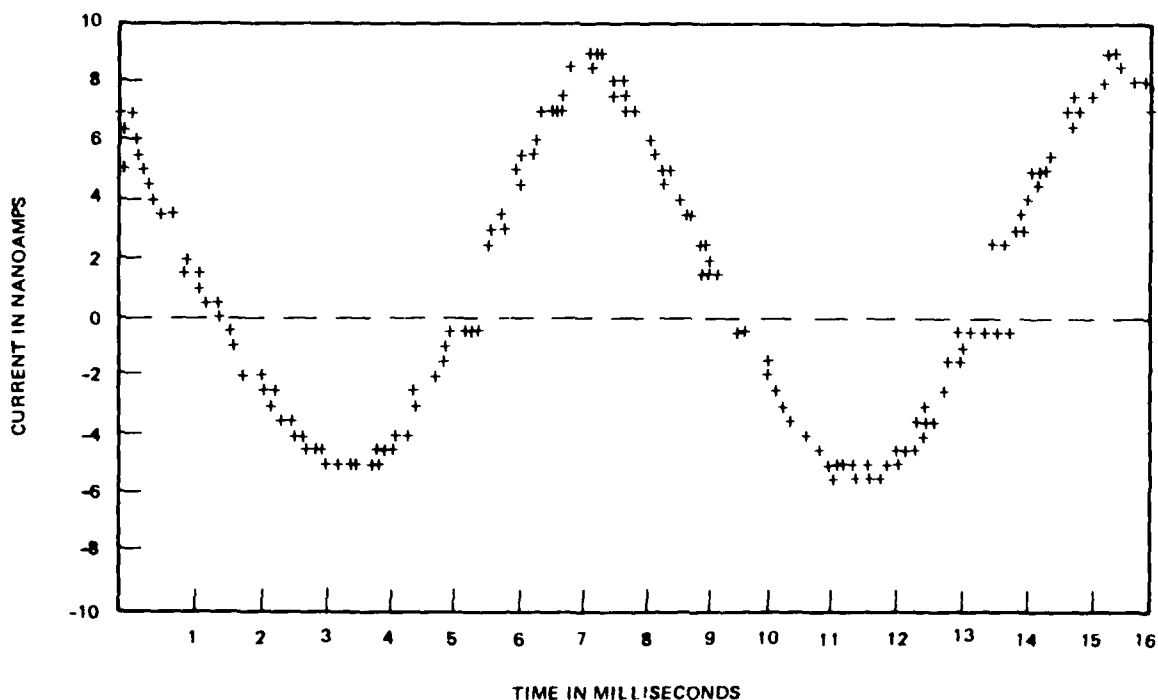


Figure 4. Current noise in S-3260 test system, 1  $\mu$ A range on system 1

TABLE 7. TYPICAL PIN CONDITION SETTLING TIME DATA, 5400 DEVICES AT 25°C

Delta	1mS	2mS	3mS	5mS	10mS	150mS	1s*	10s*	Units
Parameter	Mean Value	Mean Value	Mean Value	Mean Value	Mean Value	Mean Value	Mean Value	Mean Value	% Chng from lms
V <sub>CL</sub>	232.5	244.4	250.5	250.7	250.8	250.3	247.7	246.7	1.0 mV
V <sub>CH</sub>	2.743	2.742	2.743	2.743	2.793	2.744	2.553	2.557	0.2 V
I <sub>CS</sub>	36.37	36.39	36.36	36.36	36.34	36.24	36.46	36.34	0.5 mA
I <sub>TH1</sub>	4.53	4.54	4.55	4.56	4.57	4.56	4.80	4.89	2.0 $\mu$ A
I <sub>TH2</sub>	6.042	6.064	6.073	6.075	6.079	6.064	6.47	6.477	0.5 $\mu$ A
I <sub>TL</sub>	1.066	1.064	1.064	1.064	1.064	1.064	1.076	1.076	0.1 mA
I <sub>CCH</sub>	4.746	4.742	4.743	4.740	4.740	4.74	4.74	4.804	0.1 mA
I <sub>CCL</sub>	15.24	15.25	15.24	15.24	15.23	15.23	15.34	15.34	0.0 mA
V <sub>IC</sub>	1.102	1.102	1.102	1.102	1.102	1.101	1.079	1.078	0.1 V

\*These tests were run on only two devices. The percent change is calculated from the baseline measurements of the same two devices.

tendencies were observed in  $I_{OS}$  measurements and some  $V_{IC}$  and  $V_{OL}$  measurements, these parameters did not change as much as 3 percent, the value used as a significance guideline (Figures 5, 6, and 7).  $I_{IH}$  readings for Schottky parts rose significantly for longer delay times (Figure 8). This rise is attributed to device heating during prolonged testing. The  $I_{IH}$  results are nearly constant for pin condition settling times of 1 to 10 milliseconds. Limiting measurement to this range of pin condition settling time would ensure repeatable results. However, bench test methods require longer settling times and these longer times are more representative of devices in a hardware environment. This difference could be recognized by tightening the limits for  $I_{IH}$  and  $I_{CEX}$  when measured on ATE. Since measured  $I_{IH}$  values are well within specification over the entire temperature range, this approach would not greatly increase the number of rejected devices.

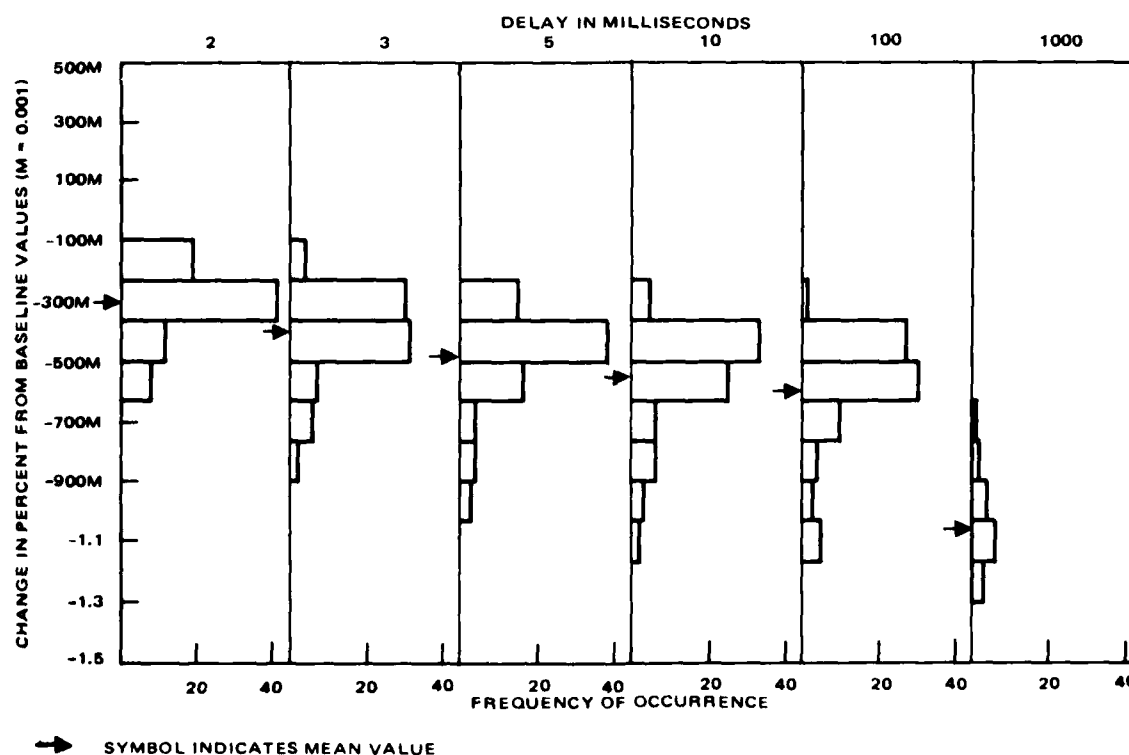


Figure 5. Histogram of pin condition settling time percent-change data for  $V_{OL}$  of 54S74 devices at 25°C

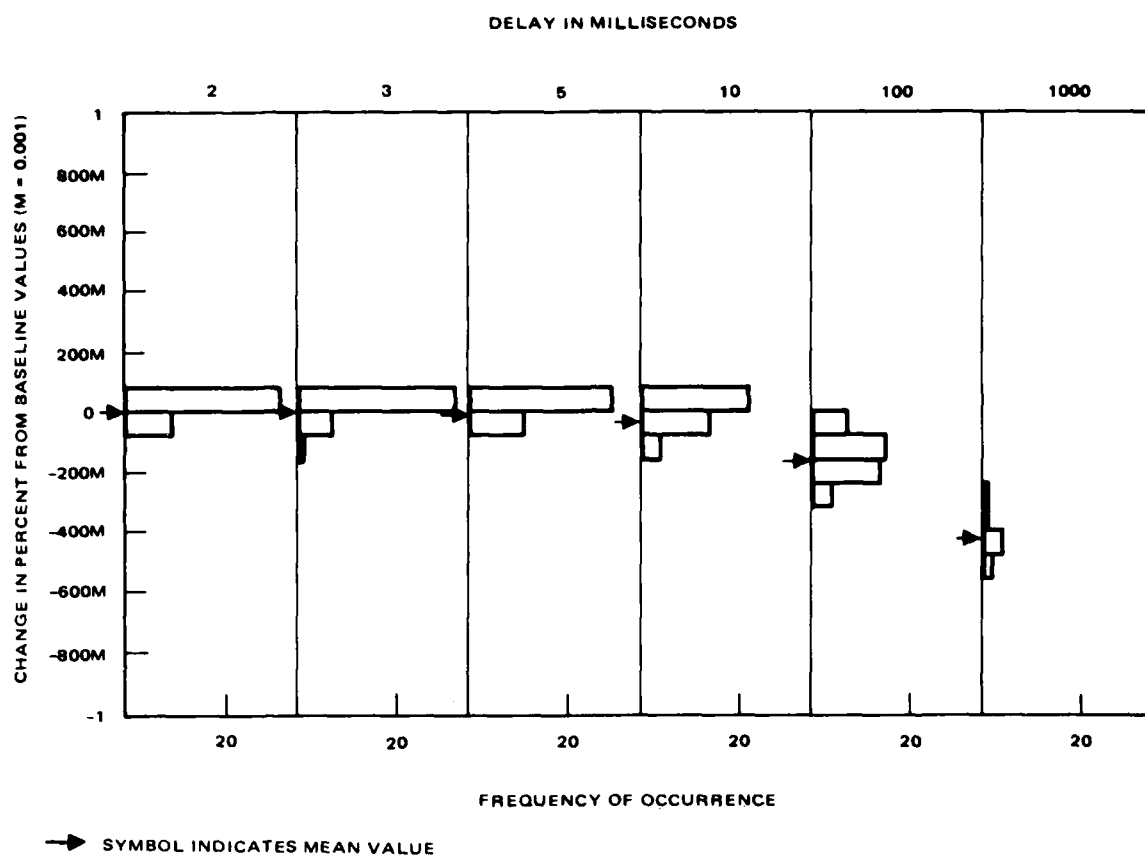


Figure 6. Histogram of pin condition settling time percent change data for  $I_{OS}$  of 54S00 devices at 25°C





Figure 7. Histogram of pin condition settling time percent change data for  $V_{IC}$  of 541b4 devices at 25°C

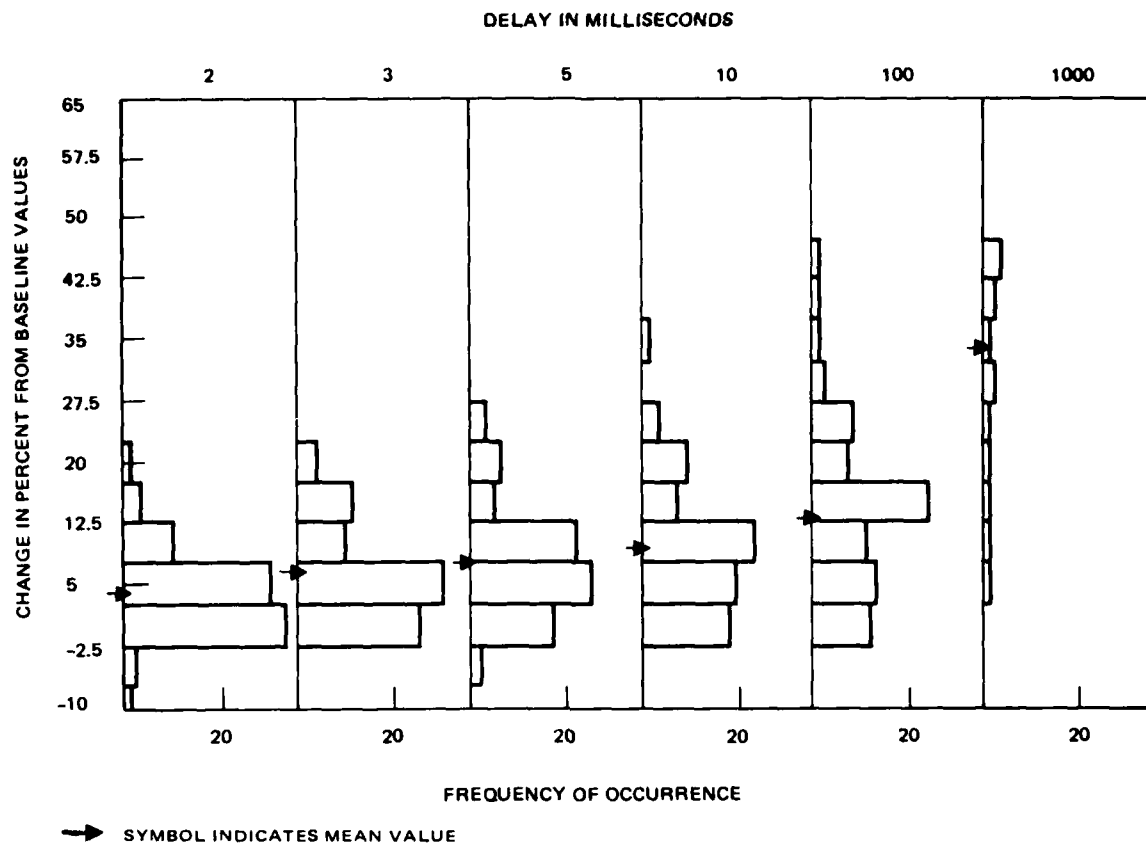


Figure 8. Histogram of pin condition settling time data for  $I_{IH5}$  of 54S74 devices at 25°C

## 5. EFFECTS OF MEASUREMENT SEQUENCE OF DIFFERENT DC PARAMETERS

### TEST PROCEDURE

This study was designed to determine whether the sequence in which various parameters are measured affects the measured value. For example, does the  $I_{IH1}$  value obtained when  $I_{IH1}$  is the first measurement performed differ from the value obtained when  $I_{IH1}$  is measured immediately after  $I_{OS}$ ? The initial test programs were modified for this study. Parameters were measured on a single representative pin, each parameter measurement was made into a subroutine, and the algorithm discussed in Section 2 was used to generate the sequences in which the parameters were called. Figure 9 diagrams the program flow for this study.

### RESULTS AND CONCLUSIONS

It might be expected that measurements sensitive to temperatures, when taken after measurements requiring high power dissipation (i. e.,  $I_{OS}$ ), would show a sequence sensitivity. This was not found to be the case. Table 8 shows that  $I_{CEX}$  and  $I_{IH}$  readings for each device under test (DUT) did not show a dependence on position with respect to  $I_{OS}$ . Indeed, no significant deviations were found during this portion of the study. Two factors which contribute to this sequence independence are the following:

1. Short measurement times and low duty cycles do not permit very much device heating, even during  $I_{OS}$  measurements.
2. ThermoStream<sup>®</sup> cooling (described in Section 6) minimizes device heating effects.

As was found in the pin condition settling time study (Section 4) and the differently defined ambient temperature study (Section 7), longer measurement times and a warmer thermal environment lead to device heating and a change in temperature sensitive parameters such as  $I_{IH}$  and  $I_{CEX}$ . To demonstrate this,  $I_{IH1}$  of the 54S00 devices was measured after  $I_{OS}$  tests in which the part was allowed 1 second for pin condition settling time during each  $I_{OS}$  measurement. The mean  $I_{IH1}$  values measured were 40 percent greater than when measured after  $I_{OS}$  measurements with 1 millisecond settling times.

MAIN PROGRAM

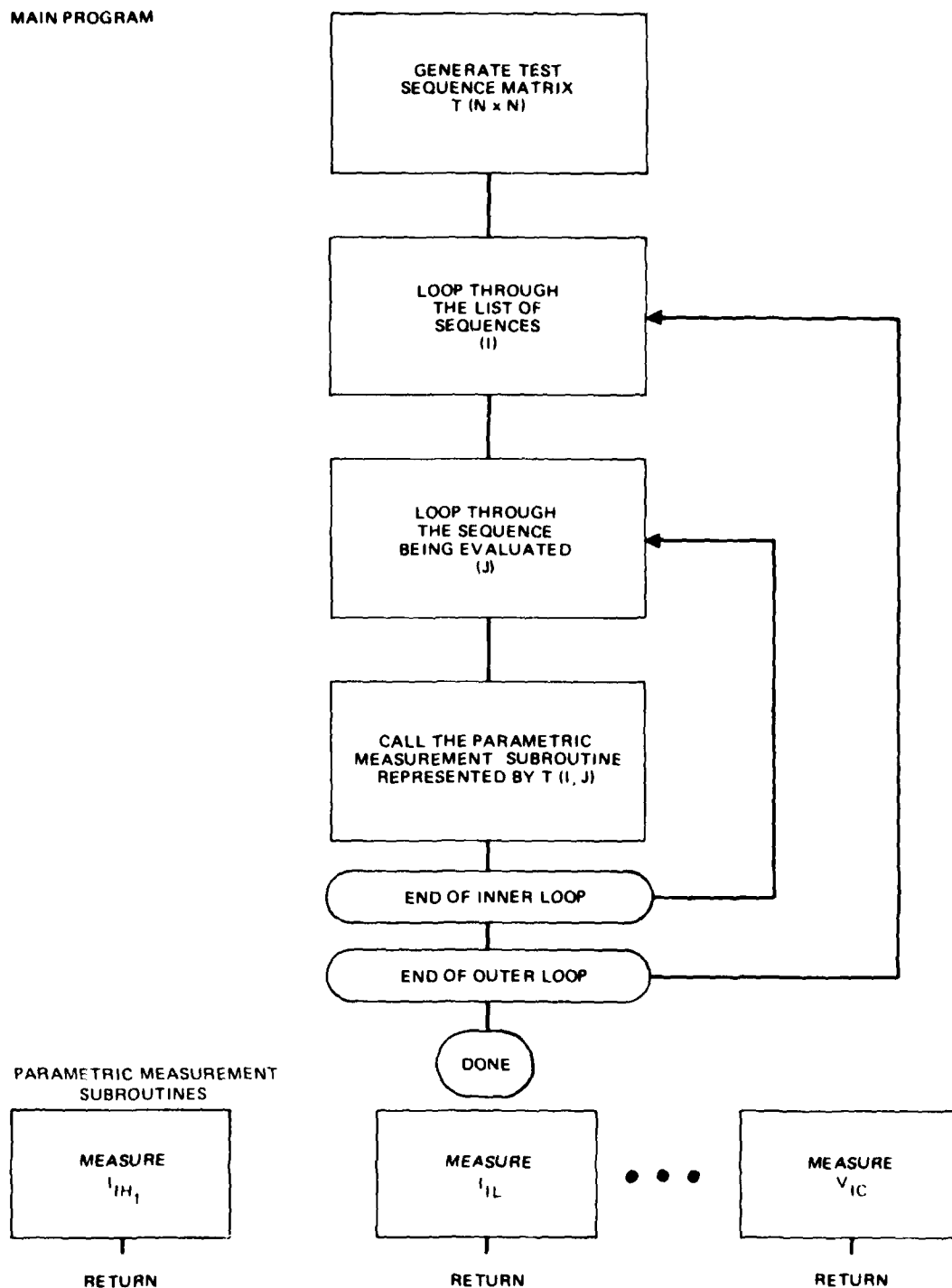


Figure 9. Flow of parameter sequence programs.

TABLE 8. SEQUENCE VARIATION OF TEMPERATURE - SENSITIVE PARAMETERS

Device Type	$I_{HH1}$			$I_{CEX}$		
	Measured Parameter			Measured Parameter		
	Mean Value when Measured First	Mean Value when Measured After $I_{OS}$	Mean Value when Measured First	Mean Value when Measured After $I_{OS}$	Mean Value when Measured After $I_{OS}$	
5400	4.54 $\mu A$	4.55 $\mu A$	N/A <sup>1</sup>	N/A	N/A	
54H00	5.83 $\mu A$	5.82 $\mu A$	N/A	N/A	N/A	
54S00	51.2 nA	49.7 nA <sup>2</sup>	6.95 $\mu A$	7.02 $\mu A$		
54LS00	16.2 nA	16.7 nA	N/A	N/A	N/A	
5474	2.35 $\mu A$	2.36 $\mu A$	N/A	N/A	N/A	
54H74	3.18 $\mu A$	3.17 $\mu A$	N/A	N/A	N/A	
54S74	51.9 nA	53.7 nA	5.87 $\mu A$	5.84 $\mu A$		
54LS74	15.5 nA	15.9 nA	N/A	N/A	N/A	
54164	10.3 $\mu A$	10.5 $\mu A$	N/A	N/A	N/A	
54LS265	17.6 nA	18.2 nA	N/A	N/A	N/A	

1 N/A - Not Applicable

2 When a 1-second delay is inserted in each  $I_{OS}$  test, this figure increases to 49.7 nA - a 40 percent increase.

## 6. EFFECTS OF TEMPERATURE SEQUENCE

### TEST PROCEDURE

Using the Temptronic TP450A ThermoStream® unit (Figure 10) to vary the temperature environment, all part types were tested as described in Section 3 at the three temperatures specified in the appropriate MIL-M-38510 slash sheet for the six combinations of three-temperature sequences. A programmed 30-second delay, accurate to within a few microseconds, was added to stabilize the DUT. The worst case temperature differential,  $+125^{\circ}\text{C}$  to  $-55^{\circ}\text{C}$ , was accomplished in less than 2 minutes under computer control. The high volume of air flow delivered by the ThermoStream® across the device under test maintains the temperature throughout the testing cycle. All temperature sequencing was also under computer control to ensure that all devices were tested under the same conditions.

### RESULTS AND CONCLUSIONS

With this test technique, CMOS input currents exhibited unusual behavior, but their behavior was not sequence related. Rather, it was almost random. The problem was isolated to a bad reed in the MC-1 low current measurement option, and the tests were repeated using a spare unit.

CMOS, Schottky, and low-power Schottky leakage current ( $I_{IN}$ ,  $I_{CEX}$ ) readings taken at  $125^{\circ}\text{C}$  during retest exhibited a dependence upon previous temperature (Figure 11). Readings at  $125^{\circ}\text{C}$  were approximately  $2^{\circ}$  warmer when preceded by  $25^{\circ}\text{C}$  measurement than when preceded by a  $-55^{\circ}\text{C}$  measurement. Increasing the stabilization time to 60 seconds reduced this dependence significantly. A similar difference was noted in  $25^{\circ}\text{C}$  readings. Those readings taken after  $125^{\circ}\text{C}$  readings showed larger leakage currents than those taken after  $-55^{\circ}\text{C}$  readings (Figure 12). Again, increasing stabilization to 60 seconds reduced the difference. The discrepancies were caused by the devices not reaching a stabilized temperature when allowed only a 30 second wait. Also, no differences were noted in the  $-55^{\circ}\text{C}$  data because the length of time required to cool to  $-55^{\circ}\text{C}$  with the Temptronic TP450A provided a built-in delay adequate to stabilize the device under test. The differences caused by inadequate time at temperature were the only differences detected.

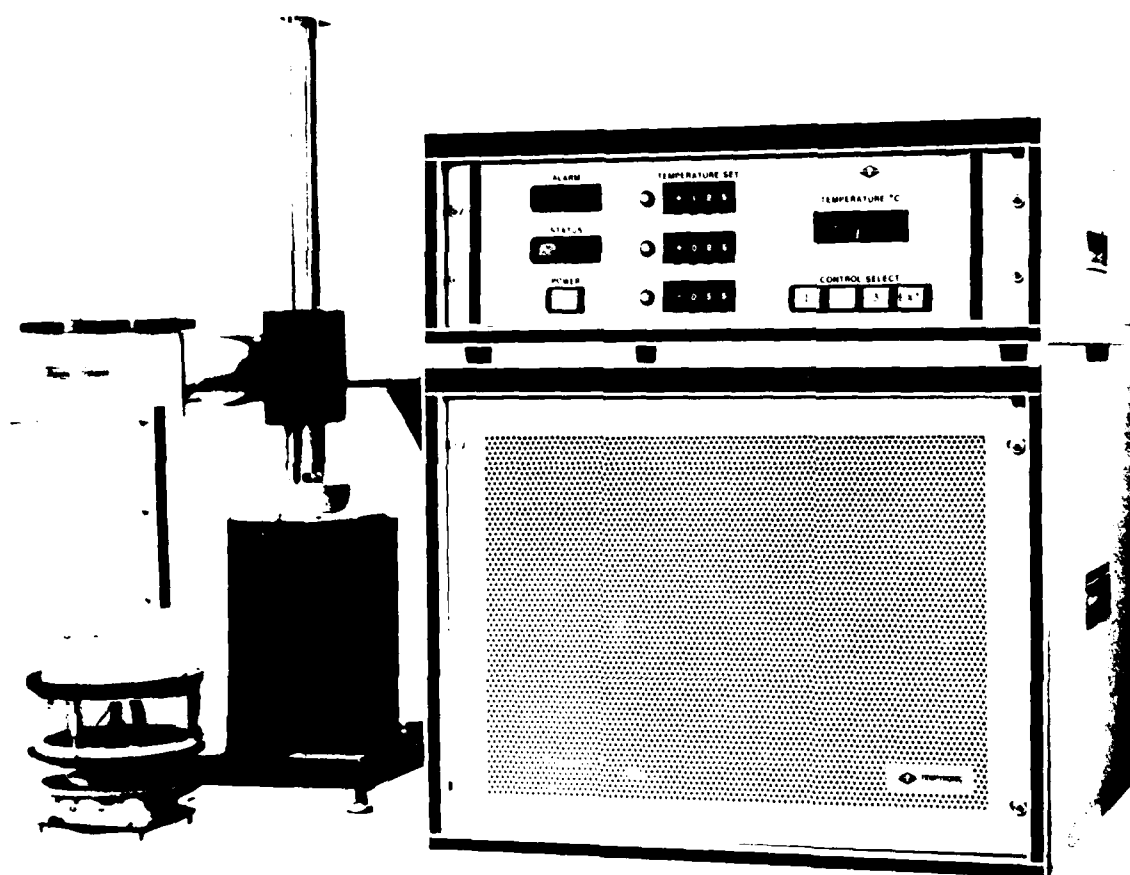


Figure 10. Temptronic TP450A ThermoStream<sup>®</sup> unit.

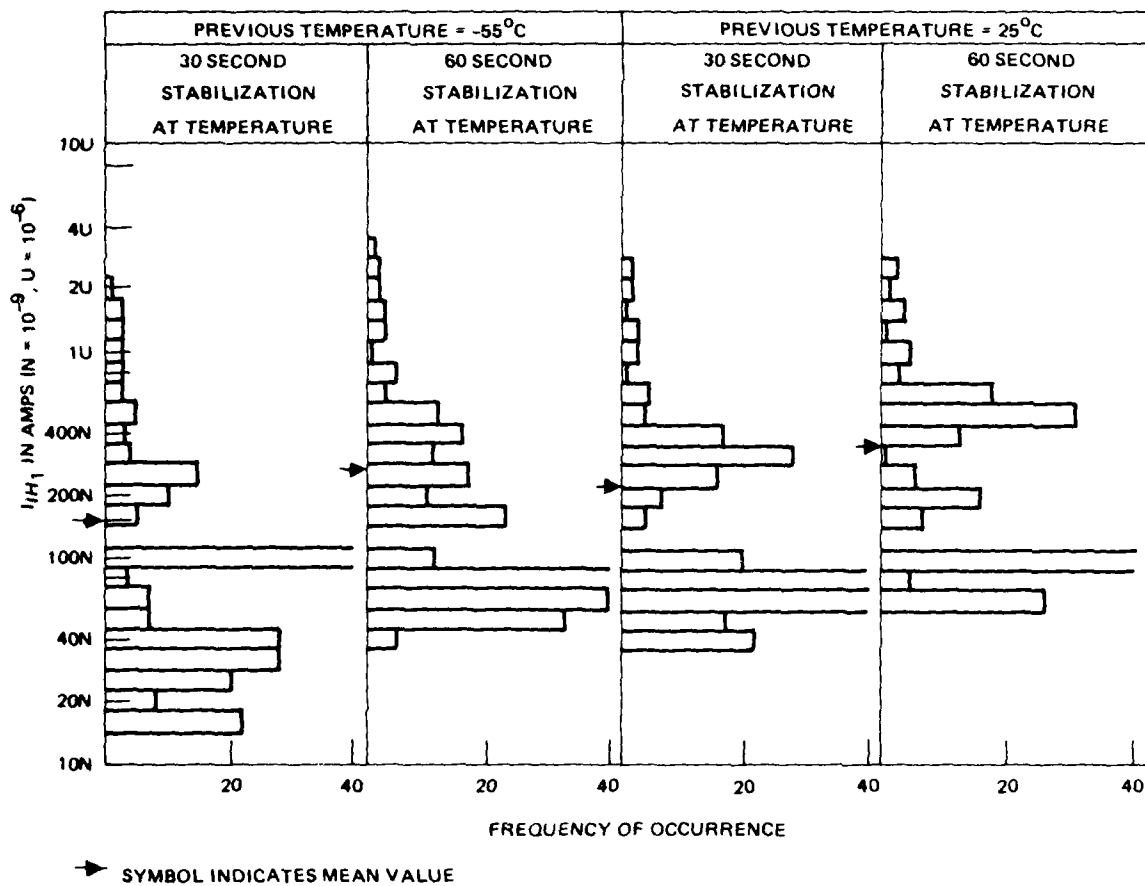


Figure 11.  $I_{IH1}$  of 54LS00 devices at  $125^{\circ}\text{C}$ .



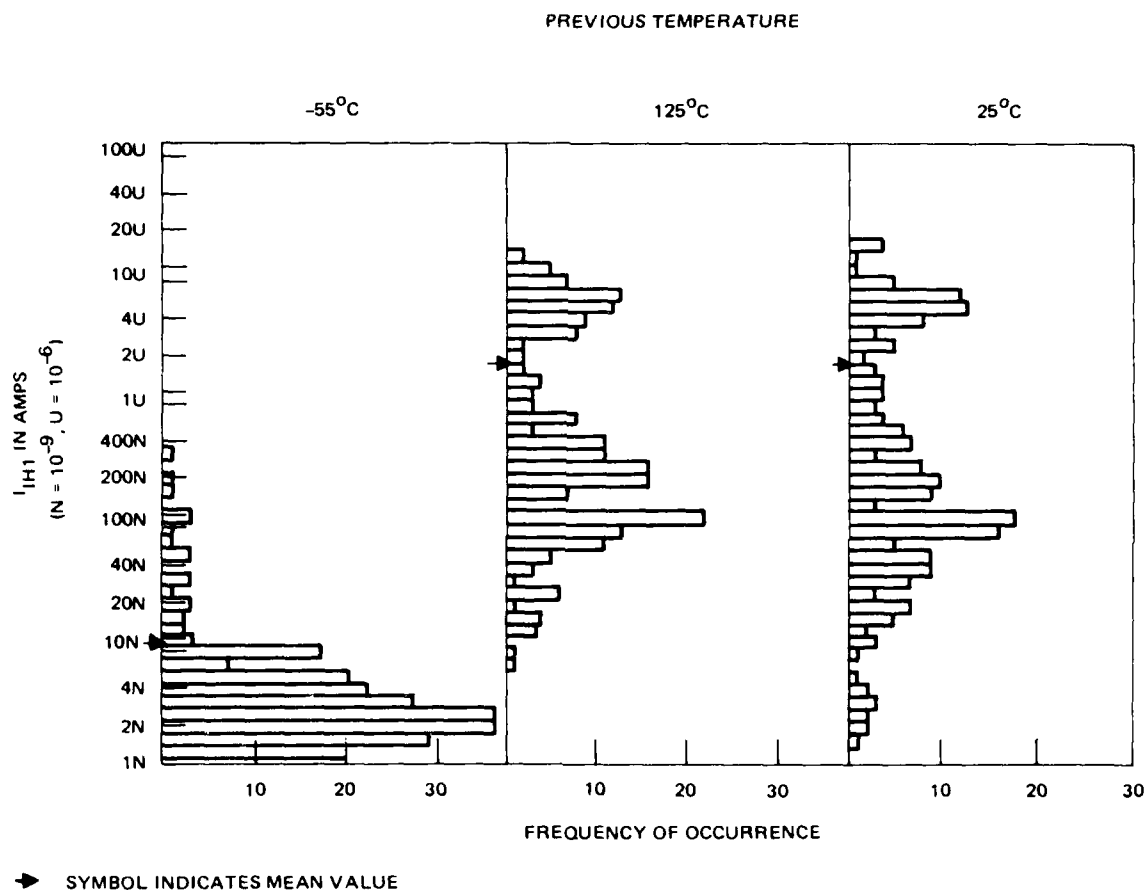


Figure 12.  $I_{H1}$  of 54LS00 devices at 25°C after 30 second stabilization time from previous temperature.

## ADDITIONAL TESTING

A short study to determine temperature differentials versus stabilization time by computing temperature from measured leakage current yielded the data shown in Figure 13. This method was used instead of the more accurate  $V_{FF}$  measurement to be described in Section 7 because large quantities of data had already been acquired which contained leakage currents but not  $V_{FF}$  data. The temperature was calculated from leakage currents rather than measured with a thermocouple because junction temperature, not case temperature, was desired. Since this data is very dependent on the thermal equipment used, its cooling and heating rates, and the flow rate of the airstream, the

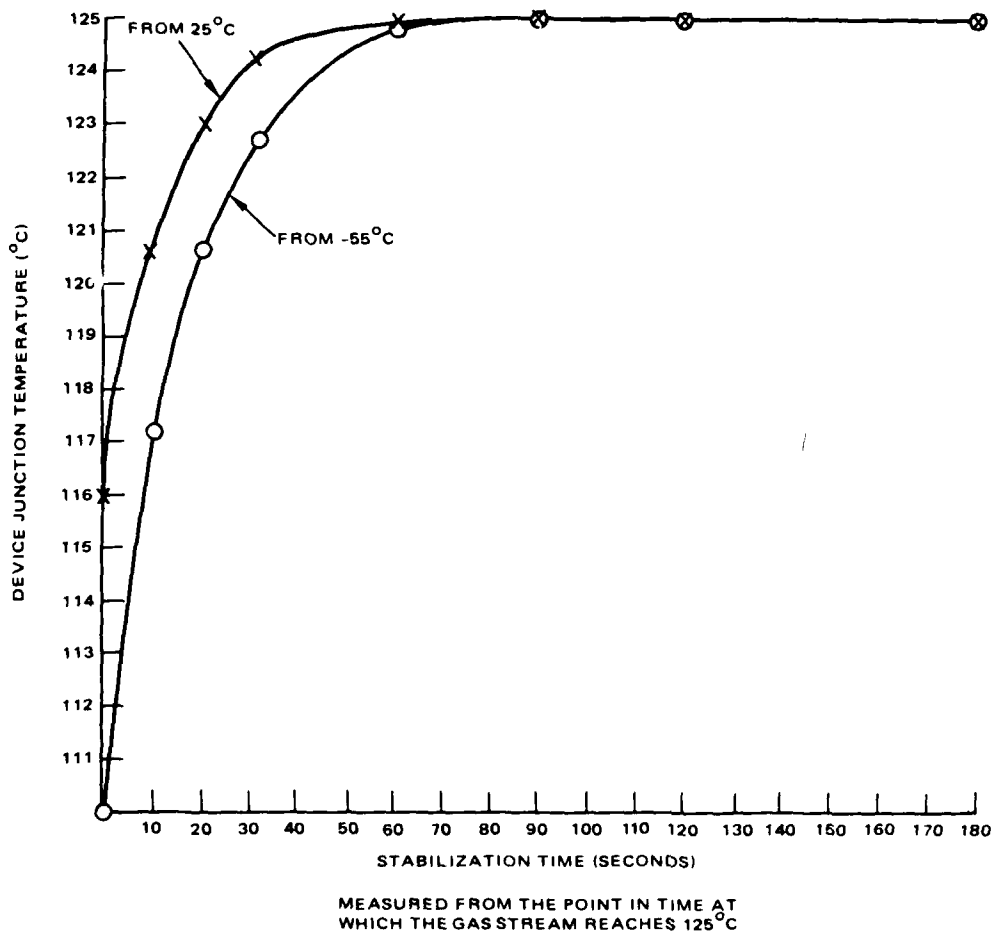


Figure 13. Temperature of the DUT versus stabilization time.

data is given for illustration only, and is not universally applicable. The flow rate of the TP450A was approximately  $3 \text{ ft}^3/\text{min}$  across the device.

## 7. EFFECTS OF DIFFERENTLY DEFINED AMBIENTS

### TEST PROCEDURE

The effects of the differently defined ambient temperature environments were studied. All samples were tested, using the initial test programs and the four temperature environments defined below:

- Method 1     Still air temperature. The temperature of the DUT was stabilized using moving air, and then shutting off the airflow and making the measurements.
- Maximum drift of the DUT temperature because of internal heating of the device is to be expected with this method. Relative humidity is kept below 40 percent by using dry nitrogen as the ambient atmosphere.
- Method 2     Moving air temperature (at least 50 ft/sec). This method uses a Temptronic TP450A ThermoStream® system with a dry nitrogen source to maintain relative humidity well below 40 percent. This method holds the temperature of the DUT closer to the ambient temperature called out in the test specification than Method 1. Some minor temperature drift occurs, depending on power dissipation, thermal resistance, and thermal time constant of the device.
- Method 3     Case temperature in still air. This method uses a thermal probe with thermal transfer cream applied to the probe tip for maximum thermal conductivity. Relative humidity is maintained below 40 percent by surrounding the DUT with a chamber that is purged by a small flow of dry nitrogen (see Figure 14). This method was used on two parts per vendor per slash sheet. Test results followed closely those from Method 1 when adequate soak time was given before shutting off the air flow.

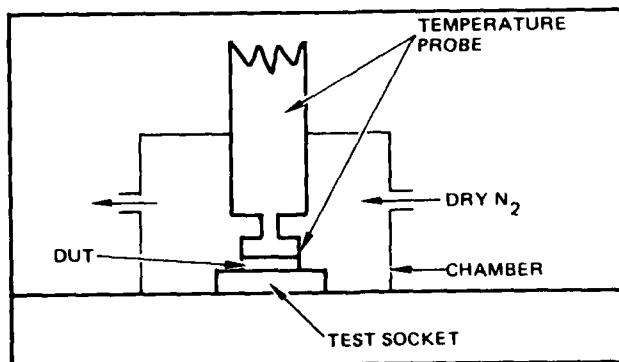


Figure 14. Chamber surrounding DUT to maintain relative humidity below 40 percent

Method 4 Case temperature in fluorocarbon. Especially for large volume testing, this is an alternative method for controlling  $T_C$  in still air. Because of the liquid-to-DUT interface, this method maintains DUT case temperature ( $T_C$ ) closer to specified  $T_A$  than the other three methods. Relative humidity in the fluorocarbon bath is negligible.

#### $V_{FF}$ MEASUREMENT

A modified  $V_{IC}$  measurement (designated  $V_{FF}$ ) was used to indicate the relative drift of DUT chip temperature away from specified  $T_A$  as a function of differently defined ambient temperatures.

$V_{FF}$  at an input pin ( $V_F$  of the input emitter-base diode, in parallel with  $V_F$  of the input clamp diode) varies with the internal temperature of the two diodes. Therefore, performing a  $V_{FF}$  measurement on one input pin of the DUT immediately before and immediately after the electrical test sequence gives an indication of how much the DUT internal temperature has drifted from the specified ambient temperature. To minimize internal heating from the  $V_{FF}$  measurement, a forcing current of 2 mA is used.

#### RESULTS AND CONCLUSIONS

Before examining the test results, calculations were performed to estimate the temperature rises expected from the study. These estimates assume a  $\theta_{JA}$  of  $20^\circ\text{C}/\text{W}$  for a 14 pin ceramic dual-in-line packaged device in fluorocarbon and  $\theta_{JA}$  of  $70^\circ\text{C}/\text{W}$  in still air. These values are tabulated in Table 9 along with the temperature values obtained using the  $V_{FF}$  measurements.

The experimental data are consistently smaller than the estimates made. This is because the DUT did not reach equilibrium temperature during test. If the DUT were continuously retested, the data from the  $V_{FF}$  measurements would more closely correspond to the estimates. This was verified for the 54S74s, which after 10 minutes of continuous test time closed to within  $0.5^\circ\text{C}$  of the estimated values. For these devices, when equilibrium temperatures were reached in still air, the temperature differentials are large enough to influence propagation delay time measurements as well (Table 10).

For the CMOS parts in this study, the temperature rise in any of the test environments is very small, and no greater than the differential arising

TABLE 9. TEMPERATURE RISE IN DIFFERENTLY DEFINED AMBIENTS

Device Type	Calculated Temperature Rises (°C)		Measured Temperature Rises (°C)			
	Still Air	Fluorocarbon	Method 1	Method 2	Method 3	Method 4
5400	4.23	1.21	0.4	0.2	0.4	0.2
54H00	5.92	1.69	0.8	0.2	0.9	0.2
54LS00	0.82	0.23	0.2	0.1	0.1	0.1
54S00	6.65	1.90	1.1	0.3	1.1	0.2
4011A	<0.1	<0.1	0.1	0.0	0.1	0.0
5474	5.95	1.70	1.0	0.4	0.9	0.3
54H74	10.8	3.10	1.8	0.7	1.7	0.4
54LS74	1.44	0.41	0.3	0.1	0.2	0.1
54S74	11.6	3.30	2.1	0.6	2.1	0.5
4013A	<0.1	<0.1	0.2	0.1	0.1	0.0
54164	11.7	3.35	2.0	0.7	1.9	0.5
54LS295	6.83	1.95	1.1	0.4	1.1	0.3
4015	<0.1	<0.1	0.1	0.0	0.1	0.0

from bringing the part to temperature on two different occasions and taking measurements. For the Schottky parts, however, the temperature differential between still air and the moving gas or fluorocarbon bath is sufficiently large to cause changes in leakage current measurements.

For the 54H00 and the 54H74, the temperature differential between devices tested in a still-air environment and those tested in a fluorocarbon bath should increase both thermal leakage currents and  $I_{IH}$ . However, thermal diode leakage is only a small part of  $I_{IH}$  on these gold doped devices and no measurements on these devices are sufficiently temperature sensitive for the temperature rise to cause a significant parameter-measurement change.

While the remaining parts exhibit some small temperature differential between still-air and fluorocarbon testing, these temperature differentials are too small to affect parameter measurements.

The data indicate that both fluorocarbon bath and gas-stream testing produce similar results and that still-air testing results in higher junction temperatures with the differential being dependent upon device power dissipation. For the simple devices (small scale or medium scale integration) of this study, small parameter measurement variations occur for Schottky parts. However, no LSI parts were studied. This data should not be extrapolated to parts with higher power dissipations, such as a Schottky PROM, or to parts where test times are sufficiently long to enable  $T_j$  to reach thermal equilibrium, as in a 16K RAM. Rather, additional data must be obtained. It has been our experience that the junction temperature differential between still-air and the fluorocarbon bath is sufficient in bipolar RAMS at  $-55^{\circ}\text{C}$  to cause parts to pass in still air and fail in the fluorocarbon bath.

TABLE 10. SCHOTTKY FLIP-FLOP PARAMETERS AS FUNCTION OF ENVIRONMENT AFTER STABILIZATION (10 MINUTE OPERATION)  
Temperature Ambient ( $T_A = 125^\circ\text{C}$ )

Parameter	Method 1 Still Air	Method 2 Thermo - Stream	Method 4 Fluoro- carbon	Fluorocarbon without 10 Min Device Operation	Units
$T_{PLH_1}$	8.36	7.96	8.32	8.16	ns
$T_{PLH_2}$	11.1	11.2	11.1	11.0	ns
$T_{PHL_2}$	9.48	8.86	9.00	8.97	ns
$T_{PHL_3}$	13.6	12.2	12.7	12.7	ns
$T_{PHL_4}$	8.12	7.50	7.71	7.64	ns
$V_{OL}$	297	300	299	300	mV
$V_{OH}$	3.021	2.955	2.940	2.939	V
$I_{OS}$	-69.25	-69.65	-69.55	-69.72	mA
$I_{IH_1}$	15.9	9.86	8.62	8.01	$\mu\text{A}$
$I_{IH_2}$	33.7	21.1	18.3	17.1	$\mu\text{A}$
$I_{IH_3}$	12.2	8.14	6.88	6.39	$\mu\text{A}$
$I_{IH_4}$	51.9	31.3	27.3	25.9	$\mu\text{A}$
$I_{IH_5}$	75.9	46.2	40.3	38.1	$\mu\text{A}$
$I_{IL_1}$	-1.46	-1.48	-1.49	-1.49	mA
$I_{IL_2}$	-3.84	-3.92	-4.01	-4.06	mA
$I_{IL_3}$	-2.86	-2.88	-2.90	-2.91	mA
$I_{IL_4}$	-2.80	-2.82	-2.83	-2.84	mA
$I_{CC}$	31.6	31.9	32.0	32.1	mA
$I_{CEX}$	2.34	1.53	1.32	1.21	$\mu\text{A}$



## 8. EFFECTS OF VARIABLE MEASUREMENT CONDITIONS

### TEST DESCRIPTION AND PROCEDURE

This task began with an inspection of the MIL-M-38510 slash sheets and a listing of all conditions that could have ambiguous interpretations. Three potential problems were identified for investigation:

1. Pulse characteristics, such as rise and fall times and set-up and hold times are often defined only by a maximum or minimum time (i. e.,  $t_r \leq 10 \text{ ns}$  or  $t_{\text{set-up}} \geq 20 \text{ ns}$ ).
2. Input variables for functional tests are defined over a range of values (i. e.,  $V_{IL} \leq 0.8 \text{ V}$ ).
3. Clock rate or pulse repetition frequency for the functional or truth table test (subgroups 7 and 8) is not specified.

Because of past experience with many of these conditions, it was apparent that testing at  $25^\circ\text{C}$  would not be sufficient to provide the needed information. Therefore, these tests were performed at all three temperatures.

Examining Figure 15 shows that propagation delay times can be affected by input rise and fall times, and by the value used to trigger comparators to start and stop the timing measurements. The difference in the timing measurement  $t_{\text{dif}}$  is the difference between the time at which the input waveform crosses the actual input threshold of each device and the time at which it

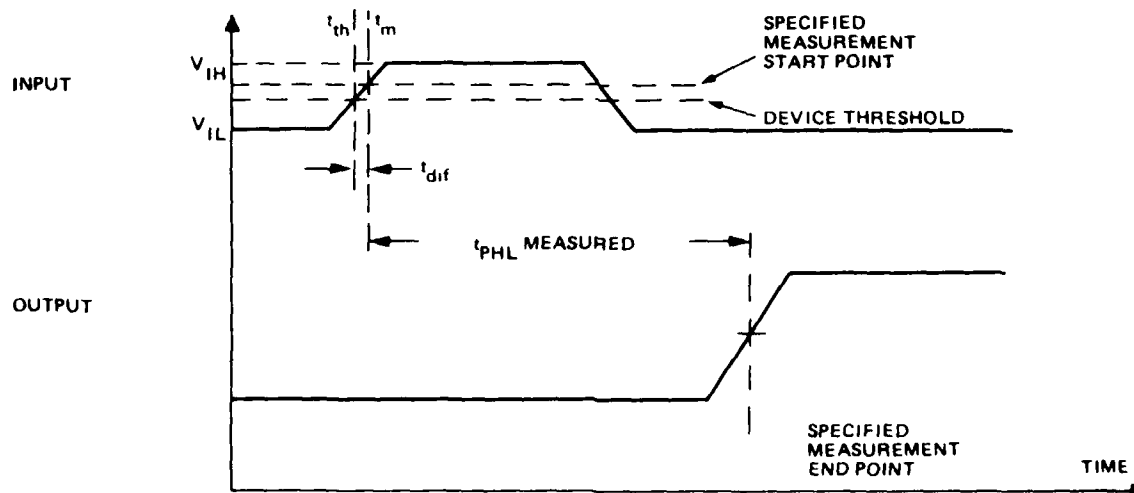


Figure 15. A  $t_{\text{dif}}$  error is included in measurements of time

crosses the assumed threshold upon which the measurement is based. This time difference is proportional to the voltage difference between actual and assumed threshold values and to the input rise or fall time. Since the rise and fall times are somewhat dependent on  $V_{IH}$  and  $V_{IL}$ , these values also become important.

Input rise and fall times were increased from 4 to 10 nanoseconds by capacitively loading the input drivers, and the devices were retested to the initial program. These results were then compared to the initial data.

Input variables for the truth table tests are loosely defined. Typically, input conditions such as  $V_{IH_{min}} \leq V_{IH} \leq V_{CC}$  and  $0 \leq V_{IL} \leq V_{IL_{max}}$  are permitted. Further, the pulse repetition rate is not specified. This allows results of truth table testing at pulse repetition rates approximating zero, using input signals with 0V to 5V swings, to be compared to results of testing with a pulse repetition rate (PRR) of  $f_{MAX}$  and input signal swings of  $V_{IL_{max}}$  to  $V_{IH_{min}}$ . To demonstrate the resulting differences, the devices were tested by varying  $V_{IH}$ ,  $V_{IL}$ , and PRR at the three temperatures.

## RESULTS AND CONCLUSIONS

Increasing the input rise and fall times from 4 to 10 nanoseconds did not affect the timing data significantly. Figure 16 shows initial data and the data generated using increased rise and fall times for 54S74 devices. A Schottky device was chosen for illustration since any change in  $t_{dif}$  becomes a significant percentage of a time measurement.

While 5400 and 54H00 devices proved insensitive to  $V_{IL}$ ,  $V_{IH}$ , and PRR variations, the 54S74 and 54LS295 devices demonstrated considerable sensitivity to  $V_{IL}$  and PRR at 125°C. A chart of  $V_{IL}$  versus frequency versus pass-fail at 125°C (Figure 17) illustrates that worst case testing ( $V_{IL_{max}}$ ,  $V_{IH_{min}}$ ,  $f_{MAX}$ ) will fail many devices which pass less severe nominal testing. To ensure repeatability,  $V_{IL}$ ,  $V_{IH}$ , and PRR must all be specified in narrower ranges. PRR could conveniently be specified either as 1 megahertz (which is an easily programmed value that is fast enough not to slow down test equipment and is standard for propagation delay measurements) or as the specified  $f_{MAX}$  limit if worst case is truly desired. In the latter case, it must be pointed out that many automated testers are not capable of performing the truth table test at  $f_{MAX}$ . Similarly,  $V_{IL}$  and  $V_{IH}$  must be specified over much tighter limits.

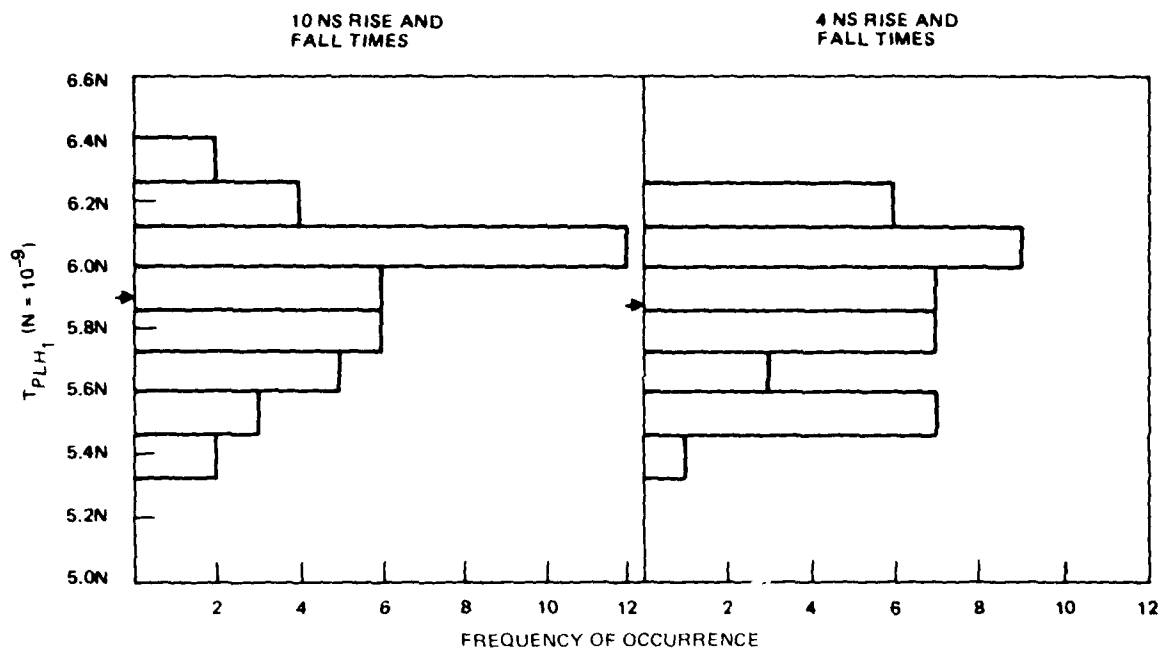


Figure 16. Histograms of  $T_{PLH1}$  data for 54S74 devices at 25°C using standard (3.5 to 4 nanosecond) and 10 nanosecond rise and fall times.

Since the truth table tests are the only tests in Table III of the specification that can indicate whether the device meets  $V_{ILmax}$  and  $V_{IHmin}$  requirements dynamically, specifying that input levels be  $V_{ILmax}$  and  $V_{IHmin}$  would provide such an indication and assurance. By specifying the sequence of the propagation delay time measurements of Table III, subgroup 9 of the specifications, the truth table of the device can be tested without using the subgroup 7 tests. Therefore, if it is not intended that a  $V_{IHmin}$  or  $V_{ILmax}$  requirement be met, the truth table test of subgroup 7 could be deleted as redundant provided that the sequence in which the subgroup 9 measurements are made is specified and that this sequence covers all state changes of the DUT.

LOW INPUT VOLTAGE ( $V_{IL}$ )	0.90	10	10	10	10	10	10
	0.85	10	10	10	8	6	5
	0.80	10	10	6	0	0	0
	0.75	10	10	1	0	0	0
	0.70	10	10	0	0	0	0
	0.65	10	8	0	0	0	0
	0.60	10	5	0	0	0	0
	0.55	10	1	0	0	0	0
	0.50	10	0	0	0	0	0
	0.45	10	0	0	0	0	0
	0.40	9	0	0	0	0	0
	0.35	9	0	0	0	0	0
	0.30	9	0	0	0	0	0
	0.25	8	0	0	0	0	0
	0.20	8	0	0	0	0	0
	0.15	7	0	0	0	0	0
	0.10	7	0	0	0	0	0
	0.05	7	0	0	0	0	0
	0.00	7	0	0	0	0	0
		48	64	80	100	125	150
							200

S 3260 CYCLE TIME IN NANoseconds

- NOTES
- 1 Dashed line denotes  $V_{IL}$  specification limit
  - 2 Ten devices were tested
  - 3  $V_{IH} = 2.0V$
  - 4 Entries are number of devices failing the condition

Figure 17. Chart of  $V_{IL}$  versus truth table test frequency at  $125^{\circ}C$  for 54LS295 devices

## 9. EFFECTS OF SEQUENCE IN TIME MEASUREMENTS

### TEST PROCEDURE

This study determined the effects of varying the order in which the timing measurements of Subgroup 9 in the specifications for flip-flops, counters, and a shift register are performed. The initial programs were modified by deleting the Subgroup 1 and Subgroup 7 tests and deactivating the temperature loops. The timing measurements were subroutined, and a comprehensive set of sequences of timing measurements were run.

### RESULTS AND CONCLUSIONS

No effects of sequence in time were found in the data. Table 11 illustrates the similarity of the data from different sequences run on 54S74s and Table 12 similarly illustrates the 54LS295 data.

While no significant changes resulted from a change in test sequence, it is not clear that the MIL-M-38510 slash sheets adequately specify the preconditioning required before each test is run. Nor does MIL-STD-883, Method 3003.1, Paragraph 3.1 clarify this problem. Without additional preconditioning information, the test conditions are almost meaningless. For example, in MIL-M-38510/9C, Page 71 (Figure 18), after running Test 165, all outputs are in a low state. Before Test 166 can be run,  $Q_B$  must be brought back to a high state so that a high-to-low transition can occur. The specification shows a transition in Figure 8, but nowhere specifies what preconditioning is required to restore  $Q_B$  to a logic 1, or that preconditioning is necessary. If a high state is present at all outputs before Test 166, Tests 166 through 172 can be run in sequence without further preconditioning. Running in other sequences requires additional preconditioning. Therefore, it would be efficient to specify the order in which Subgroup 9 tests are to be run to minimize the additional preconditioning information required.

TABLE 11. REPRESENTATIVE 54S74 SEQUENCE IN TIME DATA

Test Number	Sequence Number						
	1	4	7	10	13	16	19
88	5.52	5.49	5.56	5.51	5.54	5.50	5.56
89	6.20	6.17	6.19	6.23	6.21	6.25	6.19
90	5.88	5.84	5.86	5.88	5.93	5.92	5.85
91	6.04	6.04	6.03	6.07	6.06	6.01	6.07
92	8.17	8.15	8.17	8.16	8.16	8.14	8.12
93	10.30	10.34	10.28	10.30	10.35	10.31	10.33
94	7.62	7.58	7.60	7.59	7.60	7.61	7.64
95	8.89	8.92	8.88	8.86	8.89	8.90	8.89
96	8.78	8.81	8.76	8.74	8.79	8.78	8.77
97	8.91	8.86	8.96	8.94	8.88	8.90	8.88
98	9.61	9.58	9.64	9.58	9.62	9.62	9.59
99	9.37	9.36	9.40	9.35	9.39	9.34	9.35
100	6.62	6.66	6.62	6.64	6.62	6.60	6.65
101	7.78	7.79	7.78	7.74	7.81	7.77	7.77
102	9.05	9.09	9.07	9.02	9.06	9.08	9.07
103	10.76	10.77	10.80	10.75	10.76	10.81	10.78
104	6.25	6.23	6.28	6.28	6.28	6.26	6.27
105	7.13	7.12	7.11	7.15	7.09	7.10	7.10
106	8.85	8.84	8.81	8.82	8.87	8.83	8.82
107	11.14	11.17	11.17	11.16	11.10	11.10	11.16

All data in nanoseconds.  $T_A = 25^{\circ}\text{C}$

TABLE 12. REPRESENTATIVE 54LS295 SEQUENCE IN TIME DATA

Test Number	Sequence Number							
	1	5	9	13	17	21	25	29
81	23.6	23.3	23.4	23.6	23.6	23.4	23.5	23.5
83	25.0	24.4	24.7	24.4	24.9	25.0	24.6	24.7
85	23.4	23.5	23.2	23.6	23.4	23.4	23.4	23.5
87	23.5	23.2	23.3	23.7	23.3	23.2	23.5	23.4
89	22.0	23.2	22.9	22.7	22.2	22.3	22.4	22.8
91	22.0	23.1	22.8	22.6	22.2	22.3	22.4	22.7
93	21.4	22.7	22.3	22.2	21.8	21.8	21.9	22.2
95	21.2	22.7	22.1	22.2	21.9	21.8	21.8	22.0
97	25.7	26.1	25.2	26.6	26.1	26.6	25.3	26.0
99	25.8	26.3	25.4	26.7	26.2	26.7	25.3	26.1
101	13.9	13.7	13.8	13.8	13.8	13.6	13.8	13.6
103	13.8	13.8	13.7	13.9	13.8	13.8	13.7	13.7
105	24.7	26.3	24.5	23.1	26.3	26.6	25.4	23.2
107	23.6	25.2	23.5	22.2	25.3	25.5	24.3	22.2
109	20.5	21.1	20.6	20.3	20.3	20.1	20.1	20.9
111	20.4	20.3	20.3	20.2	20.2	20.2	20.0	20.4

All data in nanoseconds.  $T_A = 25^{\circ}\text{C}$

TABLE III Group A inspection for device type 03 - Continued  
Terminal conditions (pins not designated are open, high level or low level)

Subgroup	Symbol	MIL-STD-883 method	Case A, B, C, D	1	2	3	4	5	6	7	8	9	10	11	12	13	14	Test limits
				Input serial	Input serial	QA	QB	QC	QD	GND	Clock	Clear	QE	QF	QG	QH	VCC	Min. Max. Unit
7	Truth table test	3014		A	B	H	H	H	H	GND	A	A	H	H	H	H	4.5 V	
3				B	B	L	L	L	L		B	B	L	L	L	L		
6																		
129				A	A	H	H	H	H	GND	A	A	H	H	H	H	4.5 V	
130											B	B	L	L	L	L		
131											A	A	H	H	H	H		
132											B	B	L	L	L	L		
133											A	A	H	H	H	H		
134											B	B	L	L	L	L		
135											A	A	H	H	H	H		
136											B	B	L	L	L	L		
137											A	A	H	H	H	H		
138											B	B	L	L	L	L		
139											A	A	H	H	H	H		
140											B	B	L	L	L	L		
141											A	A	H	H	H	H		
142											B	B	L	L	L	L		
143											A	A	H	H	H	H		
144											B	B	L	L	L	L		
145											A	A	H	H	H	H		
146											B	B	L	L	L	L		
147											A	A	H	H	H	H		
148											B	B	L	L	L	L		
149											A	A	H	H	H	H		
150											B	B	L	L	L	L		
151											A	A	H	H	H	H		
152											B	B	L	L	L	L		
153											A	A	H	H	H	H		
154											B	B	L	L	L	L		
155											A	A	H	H	H	H		
156											B	B	L	L	L	L		
157											A	A	H	H	H	H		
158											B	B	L	L	L	L		
159											A	A	H	H	H	H		
160											B	B	L	L	L	L		
161											A	A	H	H	H	H		
162											B	B	L	L	L	L		
163											A	A	H	H	H	H		
8	Repeat subgroup 7 at $T_A = -125^\circ\text{C}$ and $T_A = -55^\circ\text{C}$			IN	IN	OUT	OUT	OUT	OUT	GND	IN	4.5 V				OUT	5.0 V	11 12 4
9	$t_{MAX}$ (Fig. 8)			IN	IN	OUT	OUT	OUT	OUT	GND	IN	4.5 V				OUT	5.0 V	11 12 4
164																		
165	$t_{PHL1}$ (Fig. 8)																	
166																		
167																		
168																		
169																		
170																		
171																		
172																		

Figure 18. MIL-M-38510/9C, page 71.



## 10. EFFECTS OF MEASUREMENT SEQUENCE OF THE SAME (DC) PARAMETER

### TEST DESCRIPTION AND PROCEDURE

In Section 5, the effects of measurement sequence of different DC parameters are evaluated. In that study only a single pin was measured for each parameter. In this study the sequence in which the different parameters were measured remained as in the initial programs, but for each parameter the sequence in which appropriate pins were measured was varied. The resultant data was analyzed to identify any measurement dependence upon the sequence.

### RESULTS AND CONCLUSIONS

No sequence sensitivities were encountered. Tables 13 through 15 give data for some parameters for the 54LS00, 54164, and 54H74 devices. As a note of caution, however, if  $V_{OH}$ ,  $V_{OL}$ , or  $I_{OS}$  sequences are altered, preconditioning may be required just as in the propagation delay time sequences discussed in Section 9. Without this preconditioning, parts will fail  $V_{OH}$  and  $V_{OL}$  tests if they are in the wrong state. Although they may pass the  $I_{OS}$  test, the parameter measured will not be  $I_{OS}$  since the part will be in the wrong output state. Since some specification sequences do not require preconditioning and many other sequences do, it is important to specify that  $V_{OH}$ ,  $I_{OS}$ , and  $V_{OL}$  tests be done in sequence in order to provide the required preconditioning.

TABLE 13. 54164 SEQUENCE OF SAME PARAMETER DATA FOR  $V_{OL}$ 

Pin	Sequence Number							
	1	2	3	4	5	6	7	8
$Q_A$	211.8	215.4	212.4	211.4	211.5	213.7	213.3	212.1
$Q_B$	209.4	214.3	211.4	210.9	210.4	212.1	211.9	210.3
$Q_C$	211.7	218.3	210.0	211.7	211.7	213.4	213.1	212.4
$Q_D$	211.4	219.8	212.2	212.9	212.6	211.8	212.8	211.9
$Q_E$	211.4	216.3	211.8	210.7	212.3	213.2	211.7	213.1
$Q_F$	212.2	216.7	213.1	212.5	212.2	211.8	214.9	213.3
$Q_G$	212.7	216.6	212.8	211.9	212.0	212.9	214.2	214.6
$Q_H$	213.8	218.4	212.6	212.1	215.6	213.7	214.3	215.2

All data in mV.  $T_A = 25^{\circ}\text{C}$

TABLE 14. 54LS00 SEQUENCE OF SAME PARAMETER DATA FOR  $I_{III_1}$ 

Pin	Sequence Number							
	1	2	3	4	5	6	7	8
1A	2.23	2.33	2.25	2.28	2.18	2.28	2.38	2.23
1B	1.70	2.03	1.78	1.83	1.80	1.73	1.98	2.05
2A	2.18	2.10	2.20	2.25	2.15	2.20	2.28	2.23
2B	1.90	2.05	2.15	2.18	1.90	1.93	2.00	1.99
3A	2.00	2.23	2.18	2.23	1.93	2.03	2.08	2.05
3B	1.98	1.98	2.08	1.98	2.06	1.98	2.03	1.98
4A	2.15	1.98	2.10	2.03	1.98	1.95	2.05	2.03
4B	1.73	1.80	1.95	1.85	1.83	1.80	1.83	1.78

All data in nA.  $T_A = 25^{\circ}\text{C}$

TABLE 15. 54H74 SEQUENCE OF SAME PARAMETER DATA FOR  $V_{IC}$ 

Pin	Sequence Number							
	1	2	3	4	5	6	7	8
D1	-0.869	-0.870	-0.867	-0.868	-0.869	-0.870	-0.867	-0.868
Clock 1	-0.866	-0.866	-0.867	-0.865	-0.866	-0.864	-0.863	-0.867
Clear 1	-0.844	-0.842	-0.843	-0.844	-0.845	-0.844	-0.846	-0.843
Preset 1	-0.848	-0.845	-0.848	-0.847	-0.848	-0.846	-0.847	-0.848
D2	-0.861	-0.859	-0.861	-0.861	-0.860	-0.862	-0.859	-0.861
Clock 2	-0.861	-0.860	-0.862	-0.862	-0.859	-0.861	-0.860	-0.860
Clear 2	-0.840	-0.843	-0.842	-0.842	-0.840	-0.840	-0.842	-0.841
Preset 2	-0.845	-0.846	-0.843	-0.845	-0.844	-0.845	-0.845	-0.846

All data in volts.  $T_A = 25^\circ\text{C}$

## 11. EFFECTS OF PIN APPLICATION SEQUENCE

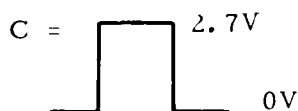
### TEST DESCRIPTION AND PROCEDURE

All tests in Subgroups 1, 7, and 9 were run with a comprehensive set of pin condition application sequences within the constraints of the specification. Since it was considered mandatory to connect GND and  $V_{CC}$  at the beginning of each test run to avoid damaging CMOS devices, we followed this practice rather than including them in the pin application sequence. Taking  $V_{OH}$  test numbers 1 through 4 as an example from MIL-M-38510/30102, we have the following pin conditions for tests 1 through 4.

TABLE 16.  $V_{OH}$  TEST PIN CONDITIONS (D-TYPE FLIP-FLOP)

Pin	CLR	D	CLK	PR	Q	$\bar{Q}$
Test 1	0.7 V	2.0 V	GND	2.0 V		-0.4 mA
Test 2	2.0 V	2.0 V	GND	0.7 V	-0.4 mA	
Test 3	2.0 V	2.0 V	C <u>1</u> /	2.0 V	-0.4 mA	
Test 4	2.0 V	0.7 V	C	2.0 V		-0.4 mA

1/



The conditions whose sequence of application were varied were 0.7 V, 2.0 V, GND or C, and -0.4 mA.

### RESULTS AND CONCLUSIONS

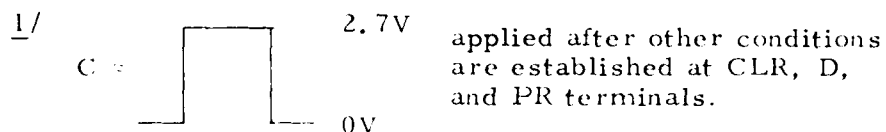
Whenever valid data was obtained, it showed no dependence on pin application sequence. However, a random pin application sequence, especially with clocked data, does not necessarily lead to valid data. In tests 3 and 4, if the clock pulse is applied before input conditions are established on D, the outputs may assume the complement state, resulting in improper test conditions for a  $V_{OH}$  measurement. In addition, in devices where outputs

are fed back to gates that drive the complement output (such as this flip-flop), the application of the load to the output after input conditions are stabilized, or after the connection of a voltmeter (a capacitive short), can frequently cause the output to change state, resulting in invalid data.

Since this procedure left open the question of damage or inconsistent data which was thought to occur by including  $V_{DD}$  and ground in the pin application sequences, an additional study was required. The CD4011A and CD4013A parts were tested by establishing a variety of 0V and 15V input conditions and then connecting ground or  $V_{DD}$  and later running the initial program to check for damage. The parts were again tested by establishing these input conditions with  $V_{DD}$  and ground, both first at 0V and then at 15V, before being switched to their appropriate values. No damage and no parameter changes were noted. However, no testing was performed to evaluate the effects of these voltage conditions on the lifetime of the devices. The absence of damage to the CMOS gates and flip-flops is attributed to the gate structure as well as to advances in input protection circuitry.

The following suggestions are made:

1. Pulsed inputs could be defined such that pulses occur after establishing dc conditions. This can be done in the MIL-M-38510 specifications in the notes that follow Table III. For example in Table 16, note 1 can be expanded to



2. Output states should be checked and re-established if necessary after forcing currents are applied and measurement systems are connected. This can be done in a suggested new paragraph in MIL-STD-883 to cover conditions unique to ATE as explained in Section 16, Suggestions for MIL-STD-883.

## 12. EFFECTS OF UNDESIGNATED PIN AMBIGUITY

### TEST DESCRIPTION AND PROCEDURE

Undesignated pins are frequently allowed to be either open, logic high, or logic low, at the discretion of the test engineer. This is the case with /05703, /00205, /00903, /02203, /07001, /07101, /30001, and /30101. The remaining five device specifications require undesignated pins to be open. In each measurement where an undesignated pin was allowed to be high, low, or open, the measurement was made five times, with the undesignated pins set to  $V_{CC}$ , logic high (2.0V), logic low (0.7 or 0.8V), ground, and open. Although the device with open input pins typically performs as expected, connecting input pins to either a high or a low state input voltage greatly decreases the susceptibility to noise spikes. This is particularly true of low-power Schottky and CMOS devices where small quantities of charge can cause changes of state.

The state of undesignated output pins can also affect the data. In counters and shift registers, the states of unmeasured output terminals are often unspecified. Loading unmeasured outputs during  $I_{OS}$  tests affects the current available to the measured output as well as the junction temperature of the DUT. In the special case of shorting more than one output at a time, as is allowed in Table III of the specifications, the absolute maximum power dissipation rating of the device is exceeded.

### RESULTS AND CONCLUSIONS

Varying the undesignated input pin conditions did not produce significant variations in measurements of DC parameters. When varying the conditions of the input pins such that undesignated outputs were switched, some propagation delay time measurements were affected.

Varying the conditions of undesignated output pins was found to affect the data in two situations. In the first situation, propagation delay times for 54164 and 54LS295 shift registers varied with the number of outputs making a state transition during the measurement cycle. The worst case  $T_{PHL1}$  for the 54164 occurred when all of the outputs started in a high state and were cleared together. This case yielded a  $T_{PHL1}$  with a mean value of 37.5 nanoseconds. When only one output was in a high state before being

cleared,  $T_{PHL1}$  was found to be 35.1 nanoseconds, 6 percent smaller than the first case. For the 54LS295, a similar but smaller change was noted. The respective mean values were 26.4 ns and 25.4 ns. A 1 nanosecond change was the limit used as a significance guideline. These differences could be avoided by specifying all output state changes that are to occur during the measurement.

The second situation is the  $I_{OS}$  test. Table 17 illustrates how  $I_{OS}$  varies with the loading of the other outputs. For the 54LS295, the measurement with all outputs grounded is 39 percent smaller than the measurement with only the measured output grounded.

Since loading more than one output at a time results in a power dissipation beyond the absolute maximum rating of the device, this situation can be rectified by specifying that only one output be grounded at a time. A proposed change in MIL-STD-883B, Method 3011 to cover this situation is discussed under the topic, "Additional Conditions for  $I_{OS}$  Measurements," in Section 15.

TABLE 17. EFFECT ON  $I_{OS}$  OF GROUNDING MORE THAN ONE OUTPUT

Device Type	I <sub>OS</sub> (mA)				Maximum Difference (%)
	Number of Additional Grounded Outputs				
	0	1	2	3	
54LS295A	-42.8	-42.5	-32.7	-26.0	39
5400	-46.7	-46.2	-45.5	-44.9	4
54H00	65.5	64.9	62.6	57.0	13
54S00	69.1	67.4	66.3	55.7	19
54LS00	26.1	26.0	25.9	25.8	1
T <sub>A</sub> = 25°C					

### 13. INVESTIGATION OF MEASUREMENT EFFECTIVENESS

Investigations were undertaken to examine the effectiveness of tri-state timing measurements, to find alternatives to the currently specified  $f_{MAX}$  tests, and to look for a method of examining  $C_{in}$  on CMOS parts on a general purpose automated test system such as the S-3260.

#### TRI-STATE TIME MEASUREMENT INVESTIGATION

Enable and disable times of tri-state outputs have been a bane to engineers using automated test equipment. Minimum capacitance on the output of an S-3260 with a comparator connected is slightly under 30 pF. When a load is connected, the capacitance increases to between 45 and 50 pF. Many commercial specifications require probe and jig capacitance of as little as 5 pF. MIL-M-38510/30606, Rev B requires 15 pF minimum for  $t_{HZ}$  and  $t_{LZ}$  measurements. Thus, it is very difficult to make the specified  $t_{HZ}$  and  $t_{LZ}$  measurements. Also, in the military specification, these same two measurements require comparator trip points of  $V_{OH}$  minus 0.5V and  $V_{OL}$  plus 0.5V. On a bench with an oscilloscope, the dynamic  $V_{OH}$  and  $V_{OL}$  levels are easily seen, but on ATE there is some complication.

Following are three methods that have been used on ATE for measuring the specified  $t_{HZ}$  and  $t_{LZ}$ :

1. Measuring DC values. This gives a false impression of output voltages that occur while operating at rated speed, especially if care is not taken to approximate the actual load.
2. Measuring dynamic levels. Output comparator reference levels are adjusted until the device fails a functional test. With this method, the measurement still depends upon where in the waveform the output is compared. In Figure 19, three different  $V_{OH}$  values are obtained from the same waveform. The value obtained depends upon the start time ( $t_B$ ) and stop time ( $t_{B'}$ ) gating the comparator.
3. Assuming  $V_{OH}$  and  $V_{OL}$  as worst case full load values permitted by the specification (that is, 2.4V and 0.4V). This is a more stringent requirement than either other method, but it is so much easier to implement that it has found some acceptance.

While all three methods give some information about the output transistors of the DUT, the circuit conditions during test have little relation to those



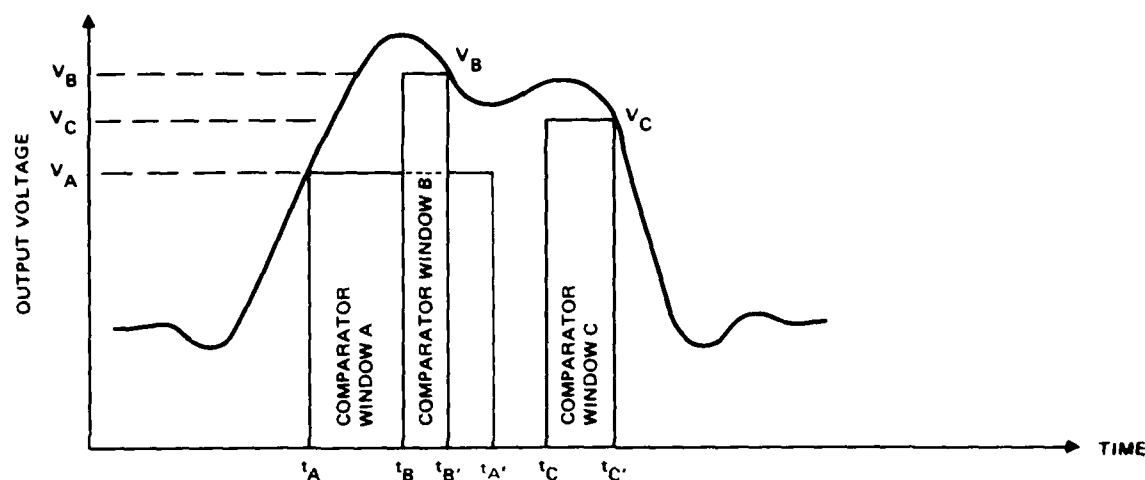
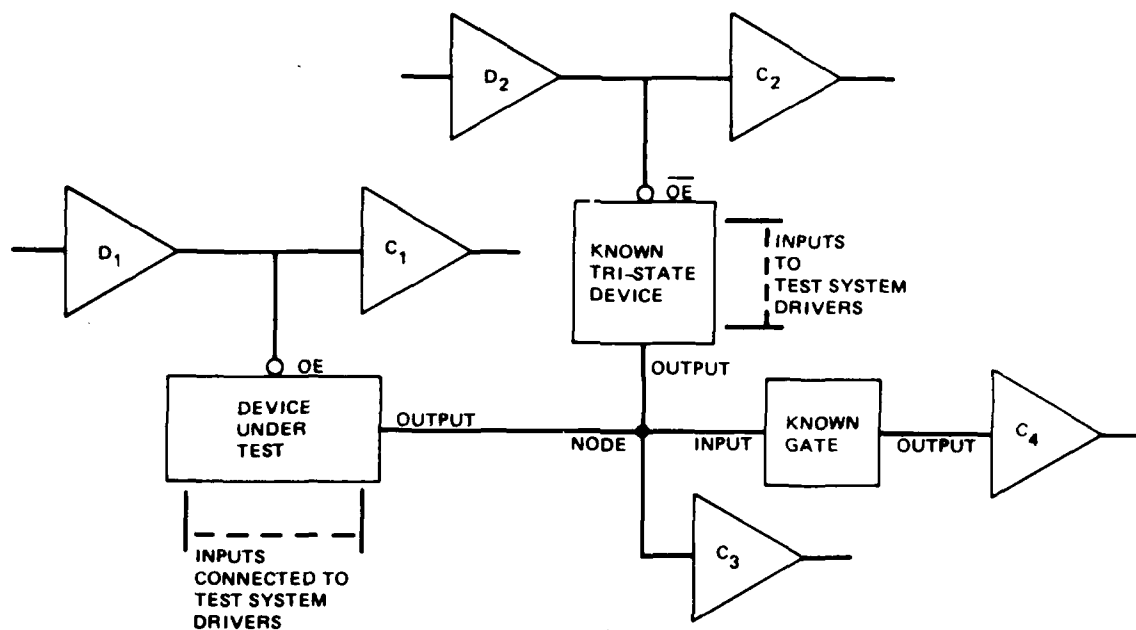


Figure 19. Result of dynamic volt measurement dependent on timing characteristics of comparator window.

encountered in actual use. The designer needs an idea of the capability of the device to release a bus ( $t_{HZ}$  and  $t_{LZ}$ ) or to be enabled and drive another device ( $t_{ZH}$  and  $t_{ZL}$ ).

One solution to the designer's problem, which removes some of the constraints imposed by ATE, is to connect the tri-state outputs to a bus structure. The node of this artificially created bus is connected to the DUT output, the output of a previously characterized tri-state device, and the input of a gate (Figure 20). The characterized tri-state device is capable of driving the gate input, as is the DUT. Bus access time ( $t_{ZH}$  and  $t_{ZL}$ ) are then measured as the time between an output enable pulse at the DUT and a change of output state of the known gate minus the previously measured delay of the gate. The gate is preset by the known tri-state device. Since the timing of this device has also been characterized, it can be made to access or release the bus node at any desired time. Bus release times ( $t_{HZ}$  and  $t_{LZ}$ ) are measured by attempting to access the bus with the known tri-state device after the test equipment attempts to disable the DUT at its output enable (OE) terminal. A bus conflict is created when two or more devices attempt to force different logic levels on the bus. An increase in the delay time from the known device enable signal to the gate output indicates that the DUT has not released the bus. The bus release time



$D_1 = D_2 =$  TEST SYSTEM DRIVERS OR PULSE GENERATOR

$C_1 = C_2 = C_3 = C_4 =$  TEST SYSTEM COMPARATORS

Figure 20. Theoretical alternative to tri-state measurements.

would be the difference between the time at which the DUT OE terminal was switched to disable the DUT output and the earliest time that the known device could access the bus without a conflict. This test method presents some problems to the test engineer who implements it. Bus elements (the known tri-state device and gate) need to be available as a load at each tri-state output of the DUT. The measurement requires the subtraction of an external gate delay. It also requires that several additional signal sources and comparators be connected. Some of these connections are to points other than DUT terminals.

Some of the problems might be eliminated by letting a driver (pulse generator) of the ATE become the known device trying to access the bus, and replacing the known gate with a passive network as is done in propagation delay measurements. Such a situation is presented in Figure 21. Now

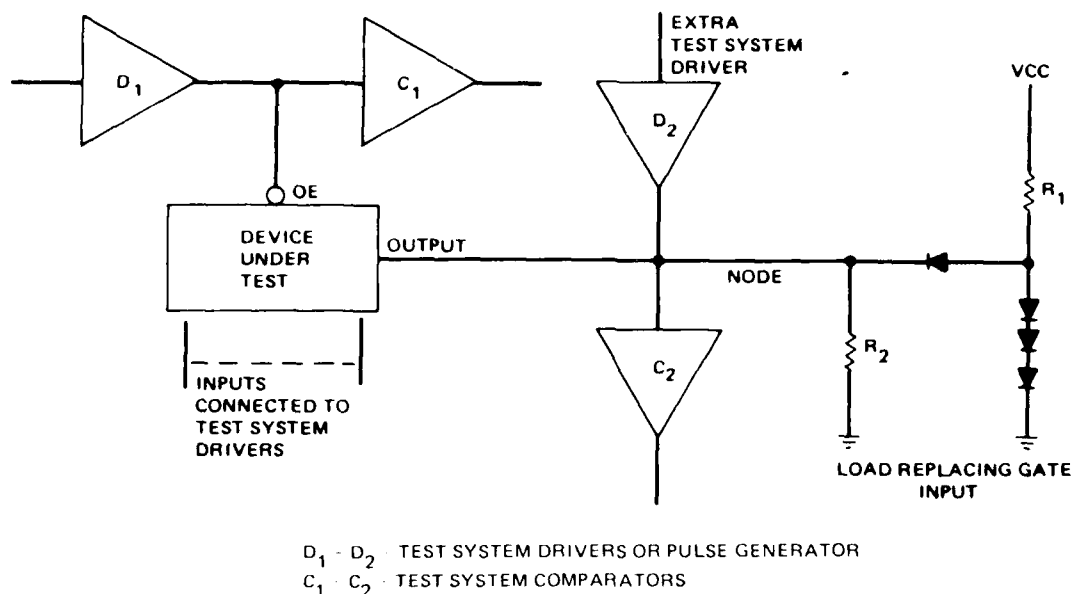
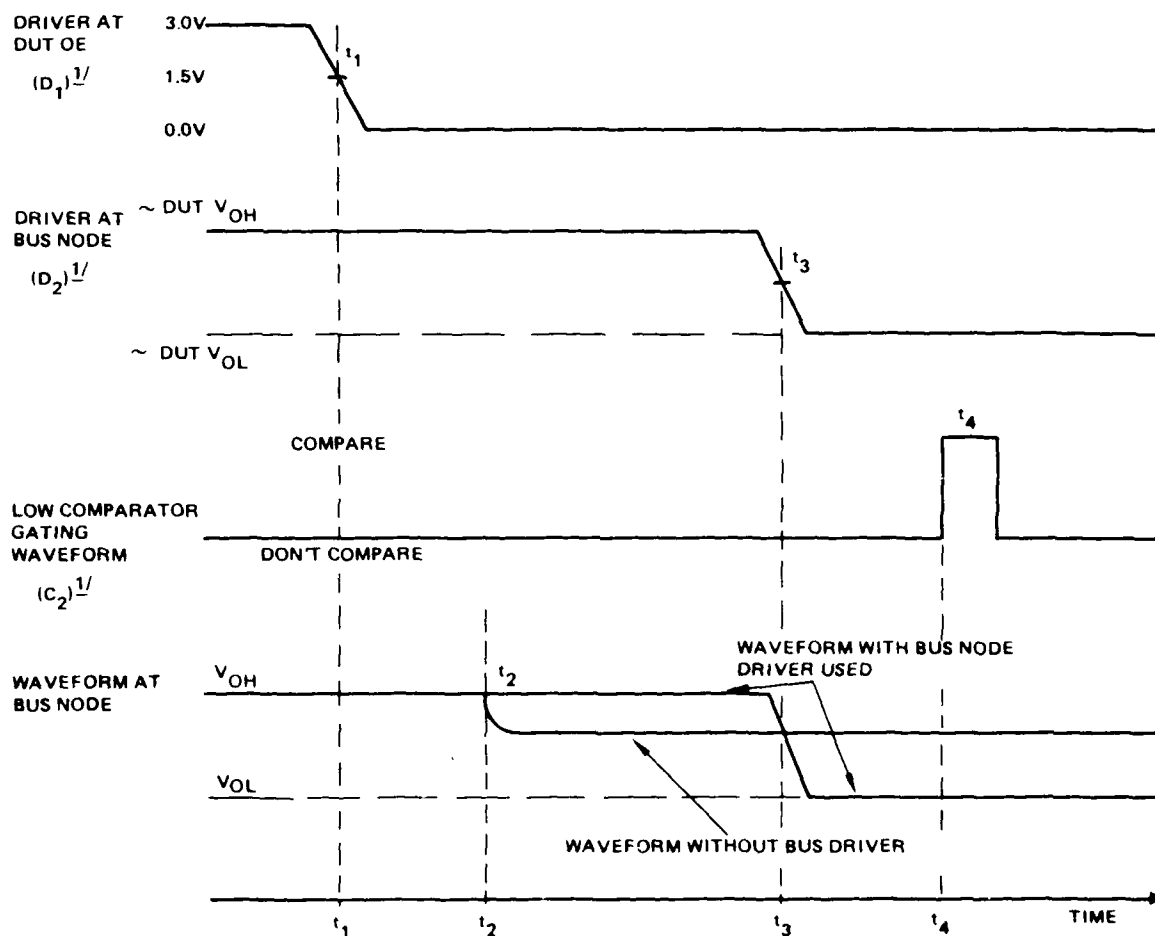


Figure 21. ATE model of Figure 20.

system timing can be adjusted to determine bus access and release times for the DUT, and no precharacterized devices are required. Since only the newest ATE permits its drivers to act as tri-state devices, implementation of this scheme on most ATE, including the S-3260, requires great care to ensure that the system driver on the bus node and the DUT output create minimum conflict on the bus. The approach taken was to activate the driver when it could clearly be done without conflict (after a disable, before an enable), and then walk the driver pulse start or end time toward a point of conflict with the DUT output. When a comparator indicates that both the driver and the DUT output are attempting to force the bus node in opposite directions, the measurement value has been determined. Examples of this method are given for  $t_{HZ}$  and  $t_{ZH}$ . The test method for  $t_{LZ}$  would follow that for  $t_{HZ}$  except for the levels of some waveforms. The method for  $t_{ZL}$  would similarly follow that for  $t_{ZH}$ .

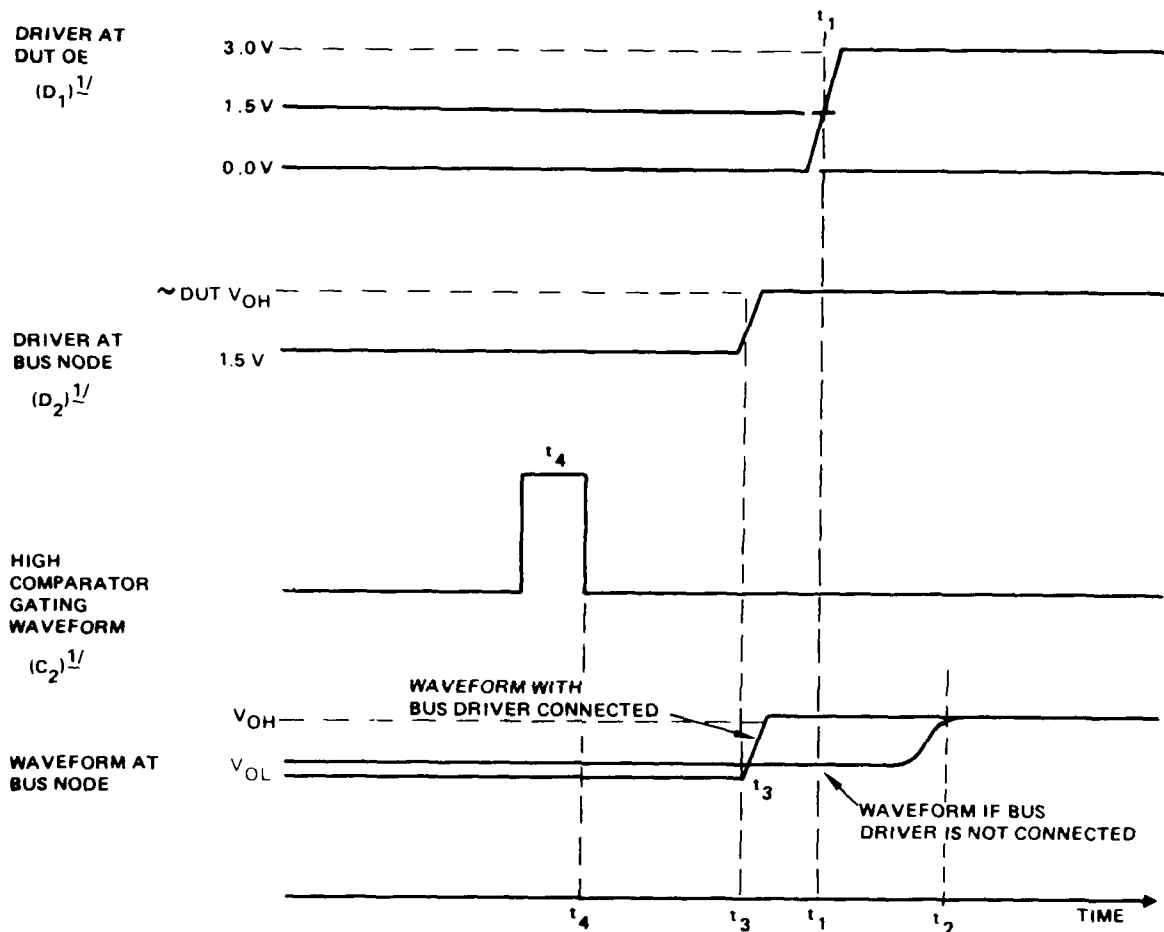
Figure 22 illustrates the initial timing of the test setup for a  $t_{\text{HZ}}$  measurement. The driver at the bus node has its logic 1 set at approximately the DUT  $V_{\text{OH}}$  level and its logic 0 set at approximately the DUT  $V_{\text{OL}}$  level. It is desirable to approach actual DUT levels, but exact matching is not required since small differences in the programmed value will affect driver loading but will not damage either the driver or the DUT. The switching of the driver ( $t_3$ ) is set to occur well beyond the suspected output disable time. The comparator logic 0 reference (locompare) level is set as close to the bus driver low level as possible without inadvertently triggering the comparator. A programmed value of 50 mV greater than the driver low level was found to be satisfactory. The comparator window time ( $t_4$ ) is initially well beyond  $t_3$  so that the comparator is not initially triggered by the falling edge of the bus driver signal. The test program reduces the value of  $t_4$  (moving  $t_4$  closer to  $t_3$ ) until a fail indication is received, signaling that the comparator is detecting the falling edge of the bus driver waveform. At this point  $t_4$  is incremented by 1 nanosecond so that the comparator will again indicate a pass condition. This timing relationship between the falling edge of bus driver waveform ( $t_3$ ) and the comparator start time ( $t_4$ ) is maintained throughout the test. Both  $t_3$  and  $t_4$  are reduced simultaneously 1 nanosecond at a time (moving closer to  $t_2$ ) until a comparator fail indication is received. This fail indication results because the device output has not fully disabled and is still applying a forcing voltage to the bus. The measured  $t_{\text{HZ}}$  would then be  $t_3 - t_1 + 1$  nanosecond. The 1 nanosecond is added because the DUT had not released the bus at  $t_3$ , but has successfully released the bus during the last pass at  $t_3 + 1$  nanosecond.

Figure 23 illustrates the initial waveforms for a bus access time ( $t_{\text{ZH}}$ ) measurement with the bus driver set at a low level of 1.5V and a high level approximating the DUT  $V_{\text{OH}}$  level. The switching time  $t_3$  is fixed and is used as a reference during this measurement. The high comparator trigger level is set 50 mV above 1.5V. Initially  $t_4$  is earlier in the test cycle (smaller) than  $t_3$ . However,  $t_4$  is incremented until a pass is recorded, and then decremented 1 nanosecond. The driver at the DUT OE terminal is initially set



NOTE:  $1/2$  NOTATION AS IN FIGURE 21.

Figure 22. Initial waveforms for suggested  $t_{HZ}$  measurement.



NOTE:  $\frac{1}{2}$  NOTATION AS IN FIGURE 21.

Figure 23. Initial waveforms for suggested  $t_{ZH}$  measurement.

to switch at a time  $t_1$  sufficiently late to ensure that the DUT will not attempt to access the bus ( $t_2$ ) until after  $t_3$ . At this time  $t_1$  and thus  $t_2$  are decremented until a pass signal is generated at the comparator, and the bus access time measurement ( $t_{ZH}$ ) is measured as  $t_3 - t_1$  nanosecond.

In practice, this method yielded results comparable to the measurements specified in the MIL-M-38510 specifications. The data in Table 18 shows

TABLE 18. COMPARISON OF SUGGESTED TRI-STATE  
METHOD TO SPECIFICATION METHOD

54LS295 Parameter	Temperature (°C)	Specification Method (ns)	Suggested Method (ns)
$t_{ZL}$	25	23.78	20.11
	-55	27.88	24.42
	125	25.70	22.08
$t_{ZH}$	25	13.51	10.47
	-55	13.93	10.81
	125	15.25	12.03
$t_{LZ}^{1/}$	25	24.75	22.37
	-55	25.22	22.91
	125	29.20	27.15
$t_{HZ}^{1/}$	25	18.49	16.03
	-55	19.16	16.57
	125	19.66	16.98

<sup>1/</sup> Specification requires a minimum load capacitance of 15 pF. A 50 pF load capacitance was used for the specification method data.

that enable and disable time readings were 2 to 3 nanoseconds faster using this method (the DUT does not need to charge or discharge an external 50 pF load, but only a load sufficient to affect the edge of the system driver waveform).

The ability of a system driver to simulate the limited drive capability of a tri-state output has properly been questioned. An S-3260 driver can source -100 mA in the high state (-170 mA when shorted to ground) and can sink +100 mA in the low state. This  $I_{OL}$  value is almost double that of a Schottky bus driver such as a 54S240, and the  $I_{OH}$  value is much greater than the -12 mA sourced by the 54S240. When compared to the values of a 54LS295, the  $I_{OL}$  and  $I_{OH}$  values of a system driver are an order of

magnitude greater. This greater current capability affects the measurements in two ways. First, the higher drive capability makes system capacitance less important since the capacitance is more rapidly charged and discharged. Second, the higher drive capability makes it more difficult for the DUT to affect the waveforms and trigger the comparators. Indeed, some difficulty was encountered in choosing appropriate comparator levels to obtain sensitivity to waveform changes without allowing the comparator to be falsely triggered by noise. The addition of a passive network to buffer the driver would enable the driver to more closely emulate the drive capability of the DUT output and thus reduce the difficulty in choosing comparator levels. On an S-3260, the addition could be accomplished by changing load modules. This approach should be examined further.

The suggested method differs from the current specification in that the intent is to describe the performance of the DUT in a bus environment rather than to measure the characteristics of the output drivers of the DUT. It does require programming effort to implement, but is less sensitive to system capacitance. Since system capacitance limitations make the current  $t_{HZ}$  and  $t_{LZ}$  measurements very difficult to implement, this is a definite advantage.

#### MAXIMUM FREQUENCY TEST INVESTIGATION

As discussed in Section 3, many slash sheets require functional tests with clock rates exceeding the maximum test frequency of the ATE (20 to 25 MHz, depending on the test system). Although test system capabilities are growing with each new generation of ATE, component speeds are increasing even faster. With the recent introduction of a 100 MHz tester came the introduction of advanced Schottky parts with an  $f_{MAX}$  of over 200 MHz. Thus, the test engineer is faced with a dilemma. If he does not test  $f_{MAX}$ , he may be passing parts that cannot meet end use requirements since the part which passes one-shot propagation delays or a 1 MHz truth table test may not pass with all timing parameters simultaneously exercised



to their limits. Since the test engineer can rarely afford to bench test 100 percent of the devices, his alternatives are the following:

1. Forget that a problem exists and drop the  $f_{MAX}$  test completely
2. Implement the test, keeping all timing relationships as tight as required by the specification except that clock rate is reduced to the test system capability
3. Implement the test as specified, at great expense, through special fixturing or bench methods
4. Bench test some fraction of a lot for  $f_{MAX}$
5. Rather than measuring  $f_{MAX}$  directly, measure some additional timing parameters and calculate an equivalent  $f_{MAX}$ .

The second and fifth methods might be acceptable both to the manufacturer and the reliability engineer if some evidence existed to support the conclusion that results produced by these methods would correlate with those produced by the  $f_{MAX}$  test implemented to the specification. The second method only produces a pass-fail indication at the programmed clock rate. This cannot be correlated to the pass-fail indication of the specification  $f_{MAX}$  test without testing parts which marginally fail  $f_{MAX}$ . No such parts were among those procured for the study. In support of the second method, it is better than no  $f_{MAX}$  test at all and should detect many  $f_{MAX}$  failures. It is the method most often implemented in an ATE environment.

The fifth method produces a value for  $f_{MAX}$ , and this value can be compared to measurements of  $f_{MAX}$  taken on the bench. For a D-type flip-flop, such as the 54S74, data setup and hold times are measured and  $f_{MAX}$  is calculated as the inverse of either of the following:

logic 1 setup time + logic 1 hold time +  $t_{PD LH}$  from CLK to  $\bar{O}$   
or

logic 0 setup time + logic 0 hold time +  $t_{PD HL}$  from CLK to  $\bar{O}$ ,  
whichever produces a lower  $f_{MAX}$  value. This fifth method was implemented in the /07101 and /02203 programs used throughout the study. These devices were then bench tested for  $f_{MAX}$ . The S-3260 calculated values were within 4 percent of actual  $f_{MAX}$  at 25°C (Table 19). The methods did not produce similar results at temperature extremes, although neither method indicated any device failures among the tested devices. One explanation for the differences at temperature extremes is that the calculated  $f_{MAX}$  did not use a measured minimum data pulse width along with the

TABLE 19. COMPARISON OF S-3260 F<sub>MAX</sub>  
TEST TO SPECIFICATION TEST

Device Type	Temperature (°C)	Specification Test (Method 3) (MHz)	S-3260 Test (Method 5) (MHz)
07101	25	91.5	95.3
	-55	83.4	97.9
	125	85.6	118.1
02203	25	45.4	46.9
	-55	36.2	47.0
	125	38.4	60.8

set-up and hold times. The simplified formula used was not adequate. However the data obtained by the alternate method, when it includes a measured data pulse width, is sufficiently useful that it has been incorporated in a number of programs when bench testing is deemed impractical.

#### MEASUREMENT OF INPUT CAPACITANCE

Alternatives to the direct measurement of input capacitance on CMOS devices were sought. Since test system capacitance was so much larger than the capacitances to be measured, no suitable alternative was discovered. As mentioned previously (Section 3, Table 4),  $C_{in}$  measurements were not part of the baseline programs used in the bulk of the study for this reason.

#### 14. DEFINE "OPEN" AND "GROUND"

##### TEST DESCRIPTION AND PROCEDURE

This task was divided into two subtasks. Subtask 1 characterized various ground and signal paths of the S-3260, using a Hewlett Packard 4271B 1 MHz LCR meter to determine what "open" and "ground" conditions are seen by a DUT. Resistance, inductance, and capacitance of each path were measured using the LCR meter. Since some of the resistances to be measured exceeded the highest measurable value on this meter, a conductance meter with a resolution of 10 picosiemens was used for these resistances. Measurements of less than one unit on this meter are interpreted as resistances greater than  $10^{11}$  ohms.

Subtask 2 required adding resistance, capacitance, and inductance to the system at DUT inputs and at the DUT ground terminal in order to seek a range of acceptable values that could be used to define "open" and "ground" conditions.

##### RESULTS AND CONCLUSIONS

The results of the first subtask appear in Table 20. Figure 24 is included to aid in locating some of the measurement points. The capacitance and parallel resistance values limit the quality of the open condition. The inductance and, to a smaller extent, series resistance values affect the ground condition. The smallest supply path inductance for a DUT after decoupling is 50 nH; that is, 12 nH from the decoupling capacitor to the DUT  $V_{CC}$  terminal, plus 38 nH from the other lead of the capacitor to the DUT ground terminal. This value assumes no inductance in the ceramic decoupling capacitor.

The second subtask was further divided. Additional resistance to ground and capacitance were added to DUT inputs in an attempt to find a worst case open condition, and additional inductance was placed in ground paths to study the ground condition. For the open condition, the additional capacitance (200 pF maximum) at DUT inputs served only to slow the rise and fall times of the system drivers and had no effect on conditions requiring some inputs to be open (such as  $V_{IC}$ ). The initial test program, as

TABLE 20. INDUCTANCE OF GROUND PATHS AND CAPACITANCE OF THE S-3260

Measured Capacitance		Value (pF)	R	Condition <sup>1/</sup>
From	To			
DT	SG	29	>10 <sup>5</sup> megohms	O
DT	SG	31	>10 <sup>5</sup> megohms	I&O
DT	SG	44	>10 <sup>5</sup> megohms	I&O, C5
DT	SG	31	>10 <sup>5</sup> megohms	I&O, C30
DT	SG	35	>10 <sup>5</sup> megohms	I&O, L1
DT	SG	36	>10 <sup>5</sup> megohms	I&O, L2
DT	SG	40	>10 <sup>5</sup> megohms	I&O, L1, L2
DT	SG	49	>10 <sup>5</sup> megohms	I&O, L1, C5
DT	SG	36	>10 <sup>5</sup> megohms	I&O, L1, C30
Measured Inductance		(nH)	Milliohms	
DG	SG <sup>2/</sup>	81	13	Hardwired
DG	LG	126	20	L1
SG	LG	52	9	L1
DT	driver	146	24	I&O
DT	SG	335	54	O
DT	SG	222	35	I&O
DT	LG	263	43	L1
DT	LG	285	49	L2
DT	SG	206	33	L1, L2
DT	SG	161	26	L1, L2, I&O
DG	ground ring at point nearest V <sub>CC</sub>	38	6	Hardwired
DT	bottom of socket	12	3	

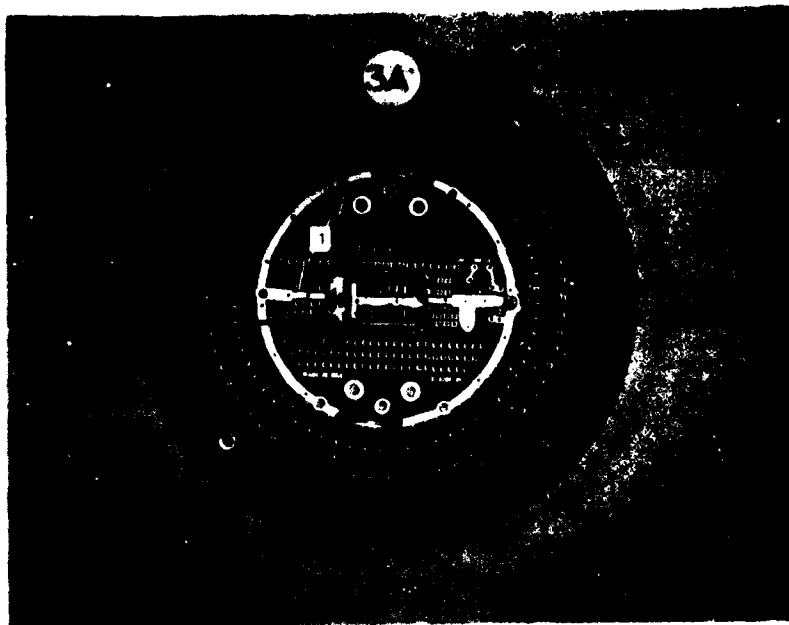
<sup>1/</sup> The following abbreviations are used:

SG - Ground ring on socket card  
 LG - Ground ring on load card  
 DG - Ground terminal of DUT  
 DT - Any DUT terminal  
 SGG - Sector card ground

L1 - Load Relay No. 1 closed  
 L2 - Load Relay No. 2 closed  
 C5 - Comparator on the 5V range  
 C30 - Comparator on the 30V range  
 O - Sector connected as an output  
 I - Sector connected as an input

<sup>2/</sup> Ground paths are not numbered unless so labeled

a) TOP VIEW



1. SOCKET CARD GROUND RING
2. DECOUPLING CAPACITOR
3. HARDWIRED GROUND CONNECTIONS

b) BOTTOM VIEW

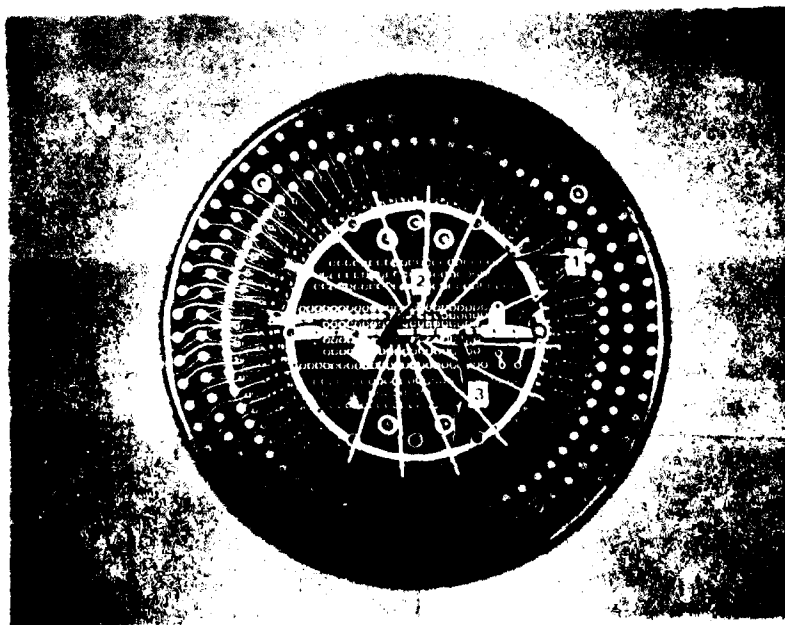


Figure 24. S3260 socket card with a 14-pin zero-insertion-force dual-in-line socket attached (approximately 1/2 scale)

defined in Section 3, was used to determine whether all DUTs would pass when 1 megohm was connected between all DUT terminals and ground. All of the TTL parts passed, and  $V_{IC}$  and  $I_{IH}$  values did not change. CMOS devices failed leakage current tests because of current flow through the added resistance, but timing parameters did not change. Since measurements of input leakage currents of less than 1 nanoamp must be made on CMOS devices with a forcing voltage of 15 volts, an open must be defined such that the leakage current through the open is at least 50 times less than the 1 nanoamp limit. This requires a resistance of more than 15 gigohms during measurement. An acceptable open condition for a CMOS input could be defined as "open  $\leq 50$  pf (a measurement system limitation) and  $\geq 10^{10}$  ohm." For the bipolar and CMOS devices, a definition for the output condition as "open  $\leq 50$  pf and  $\geq 1$  megohm" would be adequate.

Ground path inductance and resistance were increased by removing the capacitor from its position between  $V_{CC}$  and the DUT ground terminals, and placing it between the  $V_{CC}$  terminal and the ground ring of the socket card. The hardwired ground connection was then removed and the ground terminal was connected both to the I- and the O-connection of the socket card, and then grounded through reeds in an S-3260 driver. This path represents 234 nanohenries of supply path inductance. Timing measurements were affected as shown in Table 21. The data indicate that any additional inductance can affect measurements noticeably.

Specifying a maximum acceptable inductance in a ground path is very difficult. Since the current in an inductor cannot change instantaneously, the switching current transients require time to dissipate. Higher inductance results in longer time delays. ATE using device handlers to speed through-put may be very hard pressed to meet a 50 nanohenry or 100 nanohenry requirement, yet the addition of less than 200 nanohenries produces considerable differences in timing parameters. A compromise definition of an adequate ground path could be "ground  $\leq 200$  nanohenries and  $\leq 0.1$  ohm." Although these suggested values would not lead to a guarantee of data reproducible to 3 percent accuracy, as was shown in Table 21, they do recognize the limitations of ATE and could be met with careful fixturing

on a variety of test systems. Further, any device meeting test requirements on a 200 nanohenry system would meet the requirements of a system furnishing a lower inductance ground path.

TABLE 21. PROPAGATION DELAY TIMES FOR  
TWO GROUND PATH INDUCTANCES

54LS295 Parameter	At 50 nH (ns)	At 234 nH (ns)	Percent Difference
$t_{PHL}$ , data mode	29.4	31.4	6.5
$t_{PHL}$ , serial mode	28.3	29.0	2.5
$t_{PLH}$ , data mode	25.1	25.0	-0.2
$t_{PLH}$ , serial mode	24.3	24.3	0.0
$t_{ZL}$	22.6	24.4	8.0
$t_{LZ}$	37.6	33.8	-10.0

## 15. SLASH SHEET REVIEW

During the course of the investigations previously described, the MIL-M-38510 slash sheets for each of the 13 device types were studied in detail to identify typographical errors, errors of omission, implied measurement conditions, redundant or superfluous measurements, and any errors or ambiguities in the notes and figures. Table 22 lists the revision and amendment of each slash sheet used for this study. Table 23 lists each of the items identified and their location in the specifications.

TABLE 22. PROCUREMENT SPECIFICATIONS USED DURING THIS STUDY

Generic Part Number	MIL-M-38510 Slash Number	Device Type	Revision	Amendment Number	Updated Through
5400	001	04	B	3	28 Jun 1976
5474	002	05	E	4	25 Jul 1978
54164	009	03	C	5	7 Jul 1978
54H74	022	03	B	1	1 Oct 1976
54H00	023	04	A	4	1 Jun 1977
4011A	050	01	C	-	16 Nov 1977
4013A	051	01	B	2	10 Sep 1976
4015A	057	03	B	2	7 Nov 1978
54S00	070	01	-	4	5 Apr 1978
54S74	071	01	A	3	19 Apr 1979
54LS00	300	01	A	2	14 Nov 1977
54LS74	301	01	A	1	18 Apr 1978
54LS295	306	06	B	1	12 Jun 1979



TABLE 23. IDENTIFICATION AND LOCATION OF SPECIFICATION ERRORS

Discussion Item	Listing of Occurrences	Comments
Additional conditions for $I_{OS}$ measurements	<p>/001, pp. 15, 17, 19, 21, 23</p> <p>/002, pp. 38, 41, 45, 48, 52, 55, 59</p> <p>/009, pp. 62, 65, 69, 74, 79, 83</p> <p>/022, pp. 47, 50, 54, 58, 62, 64</p> <p>/023, pp. 14, 17, 19, 21</p> <p>/071, pp. 39, 47, 51, 55, 58, 59</p> <p>/300, pp. 35, 40, 44, 50, 55</p> <p>/301, pp. 76, 80, 85, 88, 92, 96, 100, 105</p> <p>/306, pp. 60, 65, 69, 73, 76, 81, 85</p> <p>Also MIL-STD-883B Test Method 3011</p>	
Minimum or Maximum Recommended Pulse Widths	/009, p. 3	
Output Loading in Figures for Measurement of Time	<p>/009, pp. 43, 46, 49, 55, 58</p> <p>/057, pp. 31, 32, 35, 37</p> <p>/306, pp. 27, 29, 32, 35, 38, 41, 44, 48</p>	<p>/009 p. 49 shows remaining outputs shorted</p> <p>/057 - The method used in these drawings is better than that used in the drawings for /009 or /306</p>
Comparator levels for truth table tests	<p>/002, pp. 39, 43, 46, 50, 54, 57, 61</p> <p>/009, pp. 63, 67, 72, 77, 81, 84</p> <p>/022, pp. 48, 52, 56, 60, 63, 67</p> <p>/071, pp. 42, 46, 50, 54, 57, 61</p> <p>/301, pp. 78, 82, 86, 90, 94, 98, 102, 107</p> <p>/306, pp. 62, 67, 71, 75, 79, 89</p>	

Table 23. Identification and Location of Specification Errors - continued

Discussion Item	Listing of Occurrences	Comments
50 Percent Duty Cycle	/001, p. 12 /002, pp. 21, 22, 23 /022, pp. 28, 29, 31, 32 /023, p. 11	
$F_{MAX}$ , $F_{CL}$ Limits Figures, conditions	/002, pp. 31, 33 /022, pp. 38, 39 /051, pp. 21, 23, 24, 27, 29	"observing proper output state changes"
A. Figures and notes	/057, pp. 31, 32, 34, 36, 38, 40 /071, p. 28	
B. Table III and notes	/002*, pp. 39, 42, 43, 46, 49, 50, 52, 53, 54, 57, 60, 61 /022*, pp. 5, 47, 48, 50, 51, 54, 58, 59, 62, 63, 65, 67 /051, pp. 34, 40 /057, pp. 44, 49, 54, 59, 64, 72 /071, pp. 40, 41, 42, 45, 46, 48, 50, 52, 54, 56, 57, 59, 61 /301, pp. 77, 78, 81, 82, 85, 86, 89, 90, 93, 94, 97, 98, 101, 102, 106, 107 /306, pp. 61, 62, 66, 67, 70, 71, 74, 75, 78, 79, 82, 83, 86, 88, 89	*Was correct in the table until amended to be incorrect
Flip-flop propagation delays	/002 /022 /071 /301	Occurs in Table III of each specification in subgroup 9, 10, and 11 testing

Table 23. Identification and Location of Specification Errors - continued

Discussion Item	Listing of Occurrences	Comments
Redundant or Omitted Measurements		
A. $I_{SS}$	/050, pp. 8	Only this case was examined. Other simi- lar cases probably exist in this and most other CMOS specifications
B. Truth table testing	All subgroup 7 or 8 testing in the specifications	
C. Redundant $f_{max}$ testing	/022, pp. 47, 48, 50, 51, 54, 55, 58, 59, 62, 63, 65, 66	
D. Insufficient $f_{max}$ testing	306, pp. 61, 66, 70, 74 78, 81, 86	
Miscellaneous Errors and Omissions		Detailed in this report, Section 15, "TYPO- GRAPHICAL ERRORS AND OMISSIONS"

## ADDITIONAL CONDITIONS FOR $I_{OS}$ MEASUREMENTS

Most manufacturers' data sheets specify that  $I_{OS}$  shall be measured with only one output grounded at a time and that no output be grounded for more than 1 second for most devices, or for more than 100 ms for some device types. Without such limitations, the device may dissipate much more power than it was designed to handle, and junction temperatures in excess of  $200^{\circ}\text{C}$  can quickly be reached. This problem has been recognized in the specification under paragraph 1.2.4. "Absolute maximum ratings," where, as Figure 25 shows, note 1 indicates that the device "Must withstand the added  $P_D$  due to short circuit conditions (e. g.,  $I_{OS}$ ) at one output for 5 seconds." As can be seen in Figure 26, note 4 to Table I of MIL-M-38510/002 adds, "Not more than one output should be shorted at a time."

However, devices are tested to the conditions of Table III, which has no notes limiting the test duration or requiring that the other outputs not be grounded (Figure 27). This specification may be technically correct since the absolute maximum ratings are not to be exceeded during test, and grounding more than one output would cause a power dissipation beyond the absolute maximum rating. However, it is not obvious from Table III that the specification prohibits this practice.

Since these limitations concern the method of measuring  $I_{OS}$ , MIL-STD-883B, Method 3011 should be modified to correct the situation. This can be done by an addition to Method 3011.1, paragraph 3, which currently reads

The device shall be stabilized at the specified test temperature.  
Each output per package shall be tested individually.

The suggested addition is

Output terminals not under test shall be open. Output terminals shall not be forced to the test condition voltage potential for a period longer than 5 seconds.

MIL-M-38510 2E  
 24 December 1974  
 SUPERSEDING  
 MIL-M-0038510 2D(USAF)  
 15 October 1973 and  
 MIL-M-38510 2C  
 7 November 1972

# MILITARY SPECIFICATION

## MICROCIRCUITS, DIGITAL, TTL, FLIP FLOPS, MONOLITHIC SILICON

This specification is approved for use by all Departments and Agencies of the Department of Defense

### 1. SCOPE

**1.1 Scope.** This specification covers the detail requirements for monolithic silicon, TTL, bistable logic microcircuits. Three product assurance classes and a choice of case outline lead finish are provided for each type and are reflected in the complete part number.

**1.2 Part number.** The complete part number shall be as shown in the following example:

M38510	002	01	B	A	C
Military designator	Detail specification	Device type (1.2.1)	Device class (1.2.2)	Case outline (1.2.3)	Lead finish (3.3)

**1.2.1 Device type.** Device type shall be as shown in the following:

Device type	Circuit
01	Single J-K master-slave flip-flop
02	Dual J-K master-slave flip-flop, no preset
03	Dual J-K master-slave flip-flop, no preset
04	Dual J-K master-slave flip-flop
05	Dual D-type edge-triggered flip-flop
06	Single edge-triggered J-K flip-flop
07	Dual D-type edge-triggered flip-flop, buffered output

**1.2.2 Device class.** Device class shall be the product assurance level as defined in MIL-M-38510.

**1.2.3 Case outline.** The case outline shall be designated as follows:

Letter	Case outline, MIL-M-38510, appendix C
A	F-1 (14-pin, 1 4" x 1 4", flat pack)
B	F-3 (14-pin, 1 8" x 1 4", flat pack)
C	D-1 (14-pin, 1 4" x 3 4", dual-in-line)
D	F-2 (14-pin, 1 4" x 3 8", flat pack)
E	D-2 (16-pin, 1 4" x 7 8", dual-in-line)
F	F-5 (16-pin, 1 4" x 3 8", flat pack)

**1.2.4 Absolute maximum ratings.**

Supply voltage range	-0.5 to 7.0 Vdc
Input voltage range	-1.5 Vdc at -12 mA to 5.5 Vdc
Storage temperature range	-65° C to 150° C
Maximum power dissipation per flip-flop, P <sub>D</sub>	110 mW <span style="border: 1px solid black; padding: 0 2px;">1</span>
Lead temperature (soldering, 10 seconds)	300° C
Thermal resistance, junction to case	θ <sub>JC</sub> { 0.09° C/mW for flat packs 0.08° C/mW for dual-in-line pack
Junction temperature	T <sub>J</sub> 175° C

1. Must withstand the added P<sub>D</sub> due to short circuit condition (e.g., I<sub>OS</sub>) at one output for 5 seconds duration.

Figure 25. Page 1 of MIL-M-38510/2E showing that the absolute maximum ratings of the device types require withstanding an I<sub>OS</sub> condition at only one pin for 5 seconds.

TABLE I. Electrical performance characteristics.

Test	Symbol	Conditions	Device type	Limits		Units
				Min	Max	
High-level output voltage	$V_{OH}$	$V_{CC} = 4.5 \text{ V}$ $I_{OH} = 400 \mu\text{A}$	All	2.4	---	Volts
Low-level output voltage	$V_{OL}$	$V_{CC} = 4.5 \text{ V}$ $I_{OL} = 16 \text{ mA}$	All	---	0.4	Volts
Input clamp voltage	$V_{IC}$	$V_{CC} = 4.5 \text{ V}$ $I_{A} = 12 \text{ mA}$ $T_A = 25^\circ\text{C}$	All	---	-1.5	Volts
Low-level input current	$I_{I1}$	$V_{CC} = 5.5 \text{ V}$ $V_{IN} = 0.4 \text{ V}$	01, 02, 03, 04, 05, 06	-0.7	-1.6	mA
			07	-0.5	-1.6	mA
Low-level input current	$I_{I2}$	$V_{CC} = 5.5 \text{ V}$ $V_{IN} = 0.4 \text{ V}$	01, 02, 03, 04, 05	-1.4	-3.2	mA
			07	-1.0	-3.2	mA
Low-level input current	$I_{I3}$	$V_{CC} = 5.5 \text{ V}$ $V_{IN} = 0.4 \text{ V}$	01, 02, 03, 04	-0.7	-3.2	mA
High-level input current	$I_{IH1}$	$V_{CC} = 5.5 \text{ V}$ $V_{IN} = 2.4 \text{ V}$	All	0	40	$\mu\text{A}$
High-level input current	$I_{IH2}$	$V_{CC} = 5.5 \text{ V}$ $V_{IN} = 5.5 \text{ V}$	All	0	100	$\mu\text{A}$
High-level input current	$I_{IH3}$	$V_{CC} = 5.5 \text{ V}$ $V_{IN} = 2.4 \text{ V}$	All	0	80	$\mu\text{A}$
High-level input current	$I_{IH4}$	$V_{CC} = 5.5 \text{ V}$ $V_{IN} = 5.5 \text{ V}$	All	0	200	$\mu\text{A}$
High-level input current	$I_{IH5}$	$V_{CC} = 5.5 \text{ V}$ $V_{IN} = 2.4 \text{ V}$	01, 02, 03, 04, 05, 07	-850	-50	$\mu\text{A}$
High-level input current	$I_{IH6}$	$V_{CC} = 5.5 \text{ V}$ $V_{IN} = 5.5 \text{ V}$	05, 07	0	300	$\mu\text{A}$
Short-circuit output current	$I_{OS}$	$V_{CC} = 5.5 \text{ V}$ $V_{IN} = 0$	All	-20	-57	mA
Supply current per device	$I_{CC}$	$V_{CC} = 5.5 \text{ V}$ $V_{IN} = 5 \text{ V}$	01	---	20	mA
			02, 03, 04	---	40	mA
			05, 06, 07	---	30	mA
Maximum clock frequency	$f_{MAX}$		01, 02, 03, 04, 05, 07	5	---	MHz
			06	7.5	---	MHz
Propagation delay to high logic level (clear or preset to output)	$t_{PLH}$		01, 02, 03, 04, 05	5	39	ns
			06	5	62	ns
			07	5	31	ns
Propagation delay to low logic level (clear or preset to output)	$t_{PHL}$	$V_{CC} = 5 \text{ V}$ $C_L = 50 \text{ pF}$ minimum $R_L = 390 \Omega \pm 5\%$	01, 02, 03, 04, 05	5	50	ns
			06	5	62	ns
			07	5	39	ns
Propagation delay to high logic level (clock to output)	$t_{PLH}$		01, 02, 03, 04, 05	5	39	ns
			06	5	62	ns
			07	5	31	ns
Propagation delay to low logic level (clock to output)	$t_{PHL}$		01, 02, 03, 04, 05	5	50	ns
			06	5	62	ns
			07	5	39	ns

1. Input condition: J or K for device types 01, 02, 03, 04, 06, and preset or D for device types 05 and 07, and clock, clear or preset for device type 06.
2. Input condition: Clock for device types 01, 02, 03 and 04, and clear or clock for device types 05 and 07.
3. Input condition: Clear or preset for device types 01, 02, 03, 04, 05, 06 and 07 and clock for device types 05 and 07.
4. No more than one output should be shorted at a time.
5. Input condition: J or K for device types 01, 02, 03, 04, 06, and D for device types 05 and 07, and clock for device type 06.
6. Input condition: Clear or preset for device types 01, 02, 03 and 04.
7. Input condition: Clock for device types 01, 02, 03, and 04.
8. Input condition: Clear or preset for device types 05 and 07.
9. See Table III for supply current limits.

Figure 26. MIL-M-38510/2E, page 3, Table I, Note 4 requires that no more than one output be shorted at one time.

See notes at end of device type 05

72

#### MINIMUM OR MAXIMUM PULSE WIDTH IN RECOMMENDED OPERATING CONDITIONS IN MIL-M-38510/9C.

Page 3 of MIL-M-38510/9C presents some confusing recommended operating conditions for device types 02 and 03 (Figure 28). The specification recommends operating the 02 device with a minimum clock pulse width of 35 nanoseconds maximum. Does this imply that operation with a clock pulse width less than 35 nanoseconds is recommended? No. What was intended was that no device should fail to perform with a clock pulse width of 35 nanoseconds, which is a worst case or maximum acceptable minimum clock pulse width. This can be stated more clearly simply by declaring that the recommended width of the clock pulse be 35ns minimum, as was done for device type 05 on the same page of the specification. An alternative approach is to delete the words maximum and minimum following the recommended limit value as was done in MIL-M-38510/306B, p. 2 (Figure 29).

#### OUTPUT LOADING IN FIGURES FOR MEASUREMENTS OF TIME

When several outputs require similarly constructed loads, as on p. 55 of MIL-M-38510/9C (Figure 30), the notation "same output load as output A" is used to avoid drawing a number of loads unnecessarily. However, the wording chosen leaves open the possibility of connecting the same load to all the outputs (shorting the outputs together) or connecting the same single load to each of the outputs in turn. This ambiguity can be avoided by changing the wording to "output load configured as load of output A."

#### COMPARATOR LEVELS FOR TRUTH TABLE TESTS

The specifications allow either of two methods to be used to determine whether proper output states are observed during truth table tests (subgroups 7 and 8). As can be seen in note 4 of Figure 31, a high can be



## Device type 02

Minimum clock pulse width- - - - -	35 ns	maximum
Minimum clear pulse width- - - - -	30 ns	maximum
Minimum preset pulse width- - - - -	30 ns	maximum
Serial input setup time- - - - -	30 ns	minimum
Serial input hold time - - - - -	0 ns	minimum

## Device type 03

Minimum clock pulse width- - - - -	30 ns	maximum
Minimum clear pulse width- - - - -	50 ns	maximum
Serial setup time- - - - -	15 ns	minimum
Serial hold time - - - - -	10 ns	maximum

## Device type 04

Width of clock input pulse - - - - -	25 ns	minimum
Width of load input pulse- - - - -	15 ns	minimum
Clock enable setup time - - - - -	30 ns	minimum
Parallel input setup time- - - - -	10 ns	minimum
Serial input setup time- - - - -	20 ns	minimum
Shift setup time - - - - -	45 ns	minimum
Hold time at any input - - - - -	10 ns	maximum

## Device type 05

Width of clock input pulse - - - - -	20 ns	minimum
Width of clear input pulse - - - - -	20 ns	minimum
Data input setup time- - - - -	20 ns	minimum
Clear input setup time - - - - -	25 ns	minimum
Hold time at any input - - - - -	0 ns	minimum
Mode control setup time- - - - -	30 ns	minimum

## Device type 06

Width of clock input pulse - - - - -	16 ns	minimum
Width of clear input pulse - - - - -	12 ns	minimum
Shift load input setup time- - - - -	27 ns	minimum
Data input setup time- - - - -	20 ns	minimum
Clear input setup time - - - - -	25 ns	minimum
Shift load release time- - - - -	10 ns	maximum
Data hold time - - - - -	0 ns	minimum

## 2. APPLICABLE DOCUMENTS

2.1 The following documents, of the issue in effect on date of invitation for bids or request for proposal, form a part of this specification to the extent specified herein.

## SPECIFICATION

## MILITARY

MTL-M-38510 - Microcircuits, General Specification for.

Figure 28. MIL-M-38510/9C, page 3.

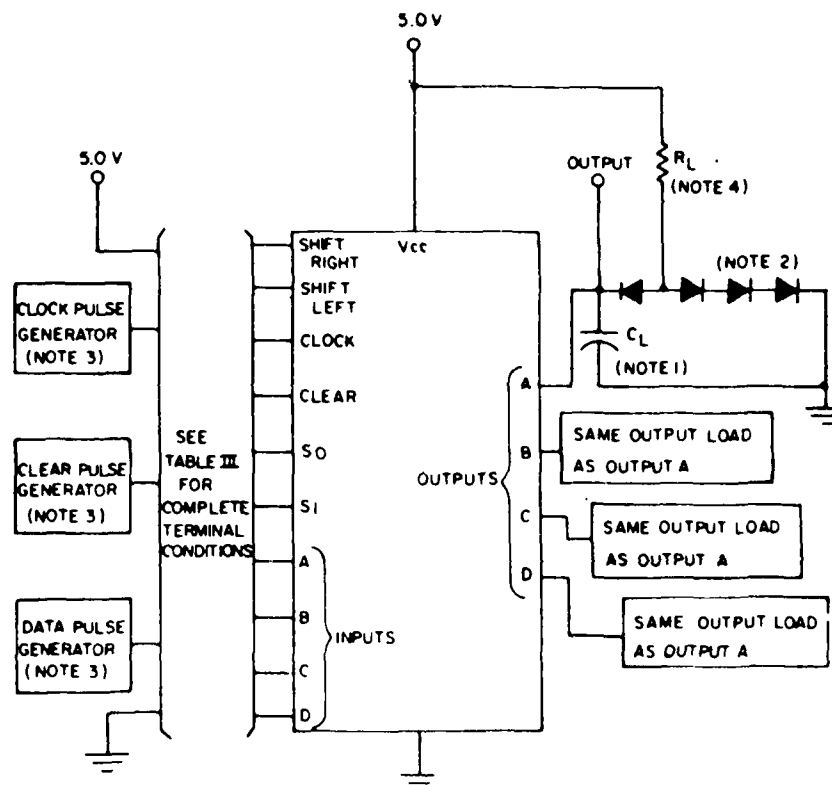
### 1.3 Absolute maximum ratings.

Supply voltage range - - - - -	-0.5 Vdc to 7.0 Vdc
Input voltage range- - - - -	-1.5 Vdc at -18 mAdc to 5.5 Vdc
Storage temperature range- - - - -	-65°C to 150°C
Maximum power dissipation per register, P <sub>D</sub> 1/	
Device type 01- - - - -	127 mWdc
Device type 02, 03 - - - - -	116 mWdc
Device type 04- - - - -	110 mWdc
Device type 05- - - - -	149 mWdc
Device type 06, 07 - - - - -	160 mWdc
Device type 08 - - - - -	198 mWdc
Device type 09 - - - - -	209 mWdc
Lead temperature (soldering, 10 seconds- - - - -	+300°C
Thermal resistance (junction to case)- - - - -	$\theta_{JC} = \begin{cases} 0.09^\circ\text{C/mW} & \text{for flat pack} \\ 0.08^\circ\text{C/mW} & \text{for dual-in-line pack} \end{cases}$
Junction temperature- - - - -	T <sub>J</sub> = +175°C

### 1.4 Recommended operating conditions.

Supply voltage- - - - -	4.5 Vdc minimum to 5.5 Vdc maximum
Minimum high level input voltage- - - - -	2.0 Vdc
Maximum low level input voltage- - - - -	0.7 Vdc
Ambient operating temperature range- - - - -	-55°C to 125°C
Minimum clock pulse width:	
Device type 01, 03, 05, 07, 09 - - - - -	20 ns
Device type 02- - - - -	18 ns
Device type 04, 06, 08 - - - - -	25 ns
Minimum clear pulse width:	
Device type 01, 05, 09 - - - - -	20 ns
Device type 02- - - - -	15 ns
Device type 04 - - - - -	30 ns
Device type 07 - - - - -	25 ns
Minimum load pulse width:	
Device type 08 - - - - -	15 ns
Minimum setup time at mode control:	
Device type 01 - - - - -	30 ns
Device type 03, 06 - - - - -	20 ns
Minimum setup time at shift/load:	
Device type 02 - - - - -	25 ns
Device type 07 - - - - -	20 ns
Device type 08 - - - - -	45 ns
Device type 09 - - - - -	30 ns
Minimum setup time at serial data:	
Device type 08 - - - - -	10 ns
Minimum setup time at serial or parallel data:	
Device type 01, 02, 03, 05, 06, 07, 09 - - - - -	20 ns
Device type 04 - - - - -	30 ns
Minimum setup time at preset:	
Device type 04 - - - - -	30 ns
Minimum setup time at inhibit:	
Device type 08 - - - - -	30 ns
Minimum hold time:	
Device type 01, 02, 03, 04, 05, 07 - - - - -	10 ns
Device type 06 - - - - -	20 ns
Device type 08, 09 - - - - -	0 ns
Minimum enable or inhibit time of clock:	
Device type 03 - - - - -	20 ns
Maximum release time of shift/load:	
Device type 02- - - - -	10 ns

1/ Must withstand the added P<sub>D</sub> due to short circuit test (e.g., I<sub>OS</sub>).



**NOTES:**

1.  $C_L = 50 \text{ pF}$  minimum including probe and jig capacitance.
2. All diodes are 1N3064, or equivalent.
3. Unless otherwise specified in the notes associated with the individual tests, all pulse generators have the following characteristics:  $Z_{OUT} = 50\Omega$ ,  $t_{TLH} \leq 7 \text{ ns}$ ,  $t_{THL} \leq 7 \text{ ns}$ ,  $V_{IH} = 3.0 \text{ V}$  minimum,  $V_{IL} = 0$ .
4.  $R_L = 400\Omega \pm 5\%$ .

Figure 30. From MIL-M-38510/9C, page 55.



measured as either 2.4V minimum or as 1.5V minimum and a low as 0.4V maximum or as 1.5V maximum. This allows two measurement standards, one less severe than the other, with the possibility of passing devices using one standard and failing them using the other. If the less severe measurement technique is acceptable, then there is no need for the more severe alternative. If the less severe technique is not acceptable, then it should be eliminated. Further discussion is withheld for the section on redundant measurements.

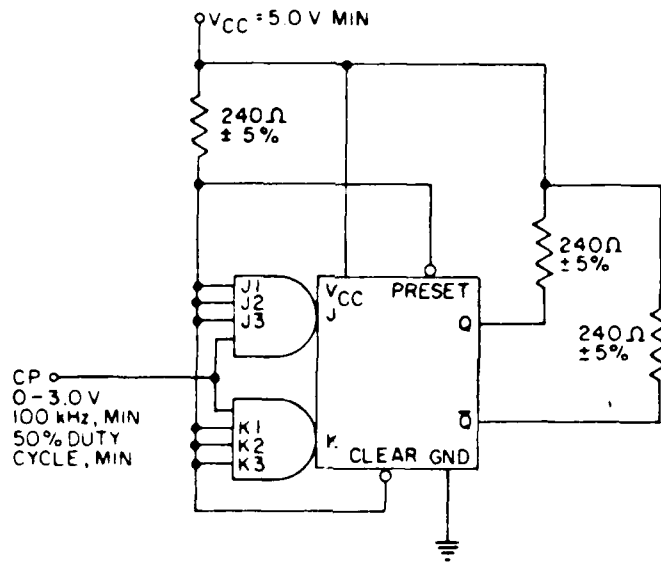
#### FIFTY PERCENT DUTY CYCLES

The burn-in circuits for MIL-M-38510/22B specify a clock pulse with a "50 percent DUTY CYCLE, MIN" as shown in Figure 32. This permits a clock pulse of 99 percent duty cycle or one of 50 percent. A 99-percent duty cycle does not exercise the device any better than a 1-percent duty cycle. The clock pulse generator would be better specified as having a "50 percent duty cycle  $\pm$  10 percent" or a "50 to 60 percent duty cycle." This was done in MIL-M-38510/301A on pages 57 to 60.

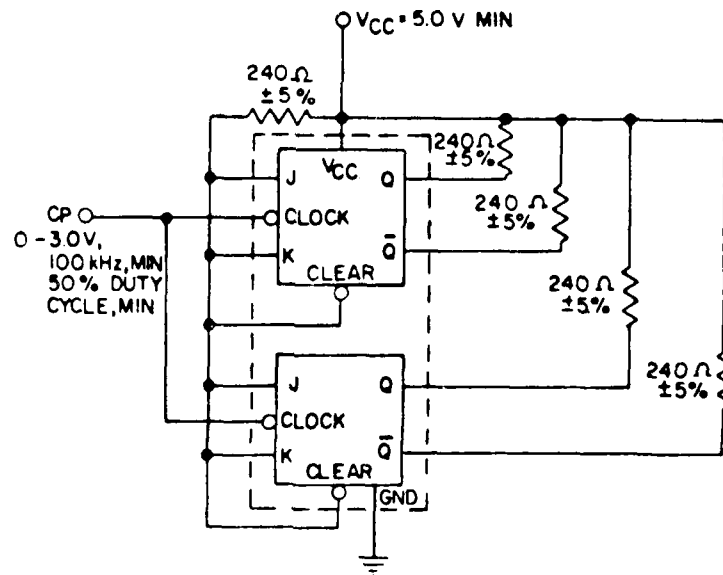
#### $f_{MAX}$ LIMITS AND TEST CONDITIONS

Several inconsistencies in MIL-M-38510 tests for  $f_{MAX}$  were noted in the /9, /22, and /71 specifications. Other inconsistencies are found in Table III.

The test circuit of page 38 of MIL-M-38510/22B is not correct for  $f_{MAX}$  since the  $\bar{Q}$  output is not connected to the D input as called out in Table III (Figure 33). Also, such a connection would load  $\bar{Q}$ , and the external load either should be adjusted to compensate or be eliminated. The voltage waveforms for the D input are not correct for  $f_{MAX}$  either, as  $t_1$ ,  $t_0$ ,  $t_{setup}$ , and  $t_{hold}$  are controlled by the characteristics of the  $\bar{Q}$  output and not by a programmable pulse generator. Each of these problems would be most easily addressed by using a separate figure for  $f_{MAX}$ , thus avoiding confusion between  $f_{MAX}$  conditions and conditions required for propagation delay time measurements. This was done in parts of MIL-M-38510/9C (e.g., page 47). Note that connecting the input of a



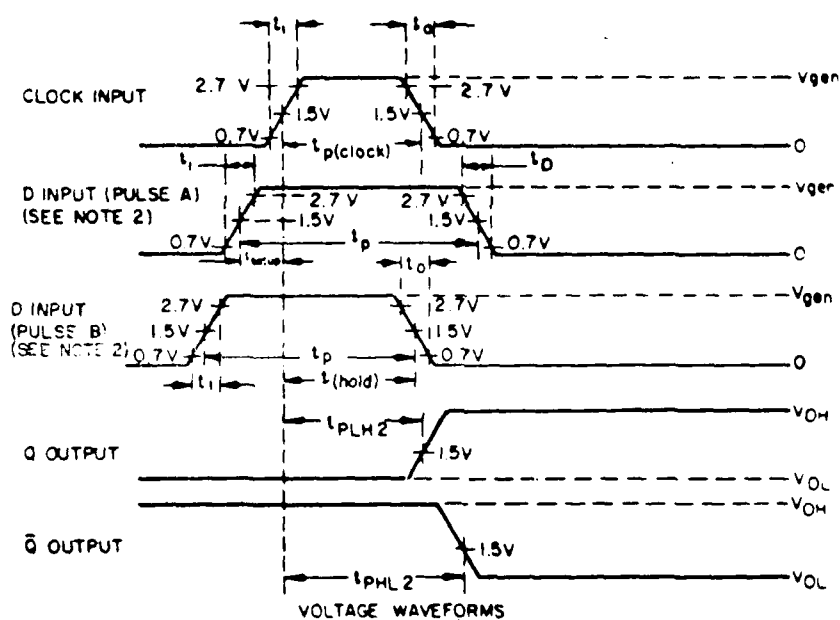
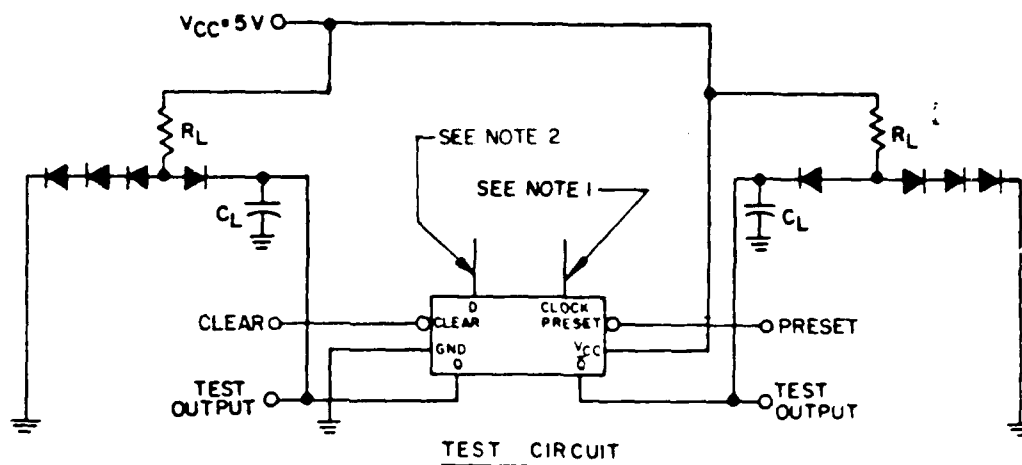
Device type 01



Device type 02

FIGURE 4 Burn-in and life-test circuits.

Figure 32. MIL-M-38510/22B, page 28, clock pulse specified as having "50 percent DUTY CYCLE, MIN."



**NOTES:**

1. Clock input pulse has the following characteristics:  $V_{gen} = 3\text{ V}$ ,  $t_0 = t_1 \leq 7\text{ ns}$ ,  $t_p(\text{clock}) = 20\text{ ns}$ , and  $\text{PRR} = 1\text{ MHz}$ . When testing  $f_{MAX}$ ,  $f = 36\text{ MHz}$  for subgroup 9 and  $f = 28\text{ MHz}$  for subgroups 10 and 11.
2. D input (pulse A) has the following characteristics:  $V_{gen} = 3\text{ V}$ ,  $t_0 = t_1 \leq 7\text{ ns}$ ,  $t_{setup} = 10\text{ ns}$ ,  $t_p = 60\text{ ns}$ , and  $\text{PRR}$  is 50% of the clock  $\text{PRR}$ . D input (pulse B) has the following characteristics:  $V_{gen} = 3\text{ V}$ ,  $t_0 = t_1 \leq 7\text{ ns}$ ,  $t_{hold} = 5\text{ ns}$ ,  $t_p = 60\text{ ns}$ , and  $\text{PRR}$  is 50% of the clock  $\text{PRR}$ .
3. All diodes are 1N3064, or equivalent.
4.  $C_L = 50\text{ pF} \pm 5\%$  (including jig and probe capacitance).
5.  $R_L = 280\Omega \pm 5\%$ .

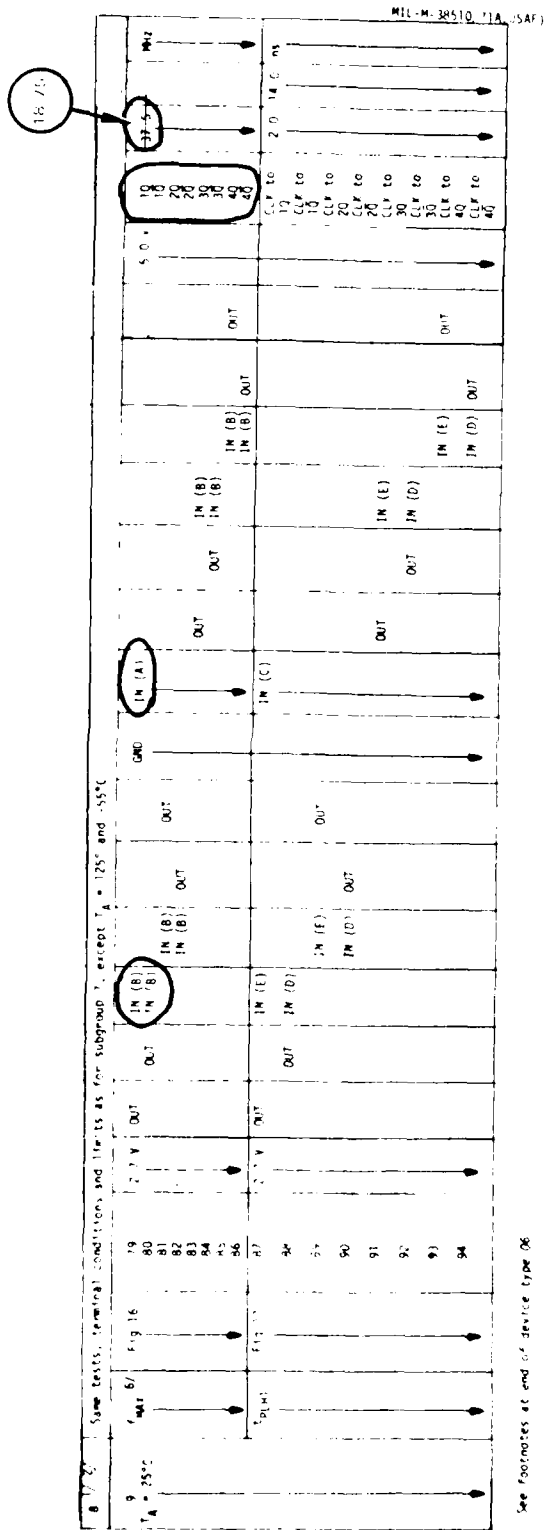
Figure 33. MIL-M-38510/22B, page 38,  $f_{MAX}$  test conditions conflict with other switching test conditions.

flip-flop to the  $\overline{Q}$  output is inconvenient on ATE, often requiring special fixturing and a connector length which adds inductance to the connection. This, in addition to the inability of ATE to run at clock rates in excess of 20 or 25 megahertz, is a reason to desire an alternative to the current  $f_{MAX}$  test method, as discussed in Section 13. The MIL-M-38510/301A method of testing  $f_{MAX}$  which uses a pulse generator on the D input instead of connecting  $\overline{Q}$  to D is much easier to implement on ATE and should be considered as a replacement for the method used in MIL-M-38510/9, /22, and /71.

The remaining inconsistencies are found in the various Table III entries pertaining to  $f_{MAX}$  for TTL or  $f_{CL}$  for CMOS devices. Referring to Figure 34, the  $f_{MAX}$  limit applies to the clock input frequency, yet the measured terminals are the outputs. Even though this is explained in note 6, it is inconsistent to measure at a Q terminal to a limit which does not apply at that terminal. Further, note 6 states that the output frequency shall be one-half of the input frequency. IN(A) and IN(B) use two different pulse repetition rates (PRR). Of which is the output one-half? In the flip-flops which require the  $\overline{Q}$  output be connected to the D input, is the output PRR required to be one-half of itself? Clearly this is not what is intended. These inconsistencies can be resolved by making the measurement limit the desired output PRR and specifying that the higher input PRR be applied in the value for  $f_{MAX}$  at the clock input.

Figure 35 is page 40 of 51B. Note "O" reads, "The maximum clock frequency ( $f_{CL}$ ) requirement is considered met if proper output state changes occur..." but "proper output state changes" are not defined in Table III. If an unusual wave shape results from the test, who determines whether the output state changes are proper? The phrase "...if the output state changes according to figure 2..." would be more precise because the truth tables of figure 2 define the proper output state changes, and note "N" defines the required output signal levels.





# NOTES

1. See Summary of attributes data sheet.
2. See pages 59 and 61.
3. See pages 59 and 61.
4. See pages 59 and 61.
5. See pages 59 and 61.
6. See pages 59 and 61.
7. See pages 59 and 61.
8. See pages 59 and 61.
9. See pages 59 and 61.
10. See pages 59 and 61.
11. See pages 59 and 61.
12. See pages 59 and 61.
13. See pages 59 and 61.
14. See pages 59 and 61.
15. See pages 59 and 61.
16. See pages 59 and 61.
17. See pages 59 and 61.
18. See pages 59 and 61.
19. See pages 59 and 61.
20. See pages 59 and 61.
21. See pages 59 and 61.
22. See pages 59 and 61.
23. See pages 59 and 61.
24. See pages 59 and 61.
25. See pages 59 and 61.
26. See pages 59 and 61.
27. See pages 59 and 61.
28. See pages 59 and 61.
29. See pages 59 and 61.
30. See pages 59 and 61.
31. See pages 59 and 61.
32. See pages 59 and 61.
33. See pages 59 and 61.
34. See pages 59 and 61.
35. See pages 59 and 61.
36. See pages 59 and 61.
37. See pages 59 and 61.
38. See pages 59 and 61.
39. See pages 59 and 61.
40. See pages 59 and 61.
41. See pages 59 and 61.
42. See pages 59 and 61.
43. See pages 59 and 61.
44. See pages 59 and 61.
45. See pages 59 and 61.
46. See pages 59 and 61.
47. See pages 59 and 61.
48. See pages 59 and 61.
49. See pages 59 and 61.
50. See pages 59 and 61.
51. See pages 59 and 61.
52. See pages 59 and 61.
53. See pages 59 and 61.
54. See pages 59 and 61.
55. See pages 59 and 61.
56. See pages 59 and 61.
57. See pages 59 and 61.
58. See pages 59 and 61.
59. See pages 59 and 61.
60. See pages 59 and 61.
61. See pages 59 and 61.
62. See pages 59 and 61.
63. See pages 59 and 61.
64. See pages 59 and 61.
65. See pages 59 and 61.
66. See pages 59 and 61.
67. See pages 59 and 61.
68. See pages 59 and 61.
69. See pages 59 and 61.
70. See pages 59 and 61.
71. See pages 59 and 61.
72. See pages 59 and 61.
73. See pages 59 and 61.
74. See pages 59 and 61.
75. See pages 59 and 61.
76. See pages 59 and 61.
77. See pages 59 and 61.
78. See pages 59 and 61.
79. See pages 59 and 61.
80. See pages 59 and 61.
81. See pages 59 and 61.
82. See pages 59 and 61.
83. See pages 59 and 61.
84. See pages 59 and 61.
85. See pages 59 and 61.
86. See pages 59 and 61.
87. See pages 59 and 61.
88. See pages 59 and 61.
89. See pages 59 and 61.
90. See pages 59 and 61.
91. See pages 59 and 61.
92. See pages 59 and 61.
93. See pages 59 and 61.
94. See pages 59 and 61.
95. See pages 59 and 61.
96. See pages 59 and 61.
97. See pages 59 and 61.
98. See pages 59 and 61.
99. See pages 59 and 61.
100. See pages 59 and 61.

Figure 34. Suggested revision for MIL-M-38510-71A, pages 59 and 61,  $f_{MAX}$  conditions and limits.

AD-A093 215

HUGHES AIRCRAFT CO CULVER CITY CA F/G 14/2  
EVALUATION OF ELECTRICAL TEST CONDITIONS IN MIL-W-38510 SLASH S--ETC(U)  
AUG 80 K SANDGREN F30602-78-C-0193  
UNCLASSIFIED HAC-FR-80-76-706 RADC-TR-80-263 NL

2 of 2  
34  
10/10



END  
DATE  
FILMED  
1-8  
DTIC



## FLIP-FLOP PROPAGATION DELAY

MIL-M-38510 requirements for flip-flop propagation delay times were reviewed in detail during this investigation. Figure 36 is a logic diagram of a D-type flip-flop. Observe that the Q output is an input to the NAND gate providing the  $\bar{Q}$  output and vice versa. This implies that Q cannot enter a low state until after  $\bar{Q}$  enters a high state and that  $\bar{Q}$  cannot enter a low state until Q enters a high state. Stated differently,  $T_{PHL}$  of one output is equal to  $T_{PLH}$  of the other output plus the delay of the final NAND gate. If a device passes  $T_{PHL}$  at Q it must have also passed  $T_{PLH}$  at  $\bar{Q}$ , since  $T_{PLH}$  at  $\bar{Q}$  was an integral part of the measurement. The  $T_{PLH}$  measurements appear to be redundant and could be deleted without reducing the thoroughness of the Table III testing. However, if those tests were deleted and a  $T_{PHL}$  failure occurred, the failing portion of the device (output gate or input gate) would not be readily identifiable. Thus the  $T_{PLH}$  test data is useful in tracking reliability problems. It is not suggested that the  $T_{PLH}$  tests be deleted.

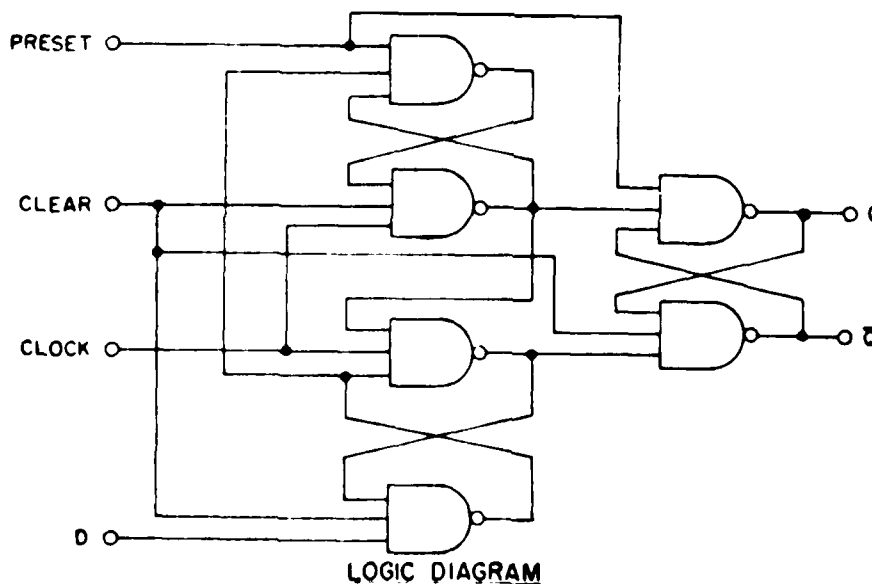


Figure 36. Logic diagram of a D-type flip-flop (from MIL-M-38510/22B, p. 17).

## REDUNDANT OR SUPERFLUOUS MEASUREMENTS AND OMITTED MEASUREMENTS

While reviewing the MIL-M-38510 slash sheets, a number of tests were encountered that do not provide additional information and are therefore redundant. Among these tests are an  $I_{SS}$  test, truth table testing, and some  $f_{MAX}$  tests. In addition, an omitted measurement for some  $f_{MAX}$  tests is discussed.

### FET Supply Current Tests

Test 18 of MIL-M-38510/50C, Table III for device type 01, shown in Figure 37 is redundant. As Figure 38 shows, the current  $I_{SS}$  consists of leakage from the input gates plus leakage from  $V_{DD}$  through the channels of the individual FETs. Tests 17 and 19 produce the two possible worst case conditions for leakage from  $V_{DD}$  by turning on either the P-channel or the N-channel FETs. Input leakage currents are measured separately as  $I_{IH}$  and  $I_{IL}$ . Considering the other  $I_{SS}$  and  $I_{IH}$  measurements, test 18 provides no new information and should be deleted.

### Truth Table Testing

As was discussed in Section 8, truth table testing can give information about dynamic  $V_{IH}$  and  $V_{IL}$  and also can assure that the devices change state according to the truth table. The current specifications do not require input conditions yielding this  $V_{IL_{max}}$  and  $V_{IH_{min}}$  information. Assurance of operation according to a truth table could be provided by the subgroup 9 tests ( $f_{MAX}$  and propagation delay time measurements) if the preconditioning and test sequence were specified. Therefore, although the truth table tests (subgroups 7 and 8) could be eliminated as redundant, they should be retained but run with tighter limits on input conditions to provide additional information.

### Testing for $f_{MAX}$

Four  $f_{MAX}$  tests are specified in Table III of MIL-M-38510/2E for device types 02, 03, 04, 05, and 07. A separate  $f_{MAX}$  test is specified for each output and each complemented output although the same set of input

[illegible]

86

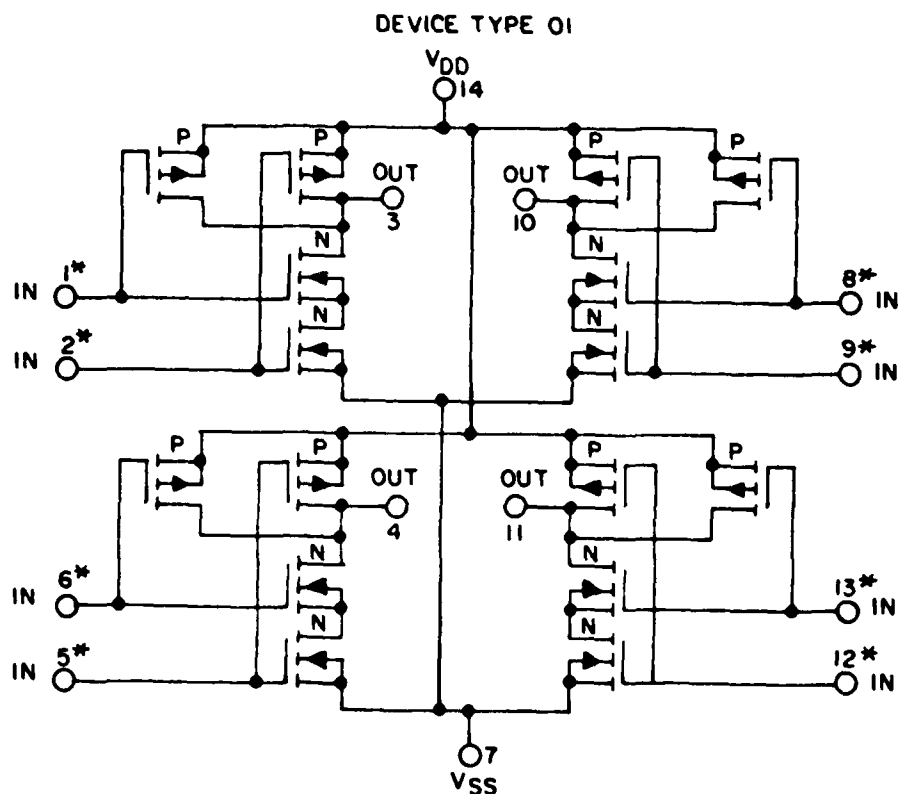


Figure 38. Schematic of CD4011A (MIL-M-38510/50C, p. 8).

conditions could be used for all four tests. In an age of multiple-trace oscilloscopes and ATE capable of testing all outputs simultaneously, one combined  $f_{MAX}$  test could replace the four currently specified. Even if this were not done, having demonstrated the interdependence of the  $Q$  and  $\bar{Q}$  outputs and utilizing the propagation delay,  $V_{OH}$ , and  $V_{OL}$  measurements to demonstrate the capability of the output to drive a load, it has been determined that testing  $f_{MAX}$  on  $Q$  output is an adequate test of the  $\bar{Q}$  output. Therefore, two of the four  $f_{MAX}$  tests may be deleted as redundant.

#### Omitted $f_{MAX}$ Measurement Condition

In some multi-stage shift registers, the  $f_{MAX}$  test is performed at only one output. For example, in MIL-M-38510/306B, page 82, only  $Q_A$  of the 54LS295 is tested. Examination of Figure 39 shows that even if

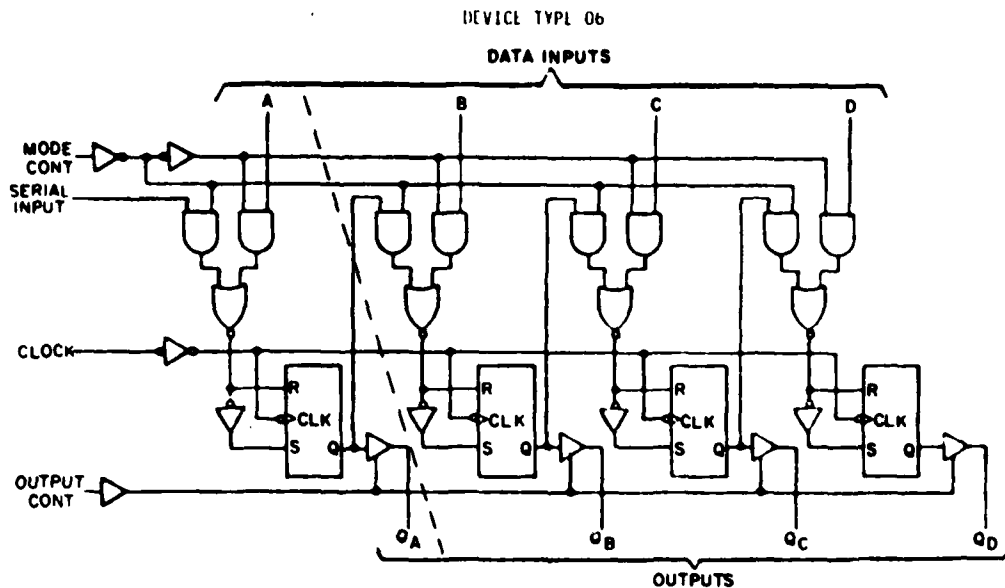


Figure 39. Logic diagram of 54LS295  
(MIL-M-38510/306B, page 24).

the longest single stage delay path is from the serial input to the output  $Q_A$ , a proper square wave output at  $Q_A$  does not guarantee a proper output at  $Q_B$ ,  $Q_C$ , or  $Q_D$ . A fault anywhere to the right of the dashed line would not be detected. As a minimum,  $Q_A$  and  $Q_D$  need to be tested. A fault in the output buffer gate at either  $Q_B$  or  $Q_C$  might still go undetected by the  $f_{MAX}$  test, since the following stage is connected before the buffer, but such faults would be detected in propagation delay tests.

Similarly, counters are sometimes tested for  $f_{MAX}$  at only the most rapidly varying stage, using the theory that no other stage has to switch at more than one-half the frequency of this stage. However, device design may take this into account, and the delay paths into subsequent stages may be sufficiently long that the second most rapidly varying stage may not meet the one-half  $f_{MAX}$  requirements. Also, signal levels within a device operated at  $f_{MAX}$  may not be sufficient to drive succeeding stages reliably. If outputs are buffered, the reduced signal level would not be apparent by any test other than  $f_{MAX}$  at both the most rapidly and least rapidly varying outputs.



## TYPOGRAPHICAL ERRORS AND OMISSIONS

In addition to those ambiguous or incorrect situations previously described, the following are minor errors that were found.

<u>Page No.</u>	<u>MIL-M-38510/2F</u>
3	Note 4 reads "Nor more..." It should read "Not more..."
16	The note "Schematic shows one-half of dual unit" should be a note for all three drawings, not just circuit B, and should be placed under the notes heading of page 17.
	<u>MIL-M-38510/9C</u>
13-14	The function tables have neither a figure number, nor a table number, nor are they text. They should be labelled as part of figure 2 "Truth tables and timing diagrams."
41	Note reads " $R_2 = R_6 = 220 \Omega \pm 10\%$ ." It should read " $R_2$ through $R_6 = 220 \Omega \pm 10\%$ " or " $R_2 = R_3 = R_4 = R_5 = R_6 = 220 \Omega \pm 10\%$ ."
49	Figure 8 shows several outputs shorted together before being connected to loads. Since there is not adequate space to draw the load eight times, $Q_B$ through $Q_H$ should be separately connected to a single rectangle in which is inscribed "Connect each output with an output load configured as the load at output A."
	<u>MIL-M-38510/22B</u>
2	In note 2, the output fanout needs further definition. A type 06 output may not drive 10 type 06 clear or clock inputs at 150 $\mu$ A each when high level current capability is only 0.5 mA. This can be corrected by restating note 2 as "Device will fanout in both high and low levels to the specified number of D, J, or K inputs of the same device type."
8	In the figure for terminal conditions of device type 02, the two K terminals, the two Q terminals, etc. need to be differentiated as K1 and K2, Q1 and Q2, etc.
31	In the figure, a current limiting resistor (240 ohm) would be appropriate between $V_{CC}$ and the inputs.
34, 36, 38, 41, 43, 45	The notes say, "When testing $f_{MAX}$ , $f = \dots$ " They should say, "When testing $f_{MAX}$ , $PRR = \dots$ "

<u>Page No.</u>	<u>MIL-M-38510/22B (continued)</u>
55	The arrow in the max column under "22" extends to Test 105. It should stop at Test 101.
67 and following pages	If page 67 is the last page of the document, where is paragraph 6 and information about the preparing activity?
	<u>MIL-M-38510/23A</u>
9-10	Less confusion would arise if all versions of these circuits were drawn with the same number of inputs.
	<u>MIL-M-38510/50C</u>
4	Paragraph 4.4.1(e) reads: "Device input pins not designated in tests may be tied to $V_{CC}$ or GND or may be open provided measurements are not affected. Device output pins not designated in tests may be loaded or open provided measurements are not affected." If measurements were to be affected, which condition would prevail? It is suggested that the following statement be added: "If measurements are found to be affected, input pins not designated in tests shall be tied to the voltage level that will yield a worst case measurement value, and output pins not designated shall be loaded."
7-8	Case dimensions are a part of the general specification MIL-M-38510D. Figure 1 is not required. The references to figure 1 should be changed to reference the appropriate figure in M38510D, Appendix C.
29	Paragraph 6.4. The symbols " $t_f$ " and " $t_r$ " are not used anywhere in this specification (" $t_{THL(1)}$ " and " $t_{TLH(1)}$ " are used), so reference to their use can be deleted.
	<u>MIL-M-38510/71A</u>
30	In note 1 the symbol "dc" is used for duty cycle. This symbol is more generally used for direct current. Duty cycle should be spelled out.
	<u>MIL-M-38510/306B</u>
58	The caption under the title reads "... or open)", it should read "... or open)"

Page No.      MIL-M-38510/306B (continued)

83              Subgroup 11 is missing.

98              Paragraph 4.5.1 Voltage and current, which reads, "All voltages given are referenced to the ground terminal. Currents given are conventional current and position when flowing into the referenced terminal," would be better placed in MIL-M-38510D or MIL-STD-883.

## 16. SUGGESTIONS FOR MIL-STD-883

This study revealed that to obtain repeatable and accurate data, certain precautions are required during test. In order to ensure that these precautions are taken, certain changes and additions to MIL-STD-883 are suggested.

As was previously discussed,  $I_{OS}$  measurements are affected if more than one output is shorted at a time, a practice that can cause power dissipation greater than the absolute maximum rating of the DUT. This practice is contrary to several general precautions in both MIL-STD-883 and MIL-M-38510 slash sheets. However, these general precautions are secondary test requirements. Therefore, the restriction is not always obvious to test engineers. This problem can be eliminated if paragraph 3 of Method 3011.1 is revised to read as follows:

The device shall be stabilized at the specified test temperature. Each output per package shall be tested individually. Output terminals not under test shall be open. Output terminals shall not be forced to the test condition voltage potential for a period longer than 5 seconds.

Differently defined temperature ambients were found to result in different junction temperatures for the same ambient or case temperature. MIL-STD-883B, paragraph 4.5.8 states

Control of junction temperature. Where the application of the specified electrical inputs during test will cause the device or any internal junction temperature ( $T_J$ ) to differ from the specified test temperature, case temperature ( $T_C$ ), or ambient temperature ( $T_A$ ) by more than  $\pm 2^\circ\text{C}$ , the following precautions shall be observed, as applicable: Where it is required that  $T_J = T_A$  or  $T_C$ , the device shall be temperature stabilized in the power-off condition until the device temperature is within  $\pm 2^\circ\text{C}$  of the case or ambient temperature, as applicable, and the prescribed power-on measurement shall be made as quickly as possible (and in no case in excess of 30 seconds) after the application of electrical inputs. Where it is required that  $T_J$  reach a normal operating level in excess of  $T_A$  or  $T_C$ , the device shall be stabilized for a sufficient period of time in a power-on condition with all specified electrical inputs applied to allow  $T_J$  to reach a temperature of at least 80 percent of its stable value under the specified test conditions.

As written, the requirement effectively minimizes the differences in junction temperature which can occur in testing the devices used in this study in differently defined ambients. It requires power-off until the test is started, as well as taking measurements as rapidly as possible to minimize test time. However, while these precautions are adequate for small scale devices, a large RAM will probably require test times in excess of 30 seconds. The  $T_J$  at normal operating level would be in excess of  $T_A$  and  $T_C$  in this case, but the actual  $T_J$  value would differ depending upon the temperature ambient. In order to standardize these measurements, MIL-STD-883B, par. 4.5.3 should be expanded to define a standard temperature ambient, with the manufacturer free to adjust other ambients to produce equivalent results. Since a fluorocarbon bath is not representative of a typical end-use environment, and both fluorocarbon and airstream environments lead to testing with  $T_J$  near to  $T_A$  or  $T_C$ , it is suggested that  $T_J$  at a "normal operating level" be further defined as  $T_J$  at a "normal operating level in an airstream with a flow in excess of 3 ft<sup>3</sup>/min over the DUT."

Measurements were found to be affected by a number of parameters unique to ATE, such as pin condition settling time and the lack of an ideal ground path. In order to remove the effects of these parameters, MIL-STD-883 needs to define some parameters of an ATE environment in the same manner that it specifies accuracy in paragraph 4.3.3 or permissible temperature variation in environmental chambers in paragraph 4.3.1.

It is suggested that a paragraph be added to Test conditions to define those parameters affecting ATE test results. This paragraph should include the definition of an acceptable open condition and an acceptable ground path. An acceptable open could be defined as  $\leq 50$  pf and  $\geq 10^{10}$  ohms and an acceptable ground path as  $\leq 0.1$  ohm and  $\leq 200$  nH.

A second item in the new paragraph should treat the problem of connecting a measurement system to an output, and then finding that the output has changed state. There should be a requirement that outputs shall be checked for proper state (proper state being defined by the type of measurement,

i.e., high level for  $V_{OH}$  and  $I_{OS}$  and low level for  $V_{OL}$ ) after the establishment of all measurement conditions. If the proper state is no longer present, the device must be conditioned to induce the proper state before measurement.

## 17. CONCLUSIONS

This study was conducted to evaluate the electrical test conditions of the MIL-M-38510 slash sheet specification for some small and medium scale integrated circuits in an automated test environment. It was determined that ATE is capable of testing these devices to end use requirements. There are some problems inherent in ATE equipment when testing to the specification requirements as they are presently written. However, if the revisions that have been recommended in this report are incorporated, the specifications and the automated test equipment will be more compatible.

The data obtained during this study supports the following conclusions:

1. The measurement sequence of different DC parameters does not affect test data provided measurement times are short.
2. Temperature sequence does not affect test data provided that an adequate stabilization time at temperature is allowed. For the TP450A used in this study, a stabilization time of 60 seconds proved adequate.
3. The sequence of AC parametric measurements does not affect test data. However, since the specification sequence minimizes required preconditioning, specifying that these measurements be run in sequence would minimize the amount of preconditioning information that should be provided.
4. The sequence of measurements of the same DC parameter does not affect test data but many sequences require additional preconditioning. The order in which  $V_{OH}$ ,  $V_{OL}$ , and  $I_{OS}$  measurements are taken on such sequential devices as flip-flops, counters, and shift registers should be specified.
5. Differently defined ambients result in different junction temperatures while testing at a specified  $T_A$  or  $T_C$ . Testing in fluorocarbon maintained  $T_J$  closest to  $T_A$ . Testing in an air stream produced junction temperatures approximating those of testing in fluorocarbon. Testing in still air or at case temperature in still air produced results similar to fluorocarbon when test times were short, but produced data indicating higher junction temperatures when test times were lengthened. A modification to MIL-STD-883 is proposed to assure that test results are not dependent on temperature environment.
6. Some variable measurement conditions should be more fully specified. Specifying  $V_{IL}$ ,  $V_{IH}$ , and PRR for truth table testing would improve the consistency of the measurements and provide  $V_{IL}$  and  $V_{IH}$  data that cannot be obtained from other measurements in the specification.

7. Pin condition settling time was found to affect DC parametric data. Schottky and low-power Schottky devices exhibited leakage currents which rose with increasing pin condition settling time. CMOS  $V_{ICP}$  measurements required pin condition settling times in excess of 20 milliseconds. In addition, the pin application sequence study indicated that attaching the measurement system of ATE to the output of the DUT can result in the DUT changing state and yielding a false measurement. To obtain consistent and repeatable data, a modification to MIL-STD-883 is required to deal with these problems which are peculiar to automated test equipment.
8. AC parametric data did not change for the devices tested for input waveform rise and fall times of 4 and 10 nanoseconds. It is concluded that specifying  $t_r$  and  $t_f$  as  $\leq 10$  nanoseconds would be acceptable from the standpoint of device performance and would permit a wider range of ATE to test to the specification.
9. The pin application sequence study revealed that in DC parametric testing, DC conditions must be established before the pulsed inputs are applied. This is necessary to ensure that the desired device state is obtained.
10. The sequence of time measurements does not affect the data, provided that the necessary preconditioning is performed. Additional preconditioning information is required in the MIL-M-38510 slash sheet specifications.
11. Undesignated prior output states were found to affect  $t_{PHL}$  for the 54164 and 54LS295 shift registers. Specifying the prior output states is required.
12. Improper loading of outputs other than the output under test affects  $I_{OS}$  measurements. An addition to MIL-STD-883, method 3011 is proposed to remove this effect.
13. Open and ground conditions in an ATE environment are less than ideal. Acceptable conditions must be defined. Conditions of open  $\leq 50$  pF,  $\geq 10^{10}$  ohm and ground  $\leq 200$  nH,  $\leq 0.1$  ohm are proposed. The proposed values are a compromise reflecting the capabilities of ATE. They also assure acceptable device performance in application.

Other conclusions drawn from this effort include the following:

1. Currently specified tri-state measurements can be implemented only with great difficulty on current ATE. An alternative technique is presented, but the need for further study is indicated.
2. Currently specified  $f_{MAX}$  tests cannot be implemented on many automated testing systems because of high PRR requirements. One alternative evaluated was found to be acceptable for most applications.



## **MISSION of Rome Air Development Center**

RADC plans and executes research, development, test and selected acquisition programs in support of Command, Control Communications and Intelligence (C<sup>3</sup>I) activities. Technical and engineering support within areas of technical competence is provided to ESD Program Offices (POs) and other ESD elements. The principal technical mission areas are communications, electromagnetic guidance and control, surveillance of ground and aerospace objects, intelligence data collection and handling, information system technology, ionospheric propagation, solid state sciences, microwave physics and electronic reliability, maintainability and compatibility.